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Study of Bird Ingestions into Small Inlet Area Aircraft Turbine Engines (May 1987-April 1989)

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FINAL REPORT

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16. Abstract

This report summarizes 2 years of data collection on ingestion of birds into small inlet area aircraft turbine engines. A total of 16.1 million engine operations were flown by aircraft equipped with the small inlet area engines (ALF502, TFE731, TPE331, and JT15D) included in the study. This includes 24 months of operations for the first three engines and 12 months of operations for the fourth. A total of 210 engine ingestion events were reported during the 2 years of data collection. This report analyzes these events to determine probability of ingestion, probability of degree of damage, probability of ingestion by phase of flight, and frequency of ingestion by geographic area.

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EXECUTIVE SUMMARY

An investigation was intitiated by the Federal Aviation Administration (FAA) Technical Center in May 1987 to determine the numbers, weight, and species of birds which are ingested into small inlet area turbofan and turboprop engines during worldwide service operation and to determine what damage, if any, results. Small inlet area engines are defined as those engines having an inlet area up to approximately 1400 square inches. This report presents an analysis of the 2 years of data. The purpose of the analysis is to assist the FAA in evaluating certification test requirements for such engines. In particular, this report presents information concerning ingestion events as related to time of day, phase of flight, month, location and bird species and weight.

Figure E-1 is an overall summary of the data that were collected during the 2-year period from May 1, 1987, to April 30, 1989. Throughout the world during that time there were approximately 16 million operations by the engines included in the data (ALF502, TFE731, TPE331 and JT15D). This figure includes 24 months of operations for the first three engines and 12 months of operation for the fourth. A total of 210 engine ingestion events were reported during this period. The probability of an engine ingestion event occurring is 1.3×10^{-5} per operation. Thus, the ingestion of a bird is a rare but not impossible occurrence.

Within the United States, the most frequently ingested bird weight is 4 ounces, while outside the United States, the most frequently ingested bird weight is 7.7 ounces. However, birds in the range of 0 to 4 ounces actually outnumber the birds in the range of 4 to 8 ounces. Within the United States, half the ingested birds weigh over 4 ounces, while outside the United States, the median weight is 7.7 ounces. Bird weights are based on identification of bird species.

Most bird ingestions occurred in the Northern Hemisphere. Several tests were made to detect seasonal patterns in these data. However, if seasonality exists, these tests as described in Section 3 were not able to detect it.

It was found that ingestions occurred more frequently in the daytime than at night. More than likely this is the result of two factors: fewer aircraft flights at night and more birds flying in the daytime.

No geographic patterns seem to be apparent in the bird ingestions in the United States. The Northeast and Midwest States seem to form a block of states with several ingestions, but no single state in that area had more than five (Ohio, second highest number in the nation). The largest number of ingestions (11) in one state occurred in California. This may be explained by a conjunction of many seabirds and a high level of aircraft activity.

It was determined that the engine ingestions could be described adequately by a Poisson distribution. This made it possible to test hypotheses about the relationship between engine size and ingestion rate. The data are consistent with the hypothesis that ingestion rates are directly related to engine cross section area. It was determined that the ingestion experience of the turboprop engine was different from that of the turbofan engines, but the reasons for this difference could not be determined.

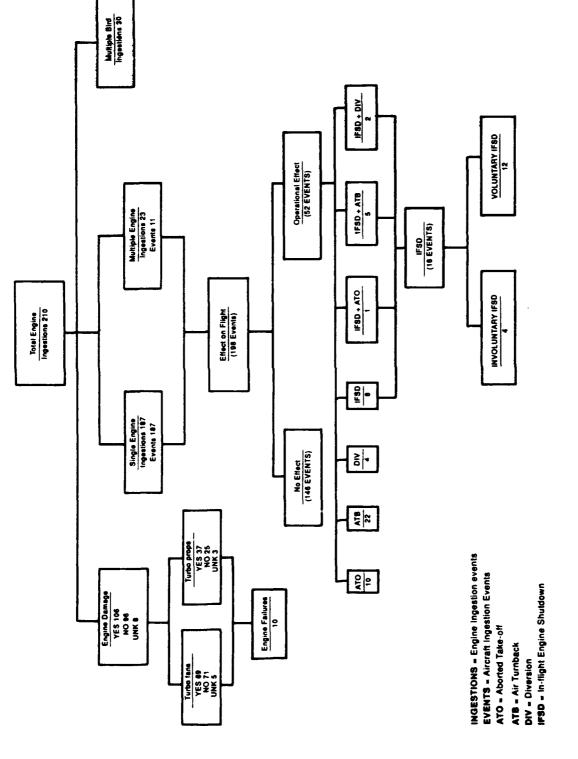


FIGURE E-1. SMALL INLET AREA TURBINE ENGINE BIRD INGESTION STUDY
DATA SUMMARY
(2 YEARS OF DATA, 5/87 TO 4/89)

It was observed that the same number of engine ingestion events occurred in the combined takeoff/climb phases of flight as in the combined approach/landing phases of flight. The ratio of landing events to approach was close to one 55:45), whereas the ratio of takeoff events to climb events exceeded ten (91:9). Less than 5 percent of all ingestion events occurred during taxi or at cruise altitude.

Engine damage occurred in 50 percent of all engine ingestion events, and it was not the case that there was a threshold bird weight such that smaller birds did no damage and larger birds always caused damage. Instead, the probability of damage increased with bird weight. However, in some events small birds caused damage, while in other events larger birds caused no damage at all. Probability-of-damage versus bird-weight curves were computed from the data. Also, the probability of engine damage is greater when the bird ingestion occurs during the takeoff and climb phases of flight than when it occurs during approach and landing. Aircraft airspeed at or above 140 knots also increases the probability of engine damage.

It was determined that 5 percent of all engine bird ingestion events resulted in an engine failure. Four engine failures were caused by birds that weighed more than 4 pounds and two were caused by birds that weighed less than 1/2 pound. Engine failures are also more likely to occur when multiple birds are ingested into an engine.

It was observed that as the level of damage increased, the probability of crew action likewise increased. For turbofan engines, the probability of crew action was 6.6 percent after engine ingestion events in which there was no damage, while probability of crew action was 42 percent after engine ingestion events in which there was severe damage. For the turboprop engine, the probability of crew action for events with no engine damage was 16 percent.

It was found that the probability of ingestion for birds in the weight range from 0 to 4 ounces (the most common range) was 1.98 per million operations. Overall, the probability of ingesting a bird was 13 per million engine operations.

A summary of the most pertinent statistics extracted from the 2 years of data is provided below:

Most Frequently Ingested Bird Weight (oz)	
United States	4
Foreign	7.7
Average Bird Weight (oz)	
United States	21
Foreign	9.2
Median Bird Weight (oz)	
United States	4
Foreign	7.7

Probability of Ingestion per Engine Operation Worldwide (all engine types) United States (JT15D engine excluded) Foreign (JT15D engine excluded)	1.3×10^{-5} 1.04×10^{-5} 1.922×10^{-5}
Most Commonly Ingested Bird	
United States	Dove
Foreign	Lapwing
Engines Experiencing Moderate/Severe Damage	
Turbofans	41
Turboprops	2
Ingestions During Phase of Flight	
Takeoff and Climb	100
Approach and Landing	100

SECTION 1 INTRODUCTION

1.1 BACKGROUND.

Contention for airspace between birds and airplanes has created a serious bird/aircraft strike hazard. Four past studies [references 1,2,3 and 4] have indicated that birdstrikes to engines are statistically rare events. The probability of a birdstrike during any given flight is extremely low; however, given the number of flights currently taking place, the expected number of birdstrikes becomes significant.

The windshield and the engines are particularly vulnerable to the birdstrike threat. Although penetration of the windshield by a bird is primarily a concern for military airplanes operating at high speeds in a low-altitude environment, such a penetration occurred on a civilian airplane resulting in the death of the copilot. Ingestion of birds into airplane engines is a safety problem for civil as well as military airplanes for it can cause significant damage to the engine, resulting in degraded engine performance and possibly failure.

In his study of bird ingestions on commercial flights, Frings [reference 1] indicated that nearly all bird ingestion events have occurred in the vicinity of airports during the noncruise phases of flight. Hovey and Skinn [references 2 and 3] reached similar conclusions. This is understandable because these phases of flight naturally occur closer to the ground where bird concentrations are higher, resulting in a higher probability of birdstrike.

The solution to the problem of engine damage resulting from bird ingestion is similar to that for windshield birdstrike, e.g., either design-consideration of the structure to withstand impact, and/or avoidance of birds. Bird avoidance can be facilitated by either of two approaches: (1) keeping airplanes out of airspaces with large bird concentrations, or (2) removing birds from these regions of airspace. The bird avoidance approach can have various degrees of success or failure for commercial air fleets because flight schedules place airplanes in specific areas at specific times and the effectiveness of airport bird control programs (if any) varies from airport to airport and country to country.

Structural design of engines to withstand bird ingestion damage can be accomplished given that realistic requirements with respect to bird sizes and numbers can be identified. Bird ingestion data for various sizes of turbofan and turboprop engines are currently being collected by several engine manufacturers. Statistical evaluation of bird ingestion data from these data collection efforts and previous bird ingestion studies will be useful in re-evaluating certification test regulations laid out in FAA Regulation 14 CFR 33.77. As a result, future engines can be designed to withstand more realistic bird threats.

1.2 OBJECTIVE.

The objective of this report is to determine the relationship of bird weight, geographic location, season, time of day, phase of flight, and engine type to the frequency of bird ingestion events and the extent of engine damage resulting from

the ingested birds. A statistical analysis was conducted of reported bird ingestion data experienced by commercial and general aviation aircraft equipped with any of four engine types (ALF502, TPE331, TFE731 and JT15D) operating worldwide over a 2-year reporting period from May 1987 through April 1989. The analysis was used to summarize the bird ingestion damage experienced by these engines. The findings of the analysis will be used to determine the adequacy of the bird ingestion test criteria as specified in FAA regulation 14 CFR 33.77 for this class of small inlet area engines. Small inlet area engines are being defined a those engines having an inlet area up to approximately 1400 square inches.

1.3 ORGANIZATION OF REPORT.

Section 2 presents engine hours and operations for the four engines. Section 3 identifies the characteristics of bird species that have been ingested and reliably identified. Section 4 describes bird ingestion rates by location, engine type, and phase of flight. Section 5 summarizes engine damage resulting from bird ingestions. Section 6 examines the probabilities of various bird ingestion events. Section 7 discusses data quality. Section 8 provides a summary of the results obtained during this phase of data analysis. Section 9 lists references used in preparation of this report. Section 10 is a glossary of terms. Appendix A provides information about size and use of the engines covered in this report. Appendix B provides the original data used in the analysis. Appendix C discusses the methods of statistical analysis used in the report, particularly hypothesis testing.

SECTION 2 ENGINE OPERATIONS

The number of engine operations is required to determine bird ingestion rates. Operations data that have been used to generate bird ingestion rates throughout the report are provided to aid in understanding this section. The reader should refer to the Glossary of Terms for definitions of the terms used.

For the ALF502, data on engine hours and engine operations were available from the manufacturer through the FAA. For the TPE331, JT15D, and TFE731, only data on engine hours were available. To obtain engine operations, average values of 0.8 operations/hours (TFE731), 0.9 operations/hours (JT15D), and 1.2 operations/hours (TPE331) were provided through the FAA. Numbers of engine operations by month for the ALF502, TFE731, and TPE331 engines are presented in tables 2.1, 2.2 and 2.3, respectively. Because total operations for the TFE731 and TPE331 engines are obtained by using the aforementioned flight hour conversion factors, certain monthly, United States, foreign, and overall total operations in tables 2.2 and 2.3 appear as incorrect sums of individual monthly operations. Rounding error accounts for the arithmetic discrepencies. Figures 2.1, 2.2 and 2.3 are histograms displaying operations by month and engine.

Data for the JT15D were provided only as a total: 872,510 hours for the period May 1, 1988, to April 30, 1989. A conversion factor of 0.9 operations/hours results in a total of 785,259 operations for this engine. No information by month is available for this engine.

TABLE 2.1. HOURS AND OPERATIONS ALF502

Da	te	Unit	ed States	1	Foreign Total		Total
Month	Year	Hours	Operations	Hours	Operations	Hours	Operations
MAY8	7	39290	44167	8275	7538	47565	51705
JUN8'	7	39290	44167	8275	7538	47565	51705
JUL8.	7	46118	53719	10336	8689	56454	62408
AUG8	7	47163	54699	12139	10130	59302	64829
SEP8	7	43865	51507	9219	7842	53084	59349
OCT8	7	46311	52987	12621	9795	58932	62782
NOV8	7	43550	50574	12377	10205	55927	60779
DEC8	7	43032	49247	11995	10418	55027	59665
JAN88	3	46366	50244	10427	10706	56793	60950
FEB88	3	46366	48185	10184	11922	56550	60107
MAR88	3	41430	48185	9304	11866	50734	60051
APR88	3	45168	49224	16300	18364	61468	67588
MAY88	3	43484	50812	17136	16020	60620	66832
JUN88	3	43724	50932	21352	19104	65076	70036
JUL88	3	44040	51086	21956	19408	65996	70494
AUG88	3	45868	53220	22224	20340	68092	73560
SEP88	3	41148	47956	23968	22932	65116	70888
OCT88	3	45200	51656	24284	23148	69484	74804
NOV8	3	42836	48216	24536	24604	67372	72820
DEC88	3	43328	48448	25760	24564	69088	73012
JAN89	•	43748	49212	26654	25851	70402	75063
FEB89	•	40056	44110	25738	26367	65794	70477
MAR89	•	30700	48780	32319	33715	63019	82495
APR89	•	40020	46648	33060	34288	73080	80936
Total	1	032101	1187981	430439	415354	1462540	1603335

TABLE 2.2. HOURS AND OPERATIONS TFE731

Da	te	United	States	F	oreign		Total
Month	Year	Hours	Operations	Hours	Operations	Hours	Operations
MAY8	-	127148	101718	45189	36151	172337	137870
JUN8		128132	102506	46060	36848	174192	139354
JUL8	-	130058	104046	46028	36822	176086	140869
AUG8	-	132051	105641	48274	38619	180325	144260
SEP8		131189	104951	46967	37574	178156	142525
OCT8	•	132677	106142	48595	38876	181272	145018
8VON	·=	134888	107910	49968	39974	184856	147885
DEC8		135142	108114	51393	41114	186535	149228
JAN8	-	131583	105266	50585	40468	182168	145734
FEB8	-	134338	107470	49942	39954	184280	147424
MAR8	_	140277	112222	52557	42046	192834	154267
APR8	-	141617	113294	53424	42739	195041	156033
MAY8	_	132631	106105	49215	39372	181846	145477
JUN8	_	131509	105207	49084	39267	0593	144474
JUL8	_	131517	105214	50924	40739	_62441	145953
AUG8	-	131881	105505	51783	41426	183664	146931
SEP8	_	130933	104746	50872	40698	181805	145444
OCT8	8	134926	107941	52596	42077	187522	150018
NOV8	8	144838	115870	54334	43467	199172	159338
DEC8	8	138015	110412	51316	41053	189331	151465
JAN8	9	135526	108421	50296	40237	185822	148658
FEB89	9	142042	113634	51414	41131	193456	154765
MAR89	9	139941	111953	54156	43325	194097	155278
APR8	9	148383	118706	56031	44825	204414	163531
Tota:	1 :	3241242	2592994	1211003	968802	4452245	3561796

Note: Detail may not add to total because of rounding.

TABLE 2.3. HOURS AND OPERATIONS TPE331

Da	te	Unit	ed States	F	oreign		Total
Month	Year	Hours	Operations	Hours	Operations	Hours	Operations
MAY8	7	206666	247999	81385	97662	288051	345661
JUN8	7	211357	253628	89138	106966	300495	360594
JUL8	7	234047	280856	93231	111877	327278	392734
AUG8	7	232892	279470	93280	111936	326172	391406
SEP8	7	232924	279509	95408	114490	328332	393998
OCT8	7	237444	284933	97521	117025	334965	401958
NOV8	7	237631	285157	101077	121292	338708	406450
DEC8	7	230677	276812	95275	114330	325952	391142
JAN8	В	237817	285380	97319	116783	335136	402163
FEB8	8	251480	301776	88360	106032	339840	407808
MAR88	3	250675	300810	93553	112264	344228	413074
APR88	3	261232	313478	100541	120649	361773	434128
MAY88	3	249151	298981	116604	139925	365755	438906
JUN88	3	253131	303757	116706	140047	369837	443804
JUL88	3	249269	299123	119622	143546	368891	442669
AUG88	3	250314	300377	120657	144788	370971	445165
SEP88	3	263965	316758	116854	140225	380819	456983
OCTS	3	252292	302750	1187 9 8	142558	371090	445308
88VON	_	255233	306280	120698	144838	375931	451117
DEC88	3	255934	307121	122375	146850	378309	453971
JAN89	•	268975	322770	121914	146297	390889	469067
FEB89	•	259072	310886	122810	147372	381882	458258
MAR89	•	254644	305573	124848	149818	379492	455390
APR89	•	266753	320104	126383	151660	393136	471763
Total	L !	5903575	7084288	2574357	3089229	8477932	10173518

Note: Detail may not add to total because of rounding.

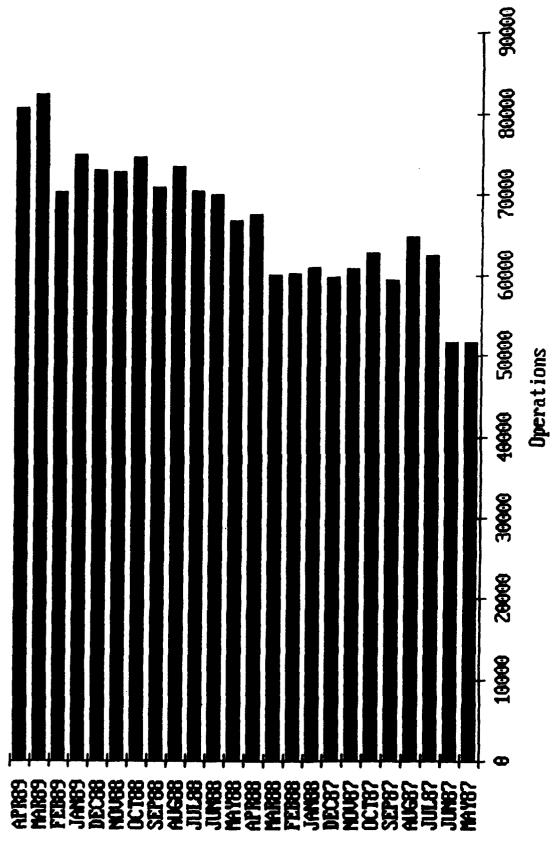


FIGURE 2.1. OPERATIONS, ALF502 ENGINE

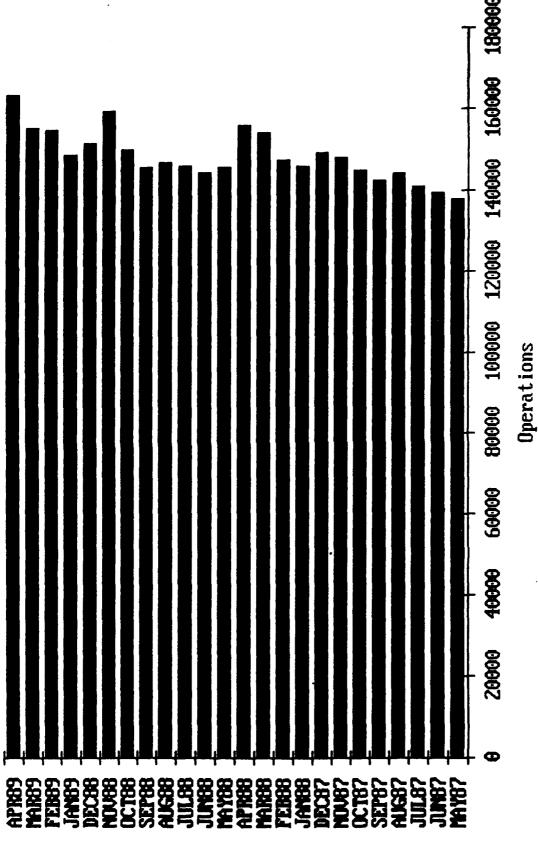


FIGURE 2.2. OPERATIONS, TFE731 ENGINE

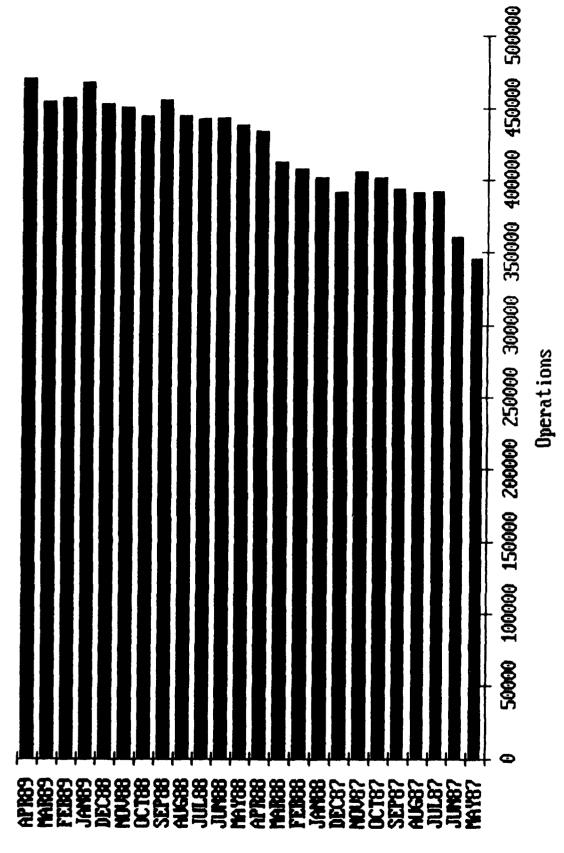


FIGURE 2.3. OPERATIONS, TPE331 ENGINE

SECTION 3 CHARACTERISTICS OF INGESTED BIRDS

The purpose of this section is to provide a description of the birds that were ingested during the period covered by the data and to provide an analysis of the extent of the bird ingestion threat. The bird related features that are described in this section include species, weight, seasonal trends, time-of-day trends, and geographic location.

Table 3.1 provides a tally of all the species that were positively identified by an ornithologist during the period covered by the data. The species are listed by order and family. One of the disappointing features of the small engine bird ingestion data base is the low bird identification rate. Out of the total of 198 aircraft ingestion events that were recorded, the bird species was positively identified in only 70 events, for a total identification rate of 35.4 percent.

Table 3.2 presents the distribution of weights for the positively identified birds. The numbers in table 3.2 reflect the number of times birds of a given weight were encountered. That is, if more than one bird was ingested in one or more engines, the bird weight was counted once only. Thus the table is not skewed by multiple-bird or multiple-engine ingestions from the same flock of birds. The bird weights are derived from the species identification and when possible are adjusted for the age and sex of the ingested bird. Figure 3.1 presents the same data in the form of a histogram.

There were 30 cases where multiple birds were ingested into the same engine, and 11 cases where bird ingestions occurred in multiple engines during the same event. These cases, of multiple bird ingestions and multiple engine events, are important from a safety standpoint. However, the data contain too few cases to allow any conclusions to be drawn.

A comparison of the distribution of bird weights for United States and foreign ingestion events was carried out using the Kolmogorov-Smirnov test. The maximum deviation between the distributions was 0.176. By chance, a deviation of 0.39 would be exceeded five times in a hundred. Hence at a significance level of 0.05, the hypothesis that the weights of ingested birds in the United States and outside the United States are the same cannot be rejected. (For a brief explanation of statistical terms see appendix C.)

Summary statistics calculated from the raw data for the United States, foreign, and worldwide bird weight distributions are presented in table 3.3. The statistics presented are the mode, the median, and the mean. These three statistics each represent an attempt to identify a "typical" member of a distribution. The mode is the most common value in the distribution, the median is the value which splits the distribution into two equal halves, and the mean is weighted by each value appearing in the distribution as well as the number of times it appears.

The mode is a relevant measure of the bird ingestion problem. It represents the weight which will be encountered most frequently. In the United States, the modal weight is 4 ounces, while outside the United States the modal weight is 7.7 ounces. Worldwide the modal weight is also 4 ounces. These modal weights correspond to the most frequently encountered species in each case. It is possible to have multimodal distributions, but the weight distributions of birds ingested during the period covered by the data turned out to be unimodal.

TABLE 3.1. TALLY OF POSITIVELY IDENTIFIED BIRD SPECIES BROKEN DOWN BY US, FOREIGN, AND OVERALL

Latin Name	Common Name	Species	si	Foreign	Unknown	Overall
Givia	Common Loon	163	-	0	0	-
Nyctanassa violacea	Yellow-crowned night heron	1127	-	0	0	•
Chen caerulescens	Snow Goose	2326	2	0	0	~
Branta canadensis	Canada goose	2,30	2	0	0	۲.
Ands americana	American wigeon	2,171	e -	0	0	-
Anas platyrhynchos	Mallard	2.184	0	-	0	_
Aythya affinis	Lesser scaup	2,1125	τ-	0	0	~
Cathartes aura	Turkey vulture	1X	₹~	0	0	-
Haliastur indus	Brahminy kite	3K31	0	-	0	-
Falco sparverius	American kestrel	5K26	-	,	0	2
Falco tinnunculus	Eurasian kestrel	5K27	0	-	0	•
Perdix perdix	Hungarian partridge	4185	0	-	0	-
Phasianus colchicus	Ring-neck pheasant	41161	τ-	0	0	-
Gallinula chloropus	Common gallinule	7M112	, -	0	0	-
Vanellus vanellus	Common Lapwing	SN1	0	'n	0	v
Charadrius vociferus	Killdeer	5N33	9	0	0	9
Charadrius mongolus	Mongolian plover	5N45	0	-	0	-
Tringa melanoleuca	Greater yellowlegs	6N19	ς-	0	0	-
Tringa flavipes	Lesser yellowlegs	6N20	-	0	0	-
Scolopax minor	American woodcock	6N37	-	0	0	-
Larus delawarensis	Ring-billed gull	14N12	m	2	0	5
Larus Canus	Common gull	14N13	,	0	0	_
Larus argentatus	Herring gull	14N14	-	0	0	-
Larus pipixcan	Franklin's gull	14N31	ς-	0	0	-
Larus ridibundus	Common black-headed guil	14N36	0	-	0	-
Columba Livia	Common rock dove	2P1	4	0	0	4
Columba palumbus	Common wood-pigeon	2P9	0	_	0	-
Streptonpelia chinensis	Spotted dove	2P65	0	-	0	
Zenaida macroura	American mourning dove	2P105	6	0	0	σ.
Tyto alba	Common barn owl	152	-	0	0	-
Asio otus	Northern long-eared owl	25120	0	-	0	_
Apus melba	Alpine swift	1052	0	-	0	-
Apus apus	Common swift	1055	0	_	0	-
Eremophila alpestris	Horned Lark	17274	~	0	0	~
Hirundo rustica	Barn swallow	18237	-	0	0	-
Delichon urbica	Common house martin	18269	0	•	0	•
Sturnus vulgaris	Common starting	21275	-	0	-	2
Turdus philomelos	Common song thrush	412282	0	•	0	-
Agelaius phoeniceus	Red-winged blackbird	94754	-	0	0	-
Sturnella magna	Eastern meadowlark	29759	-	0	0	-
Passer domesticus	House sparrow	70212	~ ~	0	0	-
Passer montanus	Eurasian tree sparrow	70223	0	-	0	~
			}	;	•	!

TABLE 3.2. DISTRIBUTION OF BIRD WEIGHTS (AIRCRAFT INGESTION EVENTS)

Weight (oz)	US	Foreign	Unknown	Total
$0 < x \le 4$	23	8	1	32
4 < x < 8	2	7	0	9
$8 < x \le 12$	3	2	0	5
$12 < x \le 16$	6	4	0	10
$16 < x \le 20$	2	2	0	4
$20 < x \le 24$	1	0	0	1
$24 < x \le 28$	1	0	0	1
$32 < x \le 36$	0	1	0	1
$36 < x \le 40$	2	0	0	2
$64 < x \le 68$	1	0	0	1
$88 < x \le 92$	2	0	0	2
$100 < x \le 104$	1	0	0	1
$124 < x \le 128$	2	0	0	2
Totals	46	24	1	71

(Note: this table includes one bat, not included in table 3.1)

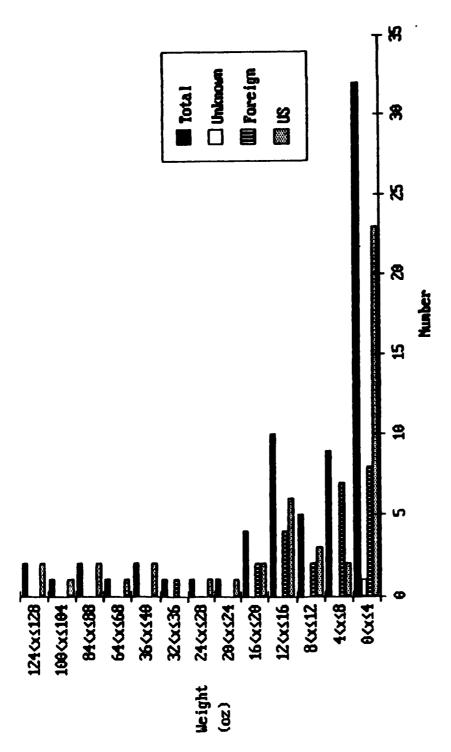


FIGURE 3.1. DISTRIBUTION OF BIRD WEIGHTS

TABLE 3.3. SUMMARY STATISTICS FOR INGESTED BIRD WEIGHTS

<u>Statistic</u>	<u>us</u>	Foreign	<u>Worldwide</u>
Mode	4	7.7	4
Median	4	7.7	7.7
Lower Quartile	3	2	3
Upper Quartile	17	14	16
Interquartile Range	14	12	13
Mean	21.01	9.21	16.77
Standard Deviation	33.20	8.19	27.65

Note: All weights in ounces.

The median is the value which divides the distribution in half. Median weights are 4 ounces in the United States, 7.7 ounces outside the United States, and 7.7 worldwide. The quartiles divide the upper and lower halves of a distribution in half. Each is a value one-quarter of the way in from the end of the distribution. In the United States, 25 percent of the birds had weight equal to or exceeding 17 ounces, while outside the United States the top 25 percent of birds had weights equal to or exceeding 14 ounces. In the United States, 25 percent of the birds weighed 3 ounces or less, while outside the United States the lowest 25 percent of the weights included birds only up to 2 ounces. The Interquartile Range (IQR) is the distance between the upper and lower quartiles - the "middle half" of the distribution. It is a measure of the dispersion of values in the distribution. In the United States the IQR is 14 ounces, while outside the United States it is 12 ounces. Worldwide it is 13 ounces. simply means that inside and outside the United States, the degree of clustering about the median is nearly the same, even though the medians differ by roughly a factor of two. However, outside the quartiles the spread of bird weights is greater in the United States. This can be seen from table 3.2, which shows that outside the United States the weight of ingested birds did not exceed 36 ounces, while in the United States there were birds with weights up to 128 ounces.

The mean is obtained by weighting each value in the distribution by the number of times which it occurs. Moreover, it is a function of the sum of all the values in the distribution. The mean tends to be influenced by extreme values. In the case of the bird weight distributions, the mean is influenced by the high values, and thus overestimates the weight of the "typical" ingested bird. The mean would be a relevant measure of ingested bird weight if damage were related to the cumulative weight of all birds ingested by a single engine, since it does depend upon the total weight of the ingested birds. However, since bird ingestion is such a rare event, the mean is not a particularly useful measure of ingested bird weight.

From the standpoint of descriptive statistics, then, the important results from table 3.3 are that the most frequently ingested birds weigh 4 ounces: but 50 percent of all ingested birds weigh 7.7 ounces or more, and fully 25 percent of all ingested birds weigh more than 16 ounces.

One issue which might be raised is the extent to which the ingestion events in which the bird weight is known are representative of all ingestion events. It might be hypothesized that the bird species is more likely to be identified (and therefore the weight known) in those cases in which greater damage has been incurred, while bird weight is less likely to be known if lesser or no damage occurred. The chi-square test was applied to this hypothesis. A chi-square value of 4.8 was obtained, comparing the actual numbers of identified birds with the hypothesis that the same fraction of birds were identified regardless of damage level. With 3 degrees of freedom, a value for chi-square of 6.25 would be exceeded with a probability of 10 percent. Hence the hypothesis that the same fraction of birds are identified regardless of the damage level cannot be rejected, and one can conclude that the ingestion events in which bird weight is known are representative of all ingestion events.

Figure 3.2 presents a histogram of ingestions by month for the 2-year period covered by the data. Each bar in figure 3.2 represents the sum of ingestions from its respective month in 2 consecutive years. It is known that the number of ingestions per month should be influenced by seasonality (bird migrations) and by

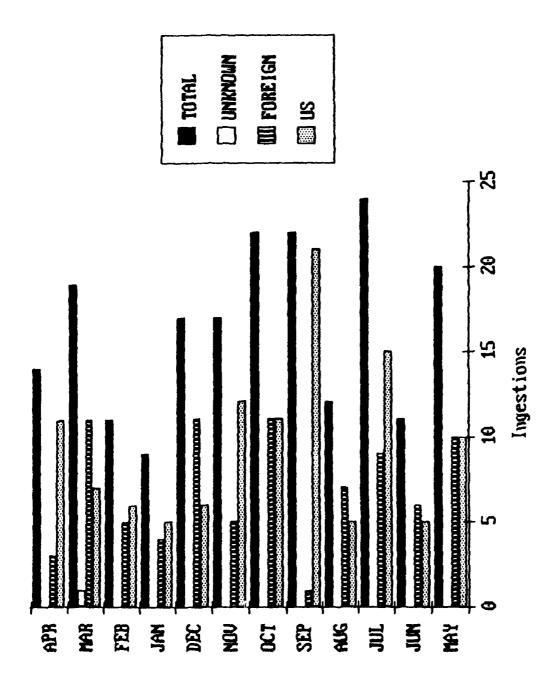


FIGURE 3.2. AIRCRAFT INGESITONS BY MONTH FOR 2 YEARS

number of operations. However, the effects of these factors could not be separately identified in the data. Since ingestion locations were known, the numbers of ingestions could be categorized as United States or foreign, and also as Northern or Southern Hemisphere. Numbers of engine operations could be separated only into United States or foreign. Hence ingestions in either hemisphere could not be normalized to numbers of operations.

The variation in number of ingestions from month to month is not only highly volatile but appears random. Several tests for randomness, trend, or seasonality were applied.

A chi-square test was used to test for differences between patterns of monthly ingestions inside and outside the United States (including both hemispheres). The test found a significant difference. However, there is some question of whether this finding should be taken seriously. Nearly half the total value of chi-square came from the months of September, in which the United States had a total of 21 ingestions while there was only one ingestion outside the United States.

A Kolmogorov-Smirnov test was likewise applied to United States versus foreign monthly ingestions. This test found that the difference between the two sets of ingestions was not significant at the l percent level. This reinforces the suggestion that the chi-square test result was the result of statistical anomaly, that is, accepting the hypothesis of a difference would be to commit a Type I error.

A chi-square test was applied to the Northern Hemisphere data alone, to detect United States versus foreign differences uncontaminated by differing seasonality in Northern and Southern Hemispheres. The difference between the two was not found to be significant.

Several tests were applied to detect seasonality if it existed.

A chi-square test was used to determine if there were significant departures from a uniform distribution across the months. This test found a significant difference. However, again a significant share of the total chi-square value was accounted for by the months of September alone. Hence this test must be viewed as possibly spurious.

A linear regression was also performed on the Northern Hemisphere ingestions on the months, in sequence, to detect any trends. The slope of the regression was -0.608 and the standard error of the slope was 0.459. Hence the slope was not significantly different from zero. On the basis of this test, the hypothesis of no trend in the data cannot be rejected.

A Fourier analysis of the month-to-month variation in ingestions in the Northern Hemisphere was carried out in an attempt to find periodicity in the data. The magnitude of the second harmonic (two peaks and two troughs) was only 23 percent of the average monthly ingestion rate. At best, this would be only weak evidence for periodicity (seasonality). Moreover, one of the troughs of the second-harmonic fit coincided with the month of the greatest number of ingestions, while one of the peaks of the second-harmonic fit coincided with the month in which ingestions were slightly below average. This result indicates that if seasonality is present in the Northern Hemisphere data, it is buried in the noise.

Figures 3.3, 3.4 and 3.5 present histograms of aircraft ingestion events by time of day for the period covered by the data. Figure 3.3 shows aircraft ingestion events by time of day. A chi-square analysis allows rejection of the hypothesis that number of ingestions is uniformly distributed throughout the day. The actual value of chi-square was 70.1, while a value of 9.4 would be exceeded by chance only 2.5 percent of the time. The variation in number of ingestions by time of day can be explained by either or both of two factors. First, many birds tend to be diurnal and are less likely to be exposed to ingestion at night. Second, most aircraft operations occur in the middlle of the day, with fewest at night. Numbers of operations in the morning and the evening are intermediate between the midday and night levels. Both these factors probably influence the variation by time of day in the number of ingestions.

During all time periods, the number of ingestions in the United States was greater than the number of ingestions outside the United States. However, a chi-square test showed that there was no significant difference in the patterns of ingestions in the United States and outside the United States by time of day. The actual value of chi-square was only 1.91. This value would be exceeded by chance 25 percent of the time. A chi-square value of 9.4 would be required for the difference to be statistically significant at 2.5 percent.

Figure 3.4 shows numbers of aircraft ingestion events in which more than one bird was ingested into the same engine. The total number of events is not sufficient to permit any statistical analysis. However, there were more ingestion events during the morning hours than in any other period of the day.

Figure 3.5 shows numbers of ingestion events in which birds were ingested in more than one engine. There were too few multiple engine ingestion events to permit any statistical analysis. The distribution appears uniform across the day, with the only difference between United States and foreign events being one foreign event of an ingestion during the night.

For some ingestions, time of day was not stated. These are shown as Unknown in figures 3.3, 3.4 and 3.5. Note that the total unknown count in figure 3.3 exceeds the sum of United States and foreign counts by one because the geographic location of one event is also unknown.

The geographic distribution of aircraft ingestion events within the United States is shown in figure 3.6. California had the largest number of aircraft ingestion events with 11. This may be due to a combination of a large coastal bird population and heavy air traffic. The state with the second largest number of aircraft ingestion events was Ohio with 5. However, there appears to be a concentration of events east of the Mississippi and south of the Great Lakes, extending to the Atlantic coast. This is probably the result of heavy air traffic in this region, with many cities, many airports, and frequent operations.

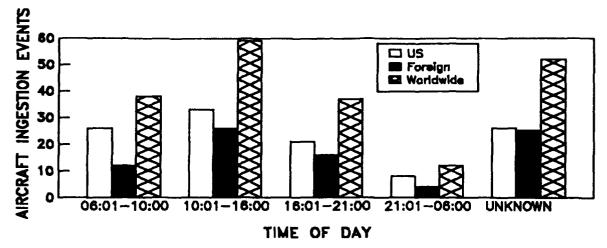


FIGURE 3.3. INGESTIONS BY TIME OF DAY

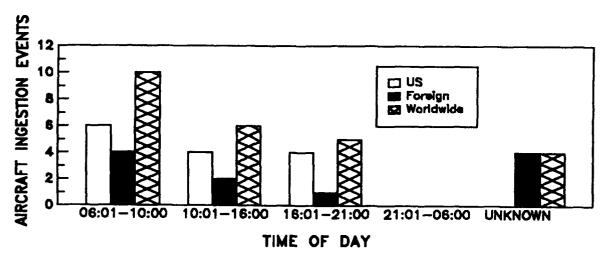


FIGURE 3.4. MULTIPLE BIRD INGESTIONS BY TIME OF DAY

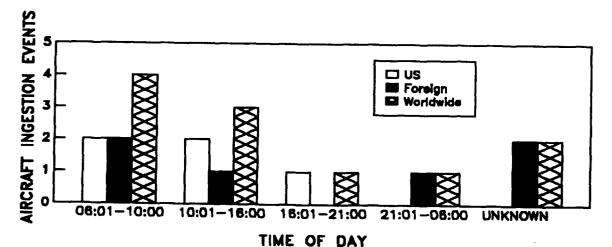


FIGURE 3.5. MULTIPLE ENGINE INGESTIONS BY TIME OF DAY

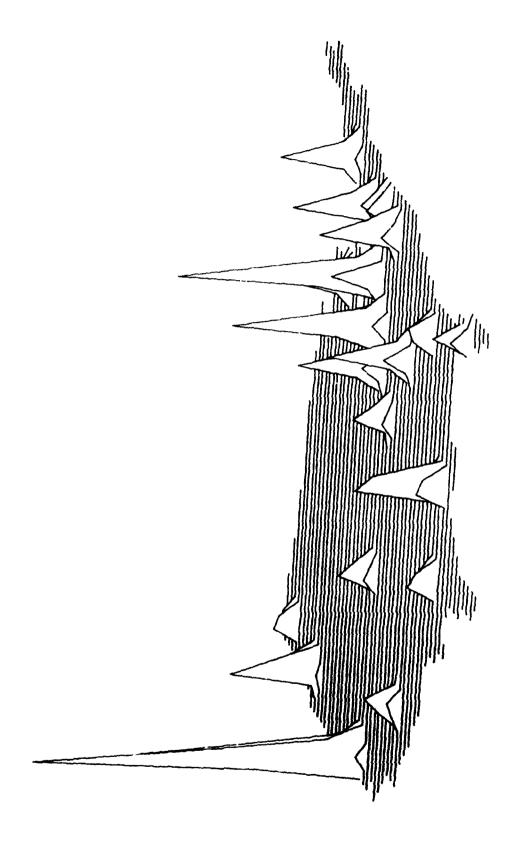


FIGURE 3.6. CONTOUR MAP OF DOMESTIC AIRCRAFT INGESTION EVENTS

SECTION 4 INGESTION RATES

This section describes the rates at which bird ingestions occurred during the period covered by the data. While *he term "rate" usually implies occurrences per unit time, in this case it refers to occurrences per engine operation or per aircraft operation. The Poisson distribution is commonly used to describe how events are randomly distributed in time, and the bird ingestion data are shown to agree with the assumption of a Poisson process. The first part of this section provides the estimates of the basic ingestion rates. The second part describes the Poisson distribution and how it relates to the bird ingestion events. The final parts discuss statistical analysis based on the assumption that bird ingestions follow a Poisson process.

4.1 INGESTION RATE ESTIMATES.

This section provides a general description of ingestion rates by location, by engine, and by phase of flight. The rates are given in terms of ingestions per 10,000 engine operations and have been adjusted for differences in inlet size of the engine where appropriate. A more detailed statistical analysis of ingestion rates is presented in subsequent sections, using statistical techniques for Poisson processes.

Table 4.1 presents engine ingestion rate data for each of the four small engines. The data presented include number of ingestions, rate per 10K operations, rate per 10K operations normalized to a 10-square-foot inlet area, and rate per 10K operations normalized to a 1-foot engine diameter. The Aerospace Industries Association (AIA) uses the inlet throat dimension in analyses involving engines. The analysis of engine dimension will therefore use throat dimension. A discussion of inlet area and inlet diameter effects on ingestion rates is given in Sections 4.4 and 4.5. These rates were calculated using the reported and estimated data on operations presented earlier in this report.

Table 4.2 presents data on engine ingestion events and rates by phase of flight for all engines and for each engine separately. The 95 percent Upper Confidence Bound on Ingestions per 10,000 operations is also given (e.g., the bounds are 95 percent likely to contain the true value, allowing for sampling fluctuation). Overall, most ingestion events occurred during takeoff, followed by the landing and approach phases. Note that those ingestion events not specifically identified with a phase of flight were allocated across phases in the same proportions as the identified ingestion events. For the individual engines, the same pattern holds generally, with the exception of the ALF502 which had seven more ingestion incidents during landing than during takeoff. Overall it appears that the takeoff phase poses the highest risk from the standpoint of rate of bird ingestions. Note that because of the small sample size, some phases of flight were not represented among the ingestion events.

TABLE 4.1. ENGINE INGESTION RATES

	<u>ALF502</u>	<u>TFE731</u>	TPE331	<u>JT15D</u>	<u>Total</u>				
Engine Ing	Engine Ingestion Events								
US	34	37	42	6	119				
Foreign	29	34	23		90				
Worldwide	63	72¹	65	10	210				
Engine Hou	Engine Hours								
US	1032101	3241242	5903575						
Foreign	430439	1211003	2574357						
Worldwide	1462540	4452245	8477932	872510	15264327				
Engine Inges	Engine Ingestion Events/10K Engine Hours								
US	0.329	0.114	0.071						
Foreign	0.674	0.281	0.089						
Worldwide	0.431	0.162	0.077	0.115	0.138				
Engine Ope	rations								
US	1187981	2592274	7084288		10864543				
Foreign	415354	968802	3089229		4473385				
Worldwide	1603335	3561076	10173517	785259	16123187				
Engine Ingest	Engine Ingestion Events 10K Engine Operations								
US	0.286	0.143	0.059						
Foreign	0.698	0.351	0.074						
Worldwide	0.393	0.202	0.064	0.127	0.130				
Inlet Area (in units of 10 square feet)									
	0.683	0.3125	0.051	0.215	•				
Engine Ingest	Engine Ingestion Events/10K ops/10 sq. ft. Inlet Area								
US	0.419	0.457	1.162						
Foreign	1.022	1.123	1.460						
Worldwide	0.575	0.647	1.253	0.592	0.725				
Worldwide ((turbofans	only)			0.610				
Inlet Diamete	Inlet Diameter (ft.)								
	2.949	1.995		1.655					
Engine Ingestion Events/10K ops/ft. inlet diam. (turbofans only)									
US	0.097	0.072							
Foreign	0.237	0.176							
Worldwide	0.133	0.101		0.077	0.110				
									

Note: One operation incident not identified as to location; included here in total but not in specific location.

TABLE 4.2. ENGINE INGESTION EVENTS AND RATES BY PHASE OF FLIGHT

	Engine Ingestion Events	Events per 10K Operations	95% Upper Bound	Events per 10K Operations per 10 sq. ft. Inlet Area
ALF502 Approach Climb Cruise Landing Takeoff Taxi	9 0 0 28 22 4	0.056 0.000 0.000 0.175 0.137 0.025	0.098 0.019 0.019 0.239 0.196 0.057	0.082 0.000 0.000 0.256 0.201 0.037
TFE731 Approach Climb Cruise Landing Takeoff Taxi	12 4 1 15 38 1	0.034 0.011 0.003 0.042 0.107 0.003	0.055 0.026 0.013 0.065 0.140 0.013	0.108 0.036 0.009 0.135 0.341 0.009
JT15D Approach Climb Cruise Landing Takeoff Taxi	2 1 2 0 5	0.025 0.013 0.025 0.000 0.064 0.000	0.080 0.060 0.080 0.038 0.134 0.038	0.118 0.059 0.118 0.000 0.296 0.000
TPE331 Approach Climb Cruise Landing Takeoff Taxi	22 4 1 12 26 0	0.022 0.004 0.001 0.012 0.026 0.000	0.031 0.009 0.005 0.019 0.035 0.003	0.424 0.077 0.019 0.231 0.501 0.000
ALL ENGINES Approach Climb Cruise Landing Takeoff Taxi	45 9 4 55 91 5	0.028 0.006 0.002 0.034 0.056 0.003	0.036 0.010 0.006 0.043 0.067 0.007	

This pattern is commonly found in birdstrike and bird ingestion studies. It arises from the fact that airports are typically located in desirable bird environs (vacant land, often near bodies of water). Since the birds congregate around airports there is a greater chance of striking or ingesting a bird during the phases of flight that take place close to the airports. An additional factor contributing to higher ingestion rates in the flight phases close to the ground is the fact that civilian aircraft usually cruise at altitudes well above bird flight routes.

4.2 THE POISSON PROCESS.

The Poisson process is the simplest type of stochastic process that describes how events are distributed in time. The Poisson process is here taken to govern ingestion events, and the times at which these events occur are random. In a Poisson process, the events are distributed somewhat evenly in time so it appears that the times at which the events occurred form a uniform distribution. This section describes some of the properties of Poisson processes that will be useful in describing bird ingestions and in testing hypotheses about bird ingestion rates.

The basis of a Poisson process is a description of the probability distribution of the number of events that occur in a given time interval. The formula for the probability of n events in an interval of length T is:

$$P(X(T) = n) = \frac{e^{-\lambda T} (\lambda T)^n}{n!}$$
 (4.1)

In this equation, the parameter λ is the mean rate at which events occur. Therefore the mean number of events in the time interval of length T is λT . Since hours of operation are not a significant measure of exposure to birdstrikes (the entire cruise portion of the flight is usually at altitudes above those at which birds are found), the time scale used will be number of engine operations rather than hours. Ingestion rates are typically reported in events per 10,000 operations which implies the use of operations as the time scale in a Poisson process.

One way in which the formula for the Poisson distribution can be derived is as the limiting distribution of the binomial distribution for large sample sizes. If the probability of a bird ingestion is the same from flight to flight then the number of ingestions in a large number of flights has a binomial distribution. If the probability of ingestion is p and the number of flights is N then the probability that n ingestions occur in the N flights is:

$$P(X(N) = n) = \begin{bmatrix} N \\ n \end{bmatrix} p^n (1-p) \left(N-n \right)$$
 (4.2)

The binomial probabilities in equation 4.2 can be approximated by a Poisson distribution with mean Np for large values of N. That is, the single flight probability of an ingestion, p, replaces λ in equation 4.1. Past studies [references 5,6] of birdstrikes have used the hypothesis that the probability of a birdstrike is proportional to the cross sectional area of the aircraft. Applying the same hypothesis to engines implies that the bird ingestion rate should be proportional to the cross sectional area of the engine.

The inlet area effect can be incorporated into the Poisson process model by letting the parameter λ represent the ingestion rate per unit area. The probability of n ingestions in N operations for an engine with inlet area A is:

$$P(X(N) = n) = \frac{e^{-\lambda AN} (\lambda AN)^n}{n!}$$
(4.3)

The hypothesis that ingestion rates should be proportional to engine cross section area assumes that birds take no evasive action when approached by an aircraft. That is, the hypothesis assumes that the engine goes through a flock of birds like a cookie-cutter. In reality, birds tuck their wings and drop when they perceive a threat. Hence the critical engine dimension may be engine diameter (vertical height), not cross section area. In that case, the probability of n ingestions in N operations for an engine with engine diameter D is:

$$P(X(N) = n) = \frac{e^{-\lambda DN} (\lambda DN)^n}{n!}$$
 (4.4)

4.3 VALIDITY OF THE POISSON PROCESS MODEL FOR BIRD INGESTION.

The applicability of the Poisson process model can be tested by analyzing the times between ingestions. The interarrival times in a Poisson process are random variables that have independent exponential distributions and the mean time between arrivals is the reciprocal of the ingestion rate. The validity of the Poisson process model can be tested by applying a goodness of fit (GOF) test for the exponential distribution to the times between ingestions.

The GOF test for the exponential distribution is a modified Kolmogorov-Smirnov (K-S) test comparing the observed cumulative distribution function (CDF) to the predicted exponential CDF based on the sample mean. The K-S test uses the test statistic D defined as the maximum vertical distance between the observed and predicted CDFs. A modification to the critical values for the test statistic is required when the predicted CDF is derived from the mean of the sample. The critical values for the modified K-S test were computed by Lilliefors [reference 7]. He presents tables of critical values for sample sizes up to 30, and formulas for approximating the critical values for larger sample sizes.

Because of the small sample size, ingestions for all engines were treated together. A visual comparison of the observed versus theoretical CDFs is presented in figure 4.1. The actual value of D obtained from the observed and theoretical CDFs was 0.065, while the critical value for a probability of 0.01 is 0.133. Hence the hypothesis of an exponential distribution for interarrival times cannot be rejected at the 0.01 level of significance. The use of a Poisson process to model bird ingestions is appropriate based on the results of this test.

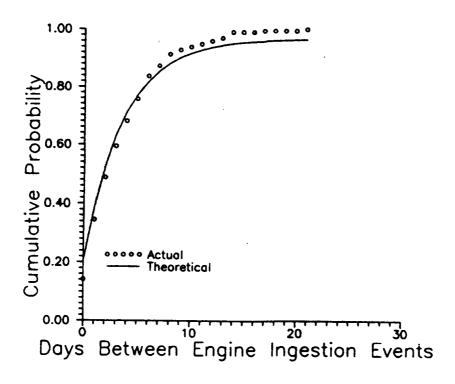


FIGURE 4.1. COMPARISON OF ACTUAL AND THEORETICAL CUMULATIVE DISTRIBUTIONS

4.4 INLET THROAT AREA EFFECT ON INGESTION RATES.

One property of the Poisson process model described in equation 4.3 is that ingestion rates should be proportional to the inlet area of the engine. (Physically, this can be thought of as relating ingestions to the volume swept out by the engine during a flight.) The dimension effect can be investigated for the sample of small engines by comparing actual ingestions with those predicted on the assumption that ingestions will be proportional to both number of operations and inlet throat area.

Because of the difficulty of comparing the inlet throat area for a turboprop engine with the area for a turbofan engine, only turbofan engines are included in this analysis.

The ingestion rate for all turbofan engines in this study is 0.610 engine ingestions/10K operations/10 square ft. inlet area. This rate can be used to compute an expected number of ingestions for each of the individual engines. When a chi-square test is applied to these expected ingestions, the value 0.47 is obtained. The critical value of chi-square for 2 degrees of freedom and probability 0.01 is 9.21. Hence the evidence is strong that the hypothesis of ingestion rate being proportional to engine inlet throat area cannot be rejected.

4.5 INLET THROAT DIAMETER EFFECT ON INGESTION RATES.

As noted above, it may be the case that engine ingestion events are related to engine inlet throat diameter rather than inlet throat area. Under the area hypothesis, an engine of twice the diameter would be expected to ingest four times as many birds. Under the diameter hypothesis, an engine of twice the diameter would be expected to ingest only twice as many birds. The results of testing the diameter hypothesis are presented here.

Because of the difficulty of defining an engine diameter for turboprop engines, where the inlet is wrapped around the propeller spinner, only turbofan engines are included in this analysis. For the turbofan engines, diameter is computed from the published area and an assumed circular cross section.

The ingestion rate for all turbofan engines in this study is 0.110 per ten thousand operations per foot of engine inlet throat diameter. This rate can be used to compute an expected number of ingestions for each of the individual engines. When a chi-square test is applied to these expected ingestions, the value 4.10 is obtained. By chance, the value 9.21 would be exceeded 1 percent of the time. Thus, strictly speaking, we cannot reject the hypothesis of ingestion rate being proportional to engine inlet throat diameter. However, the evidence for this hypothesis is much weaker than the evidence for ingestions being proportional to engine inlet throat area.

SECTION 5 ENGINE DAMAGE DESCRIPTION

Knowledge of the type of damage imposed by a well defined bird ingestion threat is useful in refining bird certification criteria that could lead to improved engine design. This section describes the information available on engine damage. The first part of this section provides descriptions of the types of damage incurred during the period covered by the data and the relationships between engine damage and bird weight, engine damage and phase of flight, engine damage and aircraft airspeed, engine damage and multiple engine and multiple bird involvement. The second part describes the statistical analysis of the relationship between bird weight and the likelihood of damage occurring in an ingestion. The third part describes any unusual crew actions taken as a result of the ingestions. The fourth part describes the engine failures that were due to bird ingestions.

5.1 ENGINE DAMAGE DESCRIPTION.

The types of damage that were identified in the data base were grouped into 14 categories which are defined in table 5.1. During the 2-year data collection period, nine of the damage categories occurred. Tabulations of the occurrences of combinations of damage categories for turbofan engines are presented in table 5.2. The triangular top portion of the table provides tallies of co-occurrences for all pairs of damage categories. The number in the top portion of the table represents the number of events in which both the row damage and the column damage occurred. The events in which more than two types of damage occurred were included in the tallies of the top portion of table 5.2, but they were not specifically identified as involving more than two types of damage. The first row of the bottom two rows of table 5.2 indicates the number of times each damage category was the only damage sustained from a bird ingestion. The second presents the number of times each damage category occurred either as the sole damage or in combination with any other damage category

TABLE 5.1. DEFINITION OF ENGINE DAMAGE CATEGORIES

DAMAGE CATEGORY	SEVERITY LEVEL	DAMAGE DEFINITION
TRVSFRAC	Severe	Transverse fracture - fan blade broken chordwise (across) and piece liberated (includes secondary hard object damage)
CORE	Severe	Bent/broken compressor blades/vanes, blade/vane clash, blocked/disrupted airflow in low, intermediate, and high pressure compressors.
FLANGE	Severe	Flange separations.
TURBINE	Severe	Turbine damage.
BE/DE>3	Moderate	More than three fan blades bent or dented.
TORN>3	Moderate	More than three torn fan blades.
BROKEN	Moderate	Broken fan blades, leading edge and/or tip pieces missing, other blades also dented.
SPINNER	Moderate	Dented, broken, or cracked spinner (includes spinner cap).
RELEASED	Moderate	Released (walked) fan blades (blade retention mechanism broken).
TORN<3	Mild	Three or fewer torn fan blades.
SHINGLED	Mild	Shingled (twisted) fan blades.
NACELLE	Mild	Dents and/or punctures to the engine enclosure (includes cowl).
LEAD_EDG	Mild	Leading edge distortion/curl.
BEN/DEN	Mild	One to three fan blades bent or dented.

TABLE 5.2. TURBOFAN ENGINE DAMAGE CAUSED BY BIRD INCESTIONS

						:		
BEN/DEN	7							
		BEN/DEN						
BE/DE>3	-	0						
•		•	BE/DE>3					
TORN<3	,- 4	-	0					
		ı	•	TORN<3				
BROKEN	0	0	m	0				
					BROKEN			
SHINGLED	0	٣	2	0	: :			
					•	SHINGLED		
TRVSFRAC	0	0	0	0	-	0		
							TRVSFRAC	
CORE	2	~	œ		2		; ; ; ,	
								CORE
NACELLE	0	0	~	0	0		0	; ~

	LEAD_EDG	BEN/DEN	BE/DE>3	TORN<3	OKEN	SHINGLED	TRVSFRAC	CORE	NACELLE
ONLY DAMAGE	·	10	16	; ; ;		0	0	7	2
TOTAL	•	18	27	m	7	S	-	21	4

The amount of data available is not sufficient to make any strong statements about correlations between types of damage. From the lower portion of the table, it can be seen that with the exception of "shingled" and "broken," when a given type of damage occurred, in half or more of the cases it was the only type which occurred (i.e., conditional probability of no other damage exceeds 0.50). "Broken" appeared by itself in only three of seven cases, or slightly less than half; shingled never occurred by itself, but always in conjunction with other kinds of damage.

The TPE331 turboprop engines did not experience any multiple damage category events. Since turboprop engines have no fan stage and no bypass airflow, a bird that is ingested goes directly into the engine core: For this reason the damage that occurred was almost always core damage. Damage to the engine core occurred in 30 events and to the engine nacelle in 1 event. No further specific damage categories were indicated for the TPE331 turboprop engine. A further description of the damage that occurred may be available in the remarks column of the bird ingestion data base (see appendix B) on an individual event basis. It should be noted that in many of the turboprop engine ingestions a blockage of airflow (i.e., primary fuel nozzle/combustor dome flow area, secondary combustion liner diffusion zones) occurred due to the bird debris and there was minor or no physical engine damage.

Tables 5.3 and 5.4 attempt to establish a relationship between the weight of the ingested bird and the resulting engine damage. Table 5.3 shows the number of engine ingestion events with and without reported damage in each specified bird weight range. The damage summaries in table 5.4 for turbofan engines and table 5.5 for turboprop engines were made by tallying the damage codes from the events shown in table 5.3 in each specified bird weight range.

Since many of the engine ingestion events have multiple damage categories, the total number of damage categories does not equal the number of engine ingestion events. Tables 5.4 and 5.5 also show the damage sustained by those engines that were considered to have failed due to the bird ingestion. See Section 5.4 for more information on engine failure.

The amount of data available is insufficient to draw any correlations between the weight of the ingested bird and the type of damage that occurs. However, tables 5.4 and 5.5 show that the majority of the ingestions (31) in which the bird weighed less than or equal to 8 ounces caused no damage. In comparison, all of the birds ingested that weighed more than 24 ounces caused some engine damage.

The relationship between engine damage, phase of flight, and aircraft airspeed is shown in tables 5.6 and 5.7. Table 5.6 depicts the relationship between engine damage and phase of flight. Of the 156 known phase-of-flight engine ingestion events, 48 percent occurred on takeoff and climb and 5 percent of the engine ingestion events that took place during takeoff and climb resulted in engine damage; in comparison, only 47 percent resulted in damage during approach and landing. This appears to establish a relationship between engine speed (thrust) and bird ingestion engine damage since engine speed would typically be higher during takeoff and climb than during approach and landing. It should be noted that the number of engine failures that occurred during takeoff and climb were only one greater than the engine failures that occurred during approach and landing.

TABLE 5.3. TALLY OF POSITIVELY INDENTIFIED BIRD SPECIES BY WEIGHT RANGE AND ENGINE TYPE

Weight Range (oz.)	Bird Indenti Turbofan	fications* Turboprop
$0 < x \leq 8$	41	7
$8 < x \le 16$	13	2
$16 < x \le 24$	4	1
$24 < x \le 32$	0	1
$32 < x \le 40$	1	2
x > 40	6	0
Totals	65	13

^{*}One counted for each engine ingestion event

TABLE 5.4. BIRD INGESTION TURBOFAN DAMAGE SUMMARY

Severity	Damage Category		Bird	Weight Range	(oz.)		
		(0 ⟨x ≼ 8)	(8 ⟨ x≤16)	(16 ⟨ x < 24)	(24 ⟨ x <u></u> ≤32)	(32⟨ x <u>≤</u> 40)	(x > 40)
	None	27	5	1	0	0	0
	Damage Unknown	1	1	0	0	0	0
	Other	3/1*	4	0	0	0	0
Mild							
	Lead-Edg Shingled Ben/Den Torn 3 Nacelle	0 1 6/1* 1	2 0 3 0 1	0 1 0 0	0 0 0 0	1 0 0 1 0	1/1* 1 1/1* 1/1* 1/1*
Moderate							
	Be/De > 3 Torn > 3 Broken Spinner Released	3 0 0 0	2 0 0 0 0	3 0 0 0	0 0 0 0	0 0 0 0	3/2* 0 2/2* 0 0
Severe							
	Trvs Frac Core Flange Turbine	0 3 0 0	0 3 0 0	0 0 0 0	0 0 0 0	0 0 0 0	1/1* 6/4* 0 0

^{*}Number of occurrences/number of occurrences when engine failed

TABLE 5.5. BIRD INGESTION TURBOPROP DAMAGE SUMMARY

^{*}Number of occurrences/number of occurrences when engine failed

TABLE 5.6. PHASE-OF-FLIGHT (POF) ANALYSIS

Aire	nown POF craft Events/ ine Ingestions (144/156)	Known POF Damaging Aircraft Events/ Engine Ingestions (87/92)	Known POF Engine Failure Ingestions (9)
Takeoff and Climb	68/75	52/56	5
Approaching and Landing	65/70	32/33	4

TABLE 5.7. AIRCRAFT AIRSPEED ANALYSIS

Aircraft Airspeed	Known Speed Engine Ingestions (123)	Known Speed Damaging Engine Ingestions (73)	Known Speed Damaging Engine Ingestions, Takeoff and Climb (42)	Known Speed Damaging Engine Ingestions, Landing and Approach (27)
< 140 Knots	87	50	30	18
≥ 140 Knots	36	23	12	9

Table 5.7 shows the number of engine ingestion events and the number of damaging engine ingestions known to have occurred below 140 knots airspeed and at or above 140 knots. The table also shows the phase of flight that these damaging engine ingestions occurred in those airspeed ranges. There were seven percent more engine ingestions that resulted in engine damage at or above 140 knots airspeed than those that occurred below 140 knots. It is also shown that a greater number of damaging ingestions occurred during takeoff and climb than during approach and landing at both aircraft airspeed ranges.

Multiple engine and multiple bird ingestion events present the greatest safety hazard to aircraft. Table 5.8 shows the number of these events that occurred. Eleven aircraft had bird ingestions into more than one engine during the same event, and four events resulted in damage to more than one engine. There were also four events where multiple birds were ingested into more than one engine, potentially the most hazardous condition an aircraft can encounter.

Table 5.8 also gives the number of engine ingestion events where more than one bird was ingested into the engine. Of the 30 multiple bird engine ingestions that occurred, 77 percent of the ingestions resulted in some engine damage. In comparison, only 46 percent of the engines that ingested a single bird resulted in some engine damage. Ten percent of the multiple bird ingestions resulted in engine failures compared to only four percent of the single bird ingestions.

5.2 PROBABILITY OF DAMAGE.

One of the key questions which inspired the bird ingestion survey is the issue of what weight bird should be simulated in certification testing. Two of the main issues in deciding what the certification bird weight should be are (1) the likelihood of ingesting a bird of that weight or heavier and (2) the likelihood that damage will result from ingesting a bird of the certification weight. The issue of bird weights is discussed in Sections 3 and 7 while the probability of damage is the topic of this section.

In general, the heavier the bird ingested, the greater the engine damage. However, the problem of relating bird weight to engine damage is made more complicated by the fact that in a few cases small birds caused considerable engine damage, while in other cases large birds were ingested with no engine damage. Figure 5.1 illustrates the variation in damage for turbofan engines. For the lowest weight range, there was one case of severe damage and two cases of mild or unspecified damage. All other ingestions resulted in no reported damage. With increasing bird weight, the proportion of ingestion events resulting in severe damage increased, as did the proportion of ingestion events resulting in mild or moderate damage. In the heaviest weight range, there were only four ingestion events (out of 21 total for this weight range) which resulted in no damage.

For the turboprop engine, the situation is somewhat different because of damage definitions that are different from those used for turbofans. Regardless of bird weight, there were no instances of damage being classified as more severe than mild. In 11 ingestion events, including 5 in the highest weight range, damage was limited to mild. There was one ingestion event in the highest weight range which resulted in no damage.

TABLE 5.8. MULTIPLE ENGINE AND MULTIPLE BIRD ANALYSIS

	Aircraft Events/ Engine Ingestions	Damaging Engine Ingestions	Engine Failure Ingestions
Multiple Engine	11/23	12/4*	0
Multiple Bird	26/30	23	3
Single Bird	175/180	83	7

 $[\]mbox{*Aircraft}$ events where more than one engine damaged

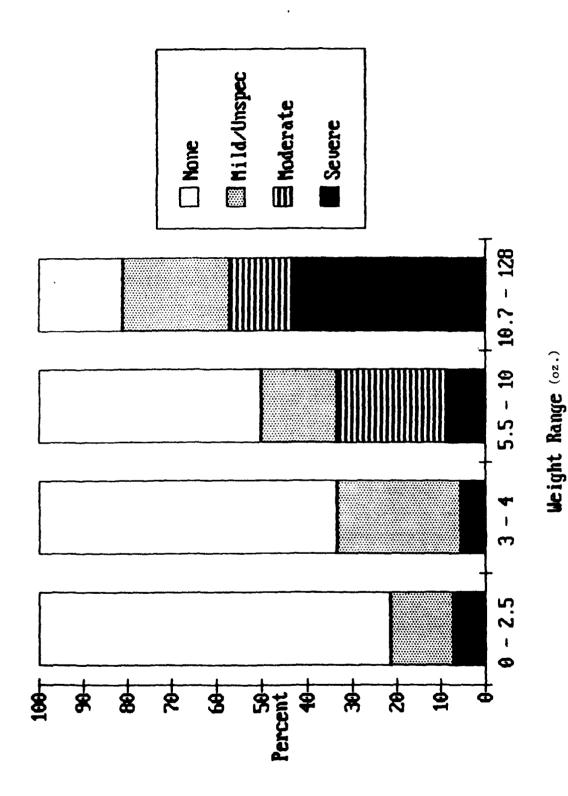


FIGURE 5.1. SERVERITY OF DAMAGE FOR TURBOFAN ENGINES VERSUS BIRD WEIGHT RANGE

This situation is similar to bioassay experiments, in which a continuous variable (dose size) produces a discontinuous result (cure/no cure, cancer/no cancer, etc.). In such experiments, it is usually found that a small dose produces the effect in a few experimental subjects, while a large dose produces the effect in many subjects. It would be more convenient, of course, if there were a threshold dose such that below the threshold, no experimental subjects showed any effect, while above the threshold all experimental subjects showed the effect. Since there is no such unique threshold, the bioassay experiments are then analyzed in terms of the probability that a given dose size will produce the response.

We have chosen to use the same method of analysis for the bird ingestion data because it has the same characteristics as bioassay data: a small "dose" may cause damage, but the likelihood of damage is greater with larger "doses." Our approach is to compute the probability of damage (POD) as a function of bird weight. The key elements are that the probability of success for a Bernoulli trial is related to a continuous stimulus variable. In bird ingestion, the Bernoulli trial is whether or not damage occurs and the stimulus variable is the weight of the ingested bird.

Linear logistic analysis is the most commonly used method of analyzing the dosage-response type of data. It is used not only in bioassay experiments, but in transportation studies involving choice of transportation mode. It has also been used successfully in relating the probability of transparencies breaking as a function of projectile size in dealing with the problem of propwash blown gravel breaking helicopter windshields. In that case, the transparency is sometimes broken by small stones; yet in other cases, it survives impact by large stones. Nevertheless, heavier stones have a greater probability of breaking the transparency. The logistic distribution function serves as the basis for the linear logistic analysis. There are several ways in which the logistic distribution function can be parameterized. The one we used is given by:

$$POD(w) = 1/(1+exp[-(\pi/\sqrt{3})(\ell n(w)-\mu)/\sigma])$$
 (5.1)

In this parameterization, w is the bird weight, μ represents the mean logarithm of bird weight, and σ is a parameter that is related to the steepness of the POD function. This parameterization is selected because of its similarity to the usual parameterization of the familiar Normal probability distribution. The logistic probability density is symmetrical about the mean μ . Therefore μ is not only the mean, it is also the median and the mode of the distribution. In particular, it is the logarithm of the bird weight with a 50 percent chance of causing damage.

The estimation of the function given in equation 5.1 has been extensively studied, and the methods have been described in the literature (see references 8 and 9). The method of maximum likelihood provides the best estimates for the type of data in the bird ingestion study since there are only a few ingestions at each weight. The software for estimating the parameters of equation 5.1 has been developed and extensively tested at the UDRI and verified by researchers at other institutions.

The types of damage were categorized as mild, moderate, or severe by the FAA. (Actual data are presented in appendix B.) Three distinct analyses were conducted based on the severity ratings. The three analyses estimated the probability of any damage at all, the probability of at least moderate damage,

and the probability of severe damage. Figures 5.2, 5.3, and 5.4 show the estimated POD functions along with confidence bounds on the POD functions for the analyses.

Figure 5.2 shows the probability of any damage occurring and includes all three severity levels as positive responses, including unspecified damage levels. The probability of any damage occurring rises steeply at first, then flattens out. There is a significant probability of damage at 20 ounces, and almost 90 percent probability of damage at 100 ounces.

Figure 5.3 shows the probability of at least moderate damage. The probability of moderate damage does not rise quite as steeply as the probability of any damage. The probability of damage reaches almost 90 percent at weights of 100 ounces.

Figure 5.4 shows the probability of severe damage. The probability of severe damage reaches about 65 percent at a weight of 100 ounces. The rise is much less steep than the two preceding curves, being almost linear.

The sample size appears to be large enough that the estimates of damage probability are reliable. Moreover, as shown in Section 3, there seems to be no relationship between severity of engine damage and the likelihood that bird weight was determined (through identification of species). Hence, there is no reason to believe that the estimates of probability of damage are biased either upward or downward from this cause.

5.3 CREW ACTION DESCRIPTION.

Two other factors that relate to the severity of engine damage are whether or not a crew action is required (aborted takeoff (ATO), air turnback (ATB), or diversion (DIV)) and whether or not the engine was shut down (IFSD) as a result of the ingestion. Table 5.9 presents the conditional probabilities that a crew action is required given the severity of the damage that the engine incurs [P(CAD)]. The probability that a crew action is required increases with the severity of engine damage as would be expected. The third column of table 5.9 contains the upper 95 percent confidence bound on the conditional probabilities presented in the second column.

A crew-initiated in-flight engine shutdown occurred in seven of the 210 engine ingestion events. There was one involuntary in-flight shutdown of a turbofan engine, and three involuntary in-flight shutdowns of a turboprop engine. This corresponds to an estimated conditional probability of an involuntary in-flight shutdown of 0.019 with a 95 percent confidence bound of 4.359 x 10^{-2} . Given the small sample size, and only 16 total instances of in-flight shutdown, no inferences can be drawn about the causes of in-flight shutdowns.

5.4 ENGINE FAILURE.

Engine failures are important areas to consider when analyzing these engine bird ingestion events. For the purpose of this study an engine failure was considered to have occurred when an engine was not able to produce and maintain usable thrust of at least 50 percent. A transverse fan blade fracture and an involuntary engine in-flight shutdown were considered to be engine failures in all cases. Otherwise, an engineering judgment was made based on the extent of engine damage, effect on flight, phase of flight, and any other factors that may have been provided in the description of the event or investigation summary.

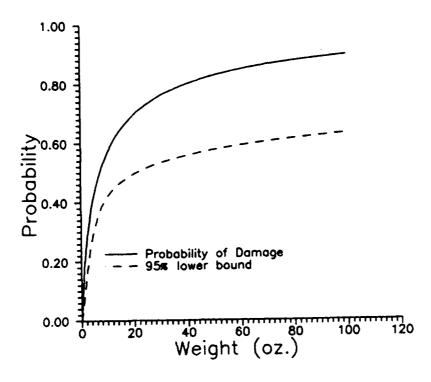


FIGURE 5.2 PROBABILITY OF ANY DAMAGE VERSUS BIRD WEIGHT

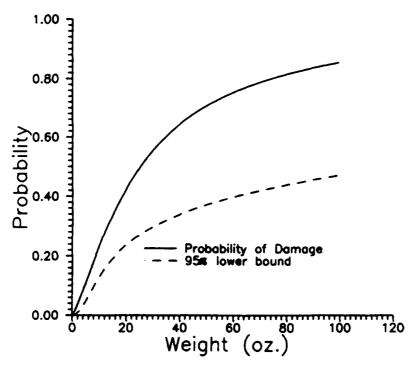


FIGURE 5.3 PROBABILITY OF AT LEAST MODERATE DAMAGE VERSUS BIRD WEIGHT

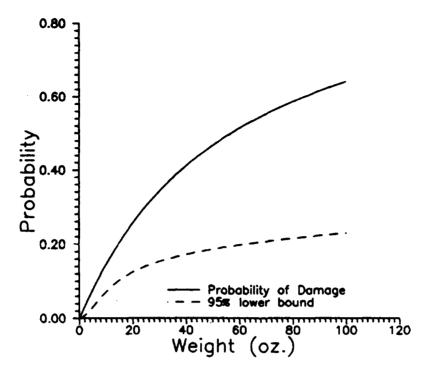


FIGURE 5.4. PROBABILITY OF SEVERE DAMAGE VERSUS BIRD WEIGHT

There were ten ingestion events which resulted in engine failure, ranging from partial power loss, through voluntary shutdown, to involuntary shutdown. The number of cases is too small for any patterns to be apparent. However, some summary is possible. Table 5.10 provides a summary of some of the important data categories for the engine ingestion events that resulted in an engine failure. Overall, five percent of the engine ingestion events resulted in an engine failure. The turbofan engine failure rate was 0.01 failures per ten thousand aircraft operations, and the turboprop engine failure rate was 0.004 failures per ten thousand aircraft operations.

Table 5.10 shows that a voluntary or involuntary in-flight shutdown of the engine occurred in eight of the ten engine failures. There was also a power loss associated with all of the engine failures where there was information reported in the power loss category. The only relationship that appears between the damage codes of these engine failures is that in all but one event there was core damage.

Reviewing the bird threat data for these engine failures shows that seven of the engine failures were caused by the ingestion of a single bird and three were caused by the ingestion of two birds. This is a much higher percentage than the fraction of all ingestion events which involved multiple birds, suggesting that engine failure is more likely in cases of multiple bird ingestion. Also, in four of the six engine failures where the bird weight was known the bird or birds weighed more than four pounds. However, the other two were caused by single birds that weighed less than 8 ounces. Comparing this with the number of engine ingestions where the bird was positively identified (table 5.3) shows that 83

TABLE 5.9. CONDITIONAL PROBABILITY OF CREW ACTION AND IN-FLIGHT SHUTDOWN GIVEN THE ENGINE DAMAGE SEVERITY

P(IIFSD D)	0.000 0.014 0.024 0.053	0.000 0.075 0.000 0.000
Involuntary Inflight (<u>IFSD D) Shutdowns</u>	1110	0 11 0 0
] P(IFSD[D)	0.000 0.072 0.098 0.158	0.080 0.225 0.500 0.000
Inflight <u>Shutdowns</u>	0 % 4 %	0 1 6 5
Upper 95% Confidence Bound	0.138 0.556 0.738 0.760	0.366 0.577 1.000 1.000
P(CA D)	0.066 0.406 0.512 0.421	0.160 0.375 0.000 0.000
Instances of Crew <u>Action*</u>	5 28 21 8	4 15 0
Engine Ingestion Events	76 69 41 19	25 40 2 2
Severity of Engine <u>Damage</u>	Turbofans: None Any Mod/Severe Severe	Turboprop: None Any Mod/Severe Severe

* Crew action includes Aborted Takeoff, Air Turnback, Diversion

IFSD - In flight shutdown

TABLE 5.10. ENGINE FAILURE SUMMARY BY BIRD WEIGHT

Bird (oz.) Weight	Number of Birds	Damage <u>Code</u>	Phase of Flight	Power Loss	In-Flight Shutdown	Crew Action
128	2	A,G,I,K	Takeoff	Flame Out	Yes	ATO
102	1	A,D,G,K	Takeoff	Momentary	No	ATB
88	2	A,B,D,K	Landing	Yes	No	None
64.5	1	A,C,E,K	Takeoff	Compressor	Involuntary	ATB
7.7	1	A,K	Approach	Spool Down	Involuntary	None
1.5	1	A,C,P	Takeoff		Vibes	DIV
	1	A,D,K,P	Unknown		Yes	
and the same	1	A,K	Approach	Flame Out	Involuntary	
	1	A,K	Approach	Spool Down	Involuntary	None
~	2	A,K	Takeoff	50%	Voluntary	ATB

Note: A description of the columns and column contents can be found in appendix ${\tt B.}$

percent of engine ingestion events, where the bird ingested weighed more than 4 pounds, resulted in engine failures, whereas only four percent of the events, where the bird ingested weighed less than 1/2 pound, resulted in engine failures.

In the six engine failure events in which weight of the ingested birds were known, the average weight was 65.3 ounces, which is much higher than either the median or the mode for ingested bird weights. That is, in cases of engine failure, the ingested bird typically was heavier than the average for all bird ingestion events. Note that the figure given above is for average weight of each ingested bird, not average ingested weight, since some of the engine failure events involved multiple ingestions. This finding is not unexpected, since a heavier bird would be expected to result in greater damage.

The failures were split almost evenly between takeoff (five engine failure events) and approach/landing (four engine failure events). (One event was not identified as to phase of flight.) For the nine engine failure events in which weather conditions are known, the sky was clear (seven cases) or had scattered clouds (two cases). This implies that weather was not a factor in engine failure.

For the nine engine failure events in which lighting conditions were known, two occurred in the dark, one at dawn, and five in light conditions. This implies that illumination was not a factor in engine failures.

The findings on weather and lighting conditions, taken together, imply that lack of visibility was not a factor in the engine failures. This is probably to be expected, since aircraft are not permitted either to land or take off in low visibility conditions, and only one of the engine failures occurred at an altitude above 1000 feet. Thus, the fact that the aircraft were flying at all would imply that visibility was acceptable at low altitude.

A final finding regarding engine failures is that in the seven of nine engine failure events in which engine location is known, the failed engine was located on the right side of the aircraft. This presents a strong contrast with the distribution of engine ingestion events where engine location is known: 98 on the right, 101 on the left, and 3 in the center. That is, for all engine ingestion events, the location is consistent with the hypothesis that engines on the left and on the right are equally likely to ingest birds. The distribution of locations for engine failures has a probability of only 0.10, and is not consistent with that hypothesis. One possible explanation is that pilots, who sit on the left, are able to see and avoid those large birds which seem to be responsible for engine failure. However, given the small number of engine failure events, this possibility is little better than pure speculation. While no convincing explanation can be offered for the discrepancy, it may be significant.

SECTION 6 PROBABILITY ESTIMATES

This section provides a summary of the probabilities of various engine ingestion events. The probability of an event is a measure of the likelihood that the event will occur. The probabilities in this section are calculated on a per engine operation basis and present information similar to the ingestion rates. The ingestion rates that were presented in Section 4 were calculated on the basis of 10,000 engine operations. In that section, it was shown that the ingestions did follow a Poisson distribution. As a consequence of the Poisson distribution, the ingestion rate per engine operation is equal to the probability of ingestion for a single operation. This section provides more details on the probabilities of various categories of bird ingestion events.

Table 6.1 provides the estimated probabilities and 95 percent confidence bounds for the entire small engine population for various bird ingestion events including all flight phases, multiple bird ingestions, and ingestions where the damage was moderate or severe. Note that one ingestion event was not identified as to location. Therefore the United States and foreign events do not add to the total for all phases.

The overall likelihood of a bird ingestion event in a single operation is about 1.3 in 100,000 thousand. Although this probability is very low, there are sufficient operations per year (over 1.6 million during the period covered by the data) that the expected number of ingestions is roughly 200. Most ingestions occur during takeoff or landing phases, so the probabilities for those phases are larger than for other phases of flight. Multiple bird ingestion events are comparatively rare, and this is reflected in the lower probabilities for these events.

Table 6.2 shows the probability of ingestion by bird weight range and location. This is computed by multiplying the overall probability of ingestion per operation for each of the regions (United States, foreign, worldwide) by the frequency of each bird weight range. The validity of this calculation is dependent on the randomness of bird identification. As discussed in Section 3, there appears to be no reason to believe that the probability of a bird being identified is correlated with degree of engine damage; hence, the assumption of randomness appears justified.

Table 6.3 shows the probability of an ingestion by bird weight range for each engine type and region (United States, foreign, worldwide). As with table 6.2, this is computed by multiplying the overall probability of ingestion per operation for each of the regions, computed separately for each engine type, by the frequency of each bird weight range. The same caveat applies as to randomness of bird identifications.

Table 6.4 shows the probability of ingestion by phase of flight for each engine type by region. It also shows the probability of multiple bird ingestions in the same engine, the probability of multiple engine ingestions, and the probability of moderate or severe damage. The table is computed by dividing the number of engine ingestion events in each of the conditions by the number of operations for the particular engine type in each region. Note that one ingestion for the TFE731 was not identified as to location. It is included in the world total for all flight phases but not in either United States or foreign ingestions.

TABLE 6.1. ENGINE INGESTION PROBABILITIES

	IABLE 0.1.	FIGURE THE		
CONDITION	ING	NGINE ESTION VENTS	PROBABILITY OF INGESTION	UPPER 95% CONFIDENCE BOUND
	•			
All Phase	S			1.460E-05
World		210	1.302E-05	1.216E-05
US		113	1.040E-05	2.300E-05
Foreign		86	1.922E-05	2.300E-03
Approach		45	2.791E-06	3.578E-06
World		45	2.853E-06	3.851E-06
US		31	2.683E-06	4.346E-06
Foreign		12	2.0032 00	
Climb		9	5.582E-07	9.741E-07
World		5	4.602E-07	9.676E-07
US		3	6.706E-07	1.733E-06
Foreign		3	0.7002 0.	
Cruise			0 4015 07	5.677E-07
World		4	2.481E-07	4.366E-07
US		1	9.204E-08	1.060E-06
Foreign		1	2.235E-07	1.0602-06
Landing				4.270E-06
World		55	3.411E-06	3.107E-06
US		24	2.209E-06	
Foreign		31	6.930E-06	9.353E-06
Takeoff				6.719E-06
World		91	5.644E-06	6./19E-06
US		52	4.786E-06	6.030E-06
Foreign		34	7.601E-06	1.012E-05
_				
Taxi		5	3.101E-07	6.520E-07
World		ō	0	2.757E-07
ປຣ		5	1.118E-06	2.350E-06
Foreign		5	1(1102 00	
Multiple	Birds		1.861E-06	2.524E-06
World		30		2.126E-06
บร		15	1.381E-06	5.163E-06
Foreign		15	3.353E-06	5.1052 00
Moderate	e to Severe	Damage		
Turbofa			C 0018-06	8.941E-06
World		41	6.891E-06	5.790E-06
US		14	3.703E-06	2.354E-05
Foreign		23	1.662E-05	Z.354E-05
Turbopr	ops	_	1 0000-07	6.188E-07
World	-	2	1.966E-07	6.696E-07
บร		1	1.412E-07	1.536E-06
Foreign		1	3.237E-07	1.3365-06
				•

Note: JT15D engine excluded in US and Foreign conditions

TABLE 6.2 PROBABILITY OF AN ENGINE INCESTION EVENT VS. BIRD WEIGHT

Weight 1	U.S. Events	U.S. Prob	Foreign <u>Events</u>	Foreign <u>Prob</u>	Unknown	Worldwide <u>Events</u>	Worldwide <u>Prob</u>
0 <x<4< th=""><th>23</th><th>2.117E-06</th><th>œ</th><th>1.788E-06</th><th>1</th><th>32</th><th>1.985E-06</th></x<4<>	23	2.117E-06	œ	1.788E-06	1	32	1.985E-06
4 <x≤8< th=""><th>7</th><th>1.841E-07</th><th>7</th><th>1.565E-06</th><th>0</th><th>6</th><th>5.582E-07</th></x≤8<>	7	1.841E-07	7	1.565E-06	0	6	5.582E-07
8 <x<12< th=""><th>Э</th><th>2.761E-07</th><th>2</th><th>4.471E-07</th><th>0</th><th>2</th><th>3.101E-07</th></x<12<>	Э	2.761E-07	2	4.471E-07	0	2	3.101E-07
12 <x<16< th=""><th>9</th><th>5.523E-07</th><th>4</th><th>8.942E-07</th><th>0</th><th>10</th><th>6.202E-07</th></x<16<>	9	5.523E-07	4	8.942E-07	0	10	6.202E-07
16 <x≤20< th=""><th>2</th><th>1.841E-07</th><th>2</th><th>4.471E-07</th><th>0</th><th>7</th><th>2.481E-07</th></x≤20<>	2	1.841E-07	2	4.471E-07	0	7	2.481E-07
20 <x<24< th=""><th>-1</th><th>9,204E-08</th><th>0</th><th>0</th><th>0</th><th>1</th><th>6.202E-08</th></x<24<>	-1	9,204E-08	0	0	0	1	6.202E-08
28 <x<32< th=""><th>1</th><th>9.204E-08</th><th>0</th><th>0</th><th>0</th><th>-4</th><th>6.202E-08</th></x<32<>	1	9.204E-08	0	0	0	-4	6.202E-08
32 <x≤36< th=""><th>0</th><th>0</th><th>-</th><th>2.235E-07</th><th>0</th><th></th><th>6.202E-08</th></x≤36<>	0	0	-	2.235E-07	0		6.202E-08
36 <x≤40< th=""><th>2</th><th>1.841E-07</th><th>0</th><th>0</th><th>0</th><th>2</th><th>1.24E-07</th></x≤40<>	2	1.841E-07	0	0	0	2	1.24E-07
64 <x≤68</x	7	9.204E-08	0	0	0	7	6.202E-08
84 <x<88< th=""><th>2</th><th>1.841E-07</th><th>0</th><th>0</th><th>0</th><th>2</th><th>1.24E-07</th></x<88<>	2	1.841E-07	0	0	0	2	1.24E-07
100 <x<104< th=""><th>1</th><th>9.204E-08</th><th>0</th><th>0</th><th>0</th><th>1</th><th>6.202E-08</th></x<104<>	1	9.204E-08	0	0	0	1	6.202E-08
124 <x≤128</x	2	1.841E-07	0	0	0	7	1.24E-07

1 Ounces

TABLE 6.3. PROBABILITIES OF AN ENGINE EVENT* AS A FUNCTION OF BIRD WEIGHT, LOCATION, AND ENGINE TYPE

		ALF502			TFE731			TPE331		JTISD
	S.U	FOREIGN	WORLDWIDE	U.S.	FOREIGN	WORLDWI DE	S. U	FOREIGN	WORLDWIDE	WORLDWIDE
Engine Operations:	1,187,981	415,354	1,603,335	2,592,274	968,802	3,561,077	7,084,288	3,089,229	10,173,518	785,259
Bird Wt Range	Prob of Ingestion	Prob. of Ingestion	Prob. of Ingestion	Prob. of Ingestion	Prob. of Ingestion					
(7	1.010	0.963	866.0	0.231	0.413	0.309	0.071	:	670.0	:
(8 ×××)	:	0.482	0.125	0.077	0.310	0.140	:	0.065	0.020	:
$(8 < X \le 12)$	780.0	0.241	0.125	0.077	0.103	0.084	:	:	:	:
(12 < X ≤ 16)	:	;	:	0 154	0.310	0.197	0.014	0.032	0.020	0.127
(16 < X s 20)	;	0 241	0.062	0.077	:	950.0	:	0.032	0.010	:
(20 < X ≤ 24)	:	:	:	0.039	;	0.028	:	:	:	:
(24 < X ≤ 28)	;	:	:	:	:	:	0.014	;	0.010	÷
(32 < X ≤ 36)	:	;	;	:	:	:	;	0.032	0.010	:
(36 < X ≤ 40)	:	:	:	0.039	:	0.028	0.014	:	0.010	:
(89 S X > 79)	780.0	:	0.062	:	:	:	:	:	;	:
(88 ≥ X > 78)	780 0		0.062	0.039	:	0.028	;	:	:	:
(100 < X ≤ 104)	:	•	:	0.039	:	0.028	:	:	:	:
(124 < X ≤ 128)	:	;	:	0.077	;	0.056	:	:	:	:
All Events	1.263	1.926	1.435	678.0	1.135	0.955	0.113	0.162	0.128	0.127

* Ingestion probabilities scaled by 10*

TABLE 6.4. ENGINE INGESTION PROBABILITIES* BY ENGINE AND LOCATION

			ALF502	05					TFE/31	31					TPE331	31		:	JTISD	۵ :
	U.S.		U.S. FOREIGN	;	WORLDWIDE		U.S.		FOREIGN	:	WORLDWIDE	 DE	U.S.	: :	FOREIGN	:	WORLDWIDE		WORLDWIDE	IDE
						;	:	:		;		:						: :	7050	
Engine Operations:	1,187,981	981	415,354		1,603,335	335	2,592,274	,274	968,802		3,561,077	11	7,084,288		3,089,229		10,173,518	8	607'09/	607
,						;		;				:		:		:		:		;
Condition Under Consideration	Ingestion Evt Prob.	ë d	Ingestion Ext Prob.	ion (gb.	Inge	Ingestion Evt Prob.	Ingestion Evt Prob.	t ion	Ingestion Evt Prob.	ion	Ingestion Evt Prob.	Ingestion Evt Prob.	Ingestion Evt Prob.	ion gb.	Ingestion Evt Prob		Ingestion Ext Prob.		Ingestion Evt Prob.	lon do
All Flight Phases	34 2.86	3	29 6.	96.9	63	3.93	37	37 1.43	34 3.51	1.51	72	2.02	45	0.59	23 0.74	. 74	0 59	99.0	10 1.27	. 27
Takeoff And Climb Phases	15 1.26	. 26		1.69	22	1.37	16	0.73	23 2	2.37	42	1.18	23 (0.32	7	0.23	30	0.29	9	0.76
Approach And Landing Phases	19 1.60	. 60	81	4,33	33	2.31	17	99.0	10	1.03	27	97.0	19	0.27	15 (64.0	34 (0.33	~	0.25
Duel Engine - Single Bird Events	3	0.25	0	:	3	0.19	0	;	-	0.10	7	0.03	0	:	-	0.03	-	0.01	0	1
Multiple Birds Single Engine Events		1 0.08	7	0.24	2	0.12	٠	0.35	4	0.41	13	0.37	~	0.03	7	90.0	4	90.0	0	:
Multiple Birds - Dual Engine Events		90.0	-	0.24	7	0 12		0.04	7	0.21	~	80.0	0	:	-	0.03		0.01	0	:
Moderate Or Severe Demage	6	3 0.25	2	0.48	\$	0.31	=	0.42	21	2.17	32	06.0	-	0.01	-	0.03	7	0.02	4	0.51

Ingestion probabilities scaled by 10^5

SECTION 7 DATA QUALITY

The interpretations derived from any large set of data are only as good as the data. The use of poor data can lead to invalid and misleading conclusions. The conclusions reached in this report should be interpreted in the context of the sources of the data and the quality of the data. The following paragraphs discuss the sources of data for the first 2 years and the quality of the data as measured by the consistency of the data collected in the first and second years.

7.1 DATA SOURCES.

The data used in this report were collected by the engine manufacturers and supplied to the FAA. The data were in turn supplied to the University of Dayton by the FAA. The method of data collection was a census rather than a survey sample. That is, the goal was to collect information on every bird ingestion event affecting the four engines in the study, during the 2-year period (second year only for the JT15D). A complete census is nearly impossible to achieve under any circumstances; therefore, estimates involving the total number of ingestions, such as ingestion rates, should be viewed as lower bounds. Other than the possibility that some ingestion events escaped the census, there were no known problems which systematically affected the reliability of the data.

7.2 INTERNAL CONSISTENCY.

The data collected during the second year should be consistent with the data collected during the first year, if the two data sets are to be combined. Hence it is necessary to compare the two data sets for consistency. This is done below, with two different tests being applied.

The first test compares the ingestion rates (ingestions per operation) for each engine for the first year and for the two years. Section 4 provided evidence that aircraft ingestion events occur according to a Poisson process so that a Z test can be used to compare the two. According to the properties of a Poisson process, the proportion of events that were recorded in the first year should be equal to the proportion of operations that were conducted in the first year.

The formula for the expected proportion of events in the first year becomes

$$P = 01/(01 + 02) \tag{7.1}$$

where 01 and 02 are the number of operations for a particular engine in the first and second years, respectively. The proportion of aircraft ingestion events in the first year is used as P along with P as defined above, in the equation for Z

$$Z = (P - P)/SORT(P*(1 - P)/N)$$
 (7.2)

where N is the total number of ingestion events for the engine.

The Z statistic defined in equation 7.2 is used to test the null hypothesis that there is no difference between the ingestion rates of a given engine between the first year and the two years taken together. Table 7.1 gives the results of the analysis. Any type of change, either increase or decrease, is important. Hence a two-sided test should be used. The critical value for a two-sided test and 5

TABLE 7.1. COMPARISON OF INCESTION RAITES FOR FIRST AND SECOND YEARS

Engine	ir.	First Year	Bot	Both Years	1	< ۵	
	Events	Events Operations	Events	Operations	д	24	7
			u.s.				
ALF502 TFE731 TPE331	12 23 19	596905 1279280 3389810	34 37 42 Foreign	1187981 2592994 7084288	0.502 0.493 0.478	0.353 0.622 0.452	-1.744 1.559 -0.339
ALF502 TFE731 TPE331	14 17 14	125013 471185 1351306	29 34 23	415354 968802 3089229	0.301 0.486 0.437	0.483	2.134 0.159 1.656

percent significance is ± 1.96 . As the table shows, only one of the Z values exceeds the ± 1.96 bound. Considering that we have performed six tests, each of which has probability 0.05 of falling "out of bounds" by pure chance, there is actually one chance in four that at least one of the six tests will fall out of bounds by pure chance. The fact that one test did exceed the limit cannot be considered strong evidence that the data are inconsistent from the first to the second year.

Another check on the consistency of the data collection is to compare the birds that were identified in the 2 years. There were too many different species and locations of ingestions, and too few of each species or location, to allow comparisons of those features. However, if the species identifications are reduced to bird weights, the cumulative weight distributions for the first and second years can be compared. Table 7.2 provides the cumulative bird weight distributions for the first and second years, worldwide. The data are plotted in figure 7.1 to provide a visual comparison. As can be seen from both the table and the figure, there are substantial differences between the distributions at the low end.

A statistical measure of the closeness of the cumulative distributions is the Kolmogorov-Smirnov D test. The D statistic is compared to a test value based on the sizes of the two samples. When the D statistic is smaller than the test value, the distributions are considered to be similar at a given significance level.

The maximum difference between the distributions in figure 7.1 is 0.413. For the sample sizes, this maximum difference should be less than 0.387 at a significance level of 0.01. The conclusion is that with a possible chance of error of 1 in 100, the two cumulative distributions are significantly different. Hence by this test, the data in the 2 years are not consistent.

In summary, the tests have found some significant differences between the data sets collected in the first and the second years. However, this need not be attributed to faults in data collection. It might also be due to changes in aircraft operational patterns or to changes in bird habits. The information available is not sufficient to distinguish between these alternative possibilities.

TABLE 7.2. CUMULATIVE DISTRIBUTIONS, FIRST AND SECOND YEARS

Weight (oz)	Year 1	Year 2
4	0.235	0.649
8	0.382	0.757
12	0.471	0.811
16	0.647	0.919
20	0.735	0.946
24	0.765	0.946
28	0.794	0.946
36	0.824	0.946
40	0.853	0.973
68	0.882	0.973
88	0.912	1
104	0.941	1
128	1	1

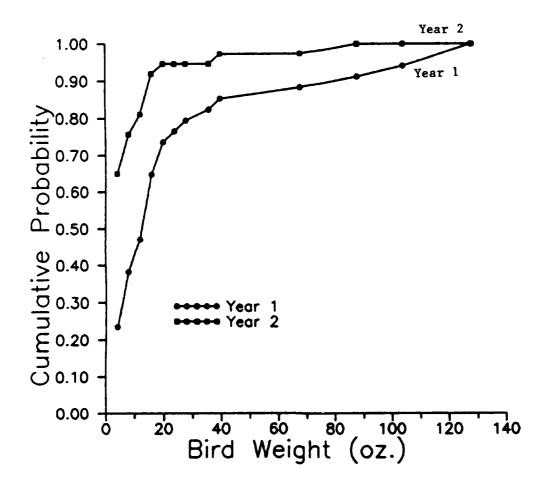


FIGURE 7.1. COMPARISON OF BIRD WEIGHT DISTRIBUTIONS, FIRST AND SECOND YEARS

SECTION 8 CONCLUSIONS

This section summarizes conclusions from the data collected.

Bird Descriptions

- . Gulls, doves, and lapwings are the birds most often ingested.
- Eighty-six percent of the birds that were positively identified by an ornithologist weighed less than or equal to one and a half pounds. In comparision ninety-two percent weighed less than or equal to two and a half pounds.
- . Fourteen percent of the engine ingestion events are multiple bird ingestions.
- . Six percent of the aircraft ingestion events are multiple engine events.
- . The identification rate does not seem to vary with degree of engine damage.
- . The weight of a bird most likely to be ingested outside the United States is approximately twice as heavy as one ingested within the United States.
- . Ingestions are least likely to occur at night.

Ingestion Rates

- . The foreign engine bird ingestion rates are higher than the United States rates.
- . Bird ingestion events can be modeled as a randomly variable Poisson process.
- . Bird ingestion rates are proportional to the engine inlet throat cross section area.
- . Turbofan engines had a higher ingestion rate than the turboprop engine.

Effect on Flight

- . Six percent of all aircraft ingestion events result in an aborted takeoff, fourteen percent result in an air turnback, and three percent result in an aircraft diversion to an alternate airport.
- . During eight percent of the aircraft ingestion events, an in-flight shutdown of an engine occurred. During two percent, there was an involuntary in-flight engine shutdown.

. The probability that a crew action is required increases with the severity of engine damage.

Engine Damage

- . Fifty percent of all engine bird ingestions result in some engine damage. Forty-eight percent for turbofans and fifty-seven percent for turboprops.
- . There does not appear to be any correlation among different types of engine damage.
- . The probability of damage increases with the weight of the bird that is ingested.
- . The probability of engine damage, given a bird ingestion has occurred, is greater when the ingestion occurs during the takeoff and climb phases of flight than those that occur during approach and landing.
- . The probability of engine damage, given a bird ingestion has occurred, is greater when the aircraft airspeed is greater than or equal to 140 knots than those that occur at less than 140 knots.
- . Five percent of all engine bird ingestions result in an engine failure.
- . Two-thirds of the engine failures, where the bird weight was positively identified, involved bird weights greater than four pounds. In comparison one-third were at weights less than one-half pound.
- . Engine failure appears more likely to occur when multiple birds are ingested.
- . The mean or average weight (65.3 oz.) of the birds that caused engine failures was heavier than the mean (16.8 oz.) for all bird ingestion events.
- . Engine failure is not necessarily associated exclusively with severe engine damage.
- . A disproportionate number of engine failures occurred on the right side of the aircraft.

Probabilities of Ingestion

. Bird ingestions are more likely during the takeoff and landing phases of aircraft operation.

Data Quality

There are some statistically significant differences between the data collected in year 1 and in year 2.

SECTION 9 REFERENCES

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SECTION 10 GLOSSARY OF TERMS

Term

Definition of Term

Ingested Bird

A bird having experienced the process of bird ingestion.

Aircraft Operation

A nonstop aircraft flight from one airport to another (includes taxiout from departure airport through taxion at arrival airport).

Engine Operation

The participation of each engine of an aircraft in an aircraft operation (e.g., a twin engine aircraft would, ideally, experience two engine operations for each aircraft operation).

Engine Ingestion Event

The simultaneous passage of one or more birds through the inlet of an engine during an engine operation.

Aircraft Ingestion Event

The simultaneous passage of one or more birds through the inlet of one or more engines of an aircraft during an aircraft operation.

Engine Hours

The total running time, measured in hours, of an engine or group of engines during a given period.

Ingestion Rate

Rate at which (aircraft or engine) events occur per flight event. Flight event refers to aircraft or airport operation. The components of ingestion rate are specified whenever this term is used. The influence of engine inlet opening size is not taken into account in this definition.

Normalized Ingestion Rate

Ingestion rate normalized to a given inlet size. Normalization allows statistical comparison of ingestion rates of engines with different inlet opening sizes.

APPENDIX A

ENGINE APPLICATIONS

<u>Engine</u>	Engine Type	Engine <u>Manufacturer</u>	Engine Face Area (in ²)	Typical Throat Area (in ²)	Typical Aircraft <u>Installation</u>
JT15D	Turbofan	Pratt & Whitney	346	310	Cessna Citation 1 & S2, Mitsubishi/Beech Diamond, Beechjet
ALF 502	Turbofan	Textron-Lycoming	1276	984	Canadair Challenger CL- 600, British Aerospace 146
TFE 731	Turbofan	Garrett	625	450	British Aerospace 125- 700 & 125-800; Dassault-Breguet Falcon 10, 100, 50, and 900; Gates Learjet 35A, 55, 55ER, and 55LR; Israel Aircraft Industries Westwind 1124 and Astra 1125; Lockheed Jetstar II; Rockwell/Sabreliner 65; Cessna Citation III
TPE 331	Turboprop	Garrett	72	73	Alaska F & W, Goose; British Aerospace, Jetstream 3, 31, 32; Carstedt, Jetliner 600; CASA 212; Cessna, Conquest 2; Commander 680, 690, 695, Turbocommander; Dornier 228; Fairchild Metro, Metro 2, 3, Merlin 2,3,4, Peacemaker, Porter; Grumann, S2 Tracker; Helitec s 55; IAI, S2 Tracker; Mitsubishi, Marquise, Soltaire, MU-2; Pilatus, Porter; Piper, Cheyenne 400; Short Brothers, Skyvan; Turbobeaver; Volpar, Turbo 18

APPENDIX B CONTENTS OF FAA BIRD INGESTION DATA BASE SMALL ENGINES MAY 1987 - APRIL 1989

This appendix presents the contents of the small engine bird ingestion data base maintained by the FAA. The appendix presents actual data extracted from the FAA data base and used in this report. The data base contents are described below:

COLUMN	DESCRIPTION OF COLUMN CONTENTS
EDATE	Date(mm/dd/yyyy) of ingestion event.
EVT#	FAA ingestion event sequence number reflecting order in which events were entered into the FAA bird ingestion data base.
ETIME	Local time of bird ingestion.
SIGN_EVT	Significant event factors. AIRWRTHY - engine related airworthiness effects INV POS LOSS - involuntary power loss MULT BIRDS - multiple birds in 1 engine MULT ENG - multiple engine ingestion (1 bird in each engine) MULT ENG-BIRDS - multiple engine ingestion and 1 or both engines sustained multiple bird ingestion TRVS FRAC - transverse fan blade fracture OTHER - other significant factor, may be reported in narrative remarks NONE - no significant factor noted
AIRCRAFT	Aircraft type.
ENGINE	Engine model. (ALF502;JT15D;TFE731) - turbofan engines (TPE331) - turboprop engine
DASH	Engine dash number
ENG_POS	Engine position of engine ingesting bird. Since each engine ingestion event has a unique record in the data base, duplicate event numbers indicate multiple engine ingestion events. This column provides record uniqueness in such cases.
DMG_CODE	Letter codes summarizing engine damage resulting from the bird ingestion. This column does not exist in the actual FAA data base, but was developed by the contractor to compress 17 YES/NO damage fields into a single column. A letter code appears for damage columns whose values are YES. Each page of damage information contains a legend identifying the damage type. In the explanation of damage

codes below, a number in parentheses indicates the damage severity code which is further explained in the SEVERITY column. The data base column name is given in the explanation of the damage code.

- A(4) ENG DAM; engine damaged due to bird ingestion
- B(3) LEAD EDG; leading edge distortion/curl, minor fan blades
- C(3) BEN/DEN; 1 to 3 fan blades bent or dented
- D(2) BE/DE 3; more than 3 fan blades bent or dented
- E(3) TORN 3; 1 to 3 fan blades torn
- F(2) TORN 3; more than 3 fan blades torn
- G(2) BROKEN; broken fan blade(s). leading edge and/or tip pieces missing; other blades also dented
- H(3) SHINGLED; shingled (twisted) fan blades
- I(1) TRVSFRAC; transverse fracture a fan blade broken chordwise (across) and the piece liberated (includes secondary hard object damage)
- J(2) SPINNER; dented, broken, or cracked spinner (includes spinner cap)
- K(1) CORE; bent/broken compressor blades/vanes, blade/vane clash, blocked/disrupted airflow in low, intermediate, and high pressure compressors
- L(3) NACELLE; dents and/or punctures to the engine enclosure (includes cowl)
- M(1) FLANGE; flange separations
- N(2) RELEASED; released (walked) fan blades (blade retention mechanism broken)
- 0(1) TURBINE; turbine damage
- P OTHER; any damage not previously listed
- Q UNKNOWN

NOTE: The maximum number of damage codes listed for an engine ingestion event is three. These three damage codes reflect the most severe damage that occurred. There may be other damage that occurred which is less severe that may be listed in the remarks column.

SEVERITY Numeric code indicating the severity of engine damage resulting from the bird ingestion. This column does not exist in the actual FAA data base, but was developed by the contractor as a result of an analysis of reported damage in the data base. The lower the severity code, the more severe the damage. The severity rating assigned to a flight is determined as the lowest severity rating attained by any of the damage categories. The corresponding severity ratings for each damage category were given in parentheses in the DMG_CODE discussion above. Turbofan engine damage severity codes:

- 1 most severe damage (damage is known)
- 2 moderately severe damage (damage is known)
- 3 least severe damage (damage is known)
- 4 damage indicated, but not specified
- 9 no damage reported

Turboprop engine damage severity codes:

1 - extremely severe damage (might jeopardize the airworthiness of the aircraft) 3 - minor damage

4 - damage indicated, but not specified

9 - no damage reported

POW LOSS Degree of power loss as a result of bird ingestion

NONE - no power loss

EPR DEC - engine pressure ratio decrease

SPOOL DOWN - engine spooled down

N1 CHANGE - N1 rotor change

N2 CHANGE - N2 rotor change

COMPRESSOR - compressor surge/stall

UNKNOWN - unknown whether power loss occurred

MAX VIBE Maximum vibration reported as a dimensionless unit.

THROTTLE Voluntary throttle change by crew in response to bird ingestion.

ADVANCE - voluntary throttle advance

RETARD - voluntary throttle retard

IDLE - voluntary throttle retard to idle

CUTOFF voluntary throttle retard to cutoff

NONE - no voluntary throttle change

IFSD Indicate whether a voluntary in-flight shutdown occurred in response to

bird ingestion.

NO - no shutdown

VIBES - shutdown due to vibrations

STAL/SURG - shutdown due to compressor stall/surge

HI EGT - shutdown due to high exhaust gas temperature

EPR - shutdown due to incorrect engine pressure ratio

INVLNTRY - involuntary engine shutdown

PARAMTRS - shutdown due to incorrect engine parameters

VLNTRY - voluntary engine shutdown

OTHER - other reasons, may be listed in remarks

UNKNOWN - unknown cause for shutdown

POF Phase of flight during which bird ingestion occurred.

(TAXI; TAKEOFF; CLIMB; CRUISE; APPROACH; LANDING; UNKNOWN)

ALTITUDE Altitude (ft. AGL) at time of bird ingestion.

SPEED Air speed (knots) at time of bird ingestion.

FL RULES Flight rules in effect at time of bird ingestion.

IFR - instrument flight rules

VFR - visual flight rules

UNK - unknown

LT COND Light conditions at time of bird ingestion.

(DARK; LIGHT; DAWN; DUSK; etc.)

WEATHER Weather conditions at time of bird ingestion.

CREW_AC Crew action taken in response to bird ingestion.

ATO - aborted takeoff

ATB - air turnback

DIV - diversion

UNK - unknown

NONE - no crew action taken

N/A - not applicable

OTHER - some action taken, may be specified in narrative remarks

CREW_AL Indicates whether crew alerted to presence of birds at time of bird ingestion.

(YES; NO; UNKNOWN)

BIRD_SEE Indicates whether ingested bird(s) seen prior to ingestion

NO - not seen

YES - seen

SEVERAL - 2 to 10 birds observed

FLOCK - more than 10 birds observed

BIRD_NAM Common bird name. Trailing asterisk (*) implies bird not

positively identified as such.

BIRD_SPE Species of positively identified bird. Alphanumeric

identification code which conforms to Edward's convention.

#_BIRDS Number of birds ingested. An asterisk (*) implies more than one

bird but the exact count is unknown.

WT OZ l Weight (oz.) of first ingested bird.

US INCID Indicates whether bird ingestion occurred within US boundaries.

(YES;NO)

CTY_PRS Scheduled city pairs of aircraft operation.

(from code: to code) 3 letter city airport code.

AIRPORT Airport at which bird ingestion event occurred.

3 letter city airport code.

LOCALE Nearest town, state, country, etc.

REMARKS Narrative description providing additional information concerning

some aspect of the ingestion.

[†] Edwards, E.P., "A Coded List of Birds of the World," IBSN:911882-04-9, 1974.

EDATE	EVT&	ETIME	SIGN_	EUT	AIRCRAFT	ENGINE	DASH	EI	ŧĞ_P0S	DMG_CODE	Severi tv	POH_L
05/03/87		2 18:00:00	нии г	prons	FALCON 50	TFE731	3	1	LEFT	H,K	1	NONE
05/11/87		3 18:45:00		OT L'DT.	BRE 125	TFE731	5		LEFT	-	_	NONE
05/14/87		1 16:30:00			BRE 146	ALF502	P.5	4	RIGHT OUTBOARD		ė,	HOHE
05/14/87		5 15:30:00			HETRO	TPE331	110	1	LEFT	H.K	3	YES
05/17/87		4 16:00:00		DIROS	SABRE 65	TFE731	3R	1	LEFT	A,C,P	3	NONE
05/20/87		7 9:30:00			CON 441	TPE331	€	1	LEFT		9	NONE
05/22/87		8 5:30:00			HETRO II	TPE331	BUH		LEFT		9	NONE
05/25/87		5	HULT	BIRDS	FALCON 10	TFE731	2	2	RIGHT	A.D	2	NONE
05/25/87	5	2 15:30:00	NONE		LEAR 35A	TFE731	2	_		A	4	
05/26/87		6	NONE		LEAR 35	TFE?31	2		RIGHT	A,D		HONE
05/31/87	' 1	4	HULT	ENG	LEAR 55	TFE731	3R		LEFT	A,K		HOHE
05/31/87	' 1	.4	HULT	ENG	LEAR 55	TFE731	38	-	RIGHT	A.K		HONE
06/17/87	1	9 14:00:00	NONE		JETSTAR	TFE731	3		LEFT OUTBOARD	H,E,E		NONE
06/17/87		.0	HONE		BRE146	RLF502	R5		RIGHT INBOARD			HONE
06/21/87	' 2	0 21:30:00			MU~2	TPE331	5	2		0.0	ب	
07/01/87	_	13	NONE		FALCON 50	TFE731	3		RIGHT	A,K	1	
07/13/87		6 20:45:00			BAE 125-700	TFE731	38.		RIGHT	A,D	2	
07/14/87		7 16:00:00			FRLCON 50	TFE731	3 11U		CENTER LEFT	ب ن		HONE
07/21/87		1 14:00:00			HETRO III	TPE331	110		RIGHT	A.X		HONE
07/22/87		22 11:30:00			HETRO III	TPE331 TFE731	5,10		LEFT	A.C		I YES I NONE
07/27/87		18	HONE		LEAR 35	TPE331	โป		LEFT	A.K		YES
07/28/87		3 17:30:00			HETRO III	ALF502	R5		LEFT INBOARD			NONE
07/30/87		1 20:00:00			BRE 146	ALF502	ĽZ		LEFT	A.D.H.L		NONE
07/31/87		12 8:40:00 19 9:14:30		DIDDC	CL600	TFE731	5		LEFT	,.,.,.,		NONE
07/31/87		19 9:14:30 26 11:00:00			LEAR 35 Casa 212	TPE331	5		LEFT	A.K		HONE
08/11/87				614.05	LEAR 35A	TFE731	ЗA		LEFT	A.C.P	3	
08/16/87			_		JS 31	TPE331	1006		RIGHT	A.K		YES
08/24/87		38 11:00:00 13	NONE		BAE 146	ALF502	R5		LEFT OUTBOARD		ğ	
- 08/26/87 - 09/09/87		.5 94 &:50:00			LEAR 55	TFE731	SAR		RIGHT	A.E		HONE
09/10/87		35 14:30:00		erens	LEAR 35	TFE731	2B		LEFT			NONE
09/10/87				ENG-BIRDS	CITATION	TFE731	3		LEFT	A.D		NONE
03/10/87				ENG-BIRDS	CITATION	TFE731	3		RIGHT			HONE
09/12/87		27 15:00:00			BRE146	ALFS02	R5		RIGHT INBOARD	A,C,H		NONE
03/12/87		27 15:00:00			BRE 146	ALF502	R5		LEFT INBOARD	, , , , , , ,		NONE
09/14/87		28 8:5::00			BRE 146	ALF502	R5		LEFT OUTBOARD			NONE
09/16/87		9 12:00:00			JS 3101	TPE331	10UF		RIGHT			NONE
09/18/87		40 9: J: 0			JS 3101	TPE331	10UF	1	LEFT			HONE
09/20/87		36 12:00:00		BIRDS	BRE 125-700	TFE731	ЗR	1	LEFT	A.D.H,K		L NONE
09/22/87		41	NONE		METRO	TPE331	16	1	LEFT	H,K	3	YES
09/22/87		94	HONE		HETRO 4	TPE331	110	2	RIGHT	A,K,P	3	3
09/28/87		42 13:00:01	NONE		JS 3101	TPE331	1006	2	RIGHT	A.F	3	SYES
10/01/87		45 9:42:00	O NONE		JS 31	TPE331	1006	2	RIGHT		•	3
10/05/87		29 13:30:00	O NONE		BRE 146	ALF502	R5		RIGHT INBOARD	A.E,C	7	3 NONE
10/08/87		30 19:45:01	O NOME		BRE 146	ALF502	P.5		LEFT INBOARD		<	HONE
10/13/87		43 9:0 1:0	O NONE		LEAR 35	TFE731	2		RIGHT	H.D,K,P		1 NONE
10/13/81	_	46 22:00:01	O NONE		BAE 3101	TPE331	1006		RIGHT			3 NONE
10/27/80	7	47 20:00:0			JS 3101	TPE331	10		RIGHT	A,K		3 FLAM
10.13078	7	55 12:.0:0	(NONE		METRO 3	TPE331	1 1U		LEFT			HONE
11/02/8	7	50 Ø: 5:D	(MONE		CITATION3	TFE731	38		: KIGHT			3 NONE
11/04/8				ENG-BIRDS	-	ALF502	R5		RIGHT INBOARD			9 NONE
11/04/8	7			ENG-BIRDS	BAE146	ALF502	R5		RIGHT DUTBOARD			A NONE
11/06/0		51 7:31:0			METRO	TPE331	1 10		LEFT	H.K		3 NO
11/11/8		56 12: 3:0			JS 31	TPE331	LOUF		LEFT	A.K.P		3 NONE
11/19/8		53 9:15:0			BRE 125-800		5R		RIGHT	H.D.P		2 NONE
11/23/8		57 19:30:0			HEAGO III	TPE331	110		RIGHT			9 NONE
11/29/8		32 17:00:0			BRE 146	ALF502	R5		LEFT OUTBOARD			9 NONE
12/03/8		54	NONE		FRLCON 10	TFE731	2		RIGHT	n +		9 NONE
12/05/8	_	64 19:00:0			TCOHM 695B		10R		RIGHT	R,K		3 NONE
12/10/8		18	NONE		BAE 146	ALF502	R5		RIGHI INBOARD	я,с,н		3 NONE
12/11/8		49	NOME		BRE146	ALF502	R5		L LEFT OUTBOARD	6 V		9 NONE
12/11/8	7	70 18:30:0	U MULI	RTKN?	JS 31	TPE331	10UF		RIGHT	A,K		3 YES

ONG_CODE	Severi TY	POH_LOSS	MAX_VIBE	THROTTLE	IFSD	POF	ALTITUDE	SPEED	FL_RULES	LT_CONDS	HEATHER
-						LOUDZAN	0	122	VFR	DUSK	SCAFTERED
Ĥ.K	1	NONE		NONE	NO	LANDING		125	VFR	LIGHT	CLEAF:
	ç	NONE		HONE	NO	LANDING	25	125	W.F.K.	LIGHT	CLEAR:
	ç	NONE		HONE	NO	TRIEDFF	10	00	VFR	LIGHT	CLEAR
H.K	5	: YES	YES		NO	LANCING	10	9 CI			
H.C.P	3	NONE	NONE	NONE	NO	TAKEOFF	25		VFR	LIGHT	CLERF:
-		NONE	NONE	HONE	NO	TAKEOFF		125	VFR	LIGHT	CLEAR
		NONE	NONE	NONE	NO	TAKEOFF	150		VFR	LIGHT	CLEAR
9.0		NONE		HONE	НO	LANDING		100	VFR	LIGHT	CLEAR.
А		9	NONE		HO	APPROACH	200	160	VFR	LIGHT	CLEAR
A.D		NONE	NONE		NO	TAKEOFF			VFR	LIGHT	CLEAR
A.K		1 NONE	NONE	HONE	но	TAKEOFF	150		IFR	LIGHT	CLERR
A.K		1 NONE	NONE	HONE	NO	TAKEOFF	150		IFR	LIGHT	CLEAR
H.E.E		NONE	NONE	NONE	H0	TAKEOFF	200	100	VFR	LIGHT	CLEAR
		9 NONE		HONE	NO	UNKNOHN					
		9 NONE	NONE	NONE	NO	TRKEOFF	1500	150	VFR	MRK	CLEAR
H,K		1	NONE		NO	UNKNOHN					
H.D		2 NONE	YES	NONE	NO	LANDING		117	VFR	DUSK	CLEAR
11.60		4 NONE	NONE	NONE	NO	APPROPICH	6000	140	VFR	LIGHT	SCATTERED
н.х		3 NONE	NONE	RETARD	NO	TRKEOFF	£	}	VFR	LIGHT	SCAFTERED
1140		4 YES	NONE		NO	TAKEUFF	C	100	VFR	LIGHT	CLEAR
A.C		3 NONE	NONE	NONE	NO	UNKNOHN			IFR	LIGHT	CLERR:
H.K		3 YES	NONE	-	NO	TAKEOFF		100	VFR	DRHM	CLEAR
n.K		9 NONE		NONE	NO	TAXI	() D	VFR	DUSK	CLEAR
் சாயர		2 NONE	YES	NONE	NO	TAKEOFF	35	140	VFR	LIGHT	CLEAR
H,0,H,L		9 NONE	NONE	NONE	NO	TAKEOFF	1	120	VFR	LIGHT	CLEAR
0 P		3 NONE	NONE	NONE	NO	CLIHB	800	110	VFR	LIGHT	RAIN
A.K		3	HIGH			TAKEOFF		-V1	VFR	LIGHT	CLERR
H.C.P		3 YES	112-011		YES	CRUISE	450	180	VFR	LIGHT	CLERR
й .К		9		NONE	NO	UNKNOHN					
o r		3 NONE	NONE	NONE	NO	TAKEOFF	1	1 10	VFR	LIGHT	OVERCAST
H.E		9 NONE	NONE	HONE	NO	TAKEOFF		128	VFR	LIGHT	CLEAR
0.0		2 NONE	NONE	HONE	NO	LANDING	24	150	VFR	LIGHT	CLEAR
A.D		9 NONE	NONE	NONE	NO	LANDING	24	(i 1 50	VFR	LIGHT	CLERR
0.6.11		3 NONE	,,01.2	NONE	NO	TAKEOFF	10	0 120	VFR	LIGHT	CLERF:
A,C,H		9 NONE		NONE	NO	TAKEOFF	10	0 120	VFR	LIGHT	CLERR
			NONE	SHUT OFF	HO	LANDING		O 85	VFR	LIGHT	CLEAR
		9 NONE	NONE	NONE	NO	LANCIING			VFR	LIGHT	SCAFTERED
		9 NONE	NONE	NONE	HO	APPROACH	10	0 125	VFR	LIGHT	OVERCAST
6 5 11 0		1 NONE	HINOR	NONE	NO	TAKEOFF		0 125	VFR	LIGHT	OVERCAST
A.D.H.K		3 YES	11211014	ADVANCE	HO	CLIHB				DARK	
H,K			NONE	NONE	NO	CLIMB			VFR	DARK	CLEAR
H,K,P		3 3 YES	NONE	HONE	NO	TAKEOFF	32	0 120	1FR	LIGHT	CLERR
H,F			HOLL	HONE	NO	REPRORCH	20	0 120	UFR	LIGHT	CLERR:
0		9 3 NONE		NONE	NO	UNKNOHN				LIGHT	SCAFTERED
ค.ย,C			NONE	HONE	NO	LANDING	20	90		DARK	SCATTERED
i e e e e		S NONE	NONE	NONE	NO	TAKEOFF		0 130	UFR	LIGHT	OVERCHST
H.D,K,P	-	9 NONE	NONE	NONE	NO	AFPROACH		120	IFR	DARK	CLEAR
0. P		3 FLAME		CUTOFF		INTRAPPEOACH	200	00 150	VFR	DARK	CLERR
H.K		9 NONE	NONE	NONE	NO	LANDING		0 80	VFR	LIGHT	SCATTERED
		9 NONE	HONE	NONE	NO	CF:UI SE	250	00 2 5 J	ŲFR	LIGHT	CLERR
		9 NONE	NONE	NONE	NO	LANDING			VFR	LIGHT	SCATTERED
) 			NONE	NONE	NO	LANDING			VFR	LIGHT	SCAFTERED
5)		9 NONE	HIGH	CUTOFF	VIBES	TAKEOFF	9	00 110	VFR	LIGHT	CLERK
H.K		3 NO	NONE	NONE	NO	LANDING		(1 90		LIGHT	CIVERCHST
A.K.P		3 NONE	NONE	NONE	NO	THKEOFF		0 120	VHC	LIGHT	OVERCAST
H.D.P		2 NONE	NONE	NONE	NO	APPROACH		30	IFR	DUSK	CLEAR
		9 NONE	NONE	NONE	NO	UNKNOWN		-		DUSK	
,		9 NONE	HUITE	NONE	NO	APPROACH	-1 ù	00 190	VFR	DARK	CLEAR!
A		9 NONE	NONE	NONE	NO	APPROACH		50 130	VFR	DUSK	CLEAR
A,K		3 NONE	NONE	*******	NO	UNKNOHN	_				
A,C,H		3 NONE 9 NONE	NONE		NO	UNKNOM!					
, v n		3 YES	SOME	NONE	NO	TAKEOFF		0 80	IFR	DARK	OVERCAST
A,K		3 FE3	JU116.								

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OF OF OF OF OF OF OF OF	EDATE	FUTO	CREH_AC	CREH_AL	EIRD_SEE	BIRD_NAM	BIRD_SPE #	EI ROS	HT_02	_1	US_INCID	CTY_PRS	AIRPORT
10 10 10 10 10 10 10 10	E DHI C	5010			CEUEDOI	CEOCH 1 X		3	i 2·	1.0	YES		
1	05/03/87	7					14812	1	. 1	0_5	YES		BKL
1	05/11/87	7						1			NO .		LEEDS
100-11-14-12-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1-	05/14/87	7				SPHKNUHA		1					PSC
1001-1979 MANNE MONE M			NONE	NO		MOCHENIC DOME	20105						HSY
MOVE NO			NONE		-		21 100						
100 20 20 20 20 20 20 20			NONE	NO			1.4643						
MONE NO			ATB	NO			14470						
1			HONE	NO	SEVERAL					4 N			
				ND	ONE								
105/31/87 NONE NO SEVERNL SULEN 1 16,0 NO NO NO NO NO NO NO NO			NONE	NO		PIGEON*							202
100-17-087 NOME				ND	SEVERAL	GULL¥							
					SEVERAL								STE
					YES	HEF:RING GULL	14814			U.U			
					NO								DOD
					NO	DOVE×			1				KUD
07/13/97					NO								MEU
11						YELLOH CROWN NIGHT HERON	1127						
1													
1													
NOTE													
07-28-97 ATO NO NO ROCK DOVE 2P1 1 14-LU YES LRA (07-29-97) NO NO NO TREE SPARROH 70223 1 1 1.0 NO LARZHO (07-29-97) NO NO NO TREE SPARROH 70223 1 1 1.0 NO LARZHO (07-29-97) NO NO NO TREE SPARROH 70223 1 1 1.0 NO LARZHO (07-29-97) NO NO NO SEVERAL REGISTRY (07-29-97) NO NO NO SEVERAL REGISTRY (07-29-97) NO													
07-30-07 NONE NO NO TREE SPARRON 70223 1 1.0 NO EMPLY NO CONTRACT SPARRON 7023 1 20.0 NO BRYAN NO CONTRACT SPARRON 7023 1 120.0 NO BRYAN NO CONTRACT SPARRON 7023 1 120.0 NO BRYAN NO CONTRACT SPARRON 7023 1 120.0 NO BRYAN NO CONTRACT SPARRON 7024 1 120.0 NO BRYAN NO CONTRACT SPARRON 7024 1 120.0 NO BRYAN NO CONTRACT SPARRON 7024 1 120.0 NO CONTRACT SPARRON 7024													
07-31-087 NONE NONE NONE SEVERAL SEAGULEX 2010 NONE SEVERAL SEAGULEX 14N12 1 16.0 NO 160-16-087 NONE NO NO NO NE COMMON HAITE SEAGULEX 1 14N12 1 16.0 NO 160-16-087 NONE NO NO NE COMMON HAITE SEAGULEX 1 150-0 NO							70223						
10 10 10 10 10 10 10 10							3K 31						
19/11/87 ATB NO SEVERAL SEAGULE 1 16.0 NO NO NO NO NO NO NO NO							2P1						IUH
No. 17.5 No.													
1				110			14N12						
1				VES	ONE				1	32.0			
NOTE NOTE NOTE NOTE SERFITER FELLOHLEGS SH19 1 6.5 YES FRED STARL NOTE STA									1				
109/10/87 NONE NO YES SPARROIN						GREATER YELLOHLEGS	6N19		1				FLD
10/10/87 NONE NO													
1									2	4.0) YES		
1									1	4.0	YES		
19/12/87 NOTE NO PLOCK HOURNING DOVE 2P105 1 4.0 YES CHH-18B CHH							2P105		1	4.0	YES	CHH-I AD	CHH
1							2P105		1	4.0	YES	CHH-I AB	
1		_				HOURIZIO ESTE			1		NO	FOL-HOR	
MONE NO ONE HOURNING DOVE 2P105 1 4.0 YES MYZUZIR7 ATB YES YES CANADA GOOSE 2J30 2 128.0 YES MYZUZIR7 ATB NO NO SEAGULL* YES MYZUZIR7 NONE NO NO NO NO MONE NO NO NO MONE NO NO NO NO NO MONE NO NO NO NO NO MONE NO NO NO NO NO NO MONE NO NO NO NO NO MONE NO NO NO NO NO NO MONE NO NO NO NO NO NO MONE NO NO NO NO NO NO NO									1				ATL
19/20/067 ATB YES YES CANADA GOOSE 2300 2 126,0 YES 126,0 YE						HOURNING DOVE	2P 105		1	4.0) YES		
1 YES									2 1	28.1	O YES		
09/22/87 NONE NO NO SEAGULL* 09/28/87 NONE NO NO NO 09/28/87 NONE NO NO 1 1 6.0 YES 10/01/87 NONE NO ONE 10/05/87 NONE 10/05/87 NONE 10/06/87 NONE 10/08/87 NONE 10/08/87 NONE 10/01/87 NONE 1					165	CHINEH OBOSE			1		YES		
1 15.0 YES 15.0					NO	SEAGIN I &					YES		
10/01/87 NONE NO ONE NO ONE NO ONE NO ONE NO ONE NO ONE COMMON LAPHING SHI 1 YES YEM-PSC 10/05/87 NONE YES ONE COMMON LAPHING SHI 1 7.7 NO PMK-PMK PMK 10/13/87 ATB NO NO SEAGULL* 1 16.0 NO YES 10/23/87 NONE NO NO ONL NO						-c100CE~			1	16.	O YES		
10.705/87 NONE NO NO SEAGULLX 1 1 7.7 NO PHK-PHK PHK 10/13/87 NONE NO NO SEAGULLX 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1									1		YES		HEH
10/05/87 NOME YES ONE COMMON LAPHING SN1 1 7.7 NO PHK-PHK PHK 10/13/87 ATB NO NO SEAGULL* 1 16.0 NO CVT 10/13/87 NOME NO NO OHL* 1 16.0 NO CVT 10/13/87 NOME NO NO OHL* 1 YES 10/13/87 NOME NO NO OHL* 1 YES 10/13/87 NOME NO NO OHL* 1 YES FRG 11/02/87 NOME NO NO NO NO OHL* 1 YES LAX-CCR CCR 11/04/87 NOME YES SEVERAL REDHINGED BLACKBIRD 64/254 \$ 2.0 YES LAX-CCR CCR 11/04/87 NOME YES SEVERAL REDHINGED BLACKBIRD 64/254 \$ 2.0 YES LAX-CCR CCR 11/04/87 NOME YES SEVERAL REDHINGED BLACKBIRD 64/254 \$ 2.0 YES LAX-CCR CCR 11/06/67 ATB NO ONE SEAGULL* 14N36 1 36.0 NO SBA 11/11/87 NOME NO ONE BLACK-HERDED GULL 14N36 1 10.0 NO EDVE 11/23/87 NOME NO FLOCK SEAGULL # 1 8.0 NO BSL 11/29/87 NOME NO SEVERAL FRANKLIN*S GULL 14N31 1 9.0 YES SMA-SJC MKC 11/05/417 NOME NO NO NO NO HAPPARE				NU							YES	YKH-PS0	
10/13/87 NONE NO NO SERGULL* 10/13/87 NONE NO NO NO SERGULL* 10/13/87 NONE NO NO NO OHL* 10/27/87 IFSD NO NO NO OHL* 10/30/87 NONE NO FLOCK SEAGULL* 11/02/87 NONE NO NO NO NO NO NO NO NO NO ONE SEAGULL* 11/04/87 NONE YES SEVERAL REDHINGED BLACKBIRD 64254 * 2.0 YES LAX-CCR CCR 11/04/87 NONE YES SEVERAL REDHINGED BLACKBIRD 64254 * 2.0 YES LAX-CCR CCR 11/06/87 ATB NO ONE SEAGULL* 11/11/87 NONE NO ONE SEAGULL* 11/11/87 NONE NO ONE BLACK-HERDED GULL 14N36 1 10.0 NO EDVE 11/23/87 NONE NO FLOCK SEAGULL * 14N36 1 10.0 NO BSL 11/29/87 NONE NO FLOCK SEAGULL * 14N36 1 9.0 YES SNA-SJC NO NO HIPP NONE NO				ucc		COMMON I REUTING	5N1			7.	7 NO	PHK-PHK	PWK
10/13/87 NONE NO NO NO NO NO NO										16.	O NO		CVT
10/13/87 NONE NO NO OHL* 10/27/87 IFSD NO NO FLOCK SEAGULL* 11/02/87 NONE NO		_				JE1100EE-			-		YES		
10/2//87 1FSU NU						∩ui ×			1		YES		
11/02/87 NONE NO									1		NO		
11/04/87 NONE YES SEVERAL REDHINGED BLACKBIRD 64254		_				BEHOULLE					YES		FRG
17/04/87 NONE YES SEVERAL REDHINGED BLACKBIRD 64254 # 2.0 YES LAM-CCR CCR 11/06/87 ATB NO ONE SEAGULL # 1 32.0 YES SBA 11/11/87 NONE NO ONE MALLARD 2J84 1 36.0 NO 11/19/87 NONE YES ONE BLACK-HERDED GULL 14N36 1 10.0 NO EDVE 11/23/87 NONE NO FLOCK SEAGULL # 1 8.0 NO BSL 11/29/87 NONE NO SEVERAL FRANKLIN*S GULL 14N31 1 9.0 YES SNA-SJC 12/103/87 NONE NO NO NO 1 NO HIP 12/10/87 NONE NO NO NO NO NO NO						PENUTMEEN DI OCKRITON	64254			2.		LAX-CCI	R CCR
11/06/87 ATB NO ONE SEAGULL* 11/06/87 ATB NO ONE MALLARD 2J84 1 36.0 NO 11/11/87 NONE NO ONE BLACK-HERDED GULL 14N36 1 10.0 NO EDVE 11/23/87 NONE NO FLOCK SEAGULL # 1 8.0 NO BSL 11/29/87 NONE NO SEVERAL FRANKLIN*S GULL 14N31 1 9.0 YES SNA-SJC 12/03/87 NONE NO NO NO NO HLP 12/10/87 NONE NO												LAX-CCI	R CCR
11/1087 NONE NO ONE MALLARD 2384 1 36.0 NO 11/1987 NONE VES ONE BLACK-HERDED GULL 14N36 1 10.0 NO EDVE 11/23/87 NONE NO FLOCK SERGULL # 1 8.0 NO BSL 11/29/87 NONE NO SEVERAL FRANKLIN*S GULL 14N31 1 9.0 YES SHA-SJC 12/103/87 NONE NO NO NO HLP 12/10/87 NONE NO													
11/19/87 NONE NO ONE BLACK-HEADED GULL 14N36 1 10.0 NO EDVE 11/23/87 NONE NO FLOCK SEAGULL # 14N36 1 10.0 NO BSL 11/29/87 NONE NO SEVERAL FRANKLIN*S GULL 14N31 1 9.0 YES SNA-SJC 12/03/87 NONE NO NO NO NO 1 NO HLP 12/10/87 NONE NO NO NO NO 1 NO HARARE 12/11/87 NONE NO		_					2.184		_				
11/23/87 NONE NO FLOCK SERGULL # 1 8.0 NO BSL 11/29/87 NONE NO SEVERAL FRANKLIN*S GULL 14N31 1 9.0 YES SHA-SJC 12/03/87 NONE NO NO NO 1 NO HLP 12/10/87 NONE NO NO NO 1 NO HRRARE 12/11/87 NONE NO NO NO 1 NO HRRARE		_							1				EDVE
11/29/87 NONE NO FECEN SENDELL W 11/29/87 NONE 11/29/87 NONE NO SEVERAL FRANKLIN'S GULL 14N31 1 9_0 YES MKC 11/40/8/17 NONE NO NO 1 NO 1 NO 1 NO 1 NO 1 NO 1 N							2		ī	8.	ON 13.		BSL
17/29/07 NONE NO SEVERAL FRANKLIN'S GULL 14M31 1 9.0 YES MKC 17/03/07 NONE NO NO 1 NO HLP 17/10/087 NONE NO NO NO 1 NO HARARE 17/11/087 NONE NO NO NO 1 NO HARARE				MU	LEUCK	SCHOOLL #			-			SNA-SJ	
12/13/87 NONE NO SEVERAL PRANKLIN'S BULL 1 NO HLF 12/15/H7 NONE NO NO 1 NO 12/10/87 NONE NO NO 1 NO HARARE 12/11/87 NONE NO NO NO 1 NO HARARE				NO	CENEDO	EDONNI TNYC CIR.	14N31			9.			
12/10/87 NONE NO NO 1 NO 12/10/87 NONE NO NO 1 NO HARARE 12/11/87 NONE NO NO 50 1 NO 5		_				- LYUNYTH > OATT	1 1101		ī				
12/10/87 NONE NO NO HARARE 12/11/87 NONE NO NO COMMON LOCATION SNI # 7-7 NO		_							ī				
12/11/87 NUME NO NO COMMON LOCATION SHI \$ 1-7 NO									-				HARARE
35/31/85 MIN UN NO COUNTRY CHEMY AND CHEMY						COMMON LOCALING	SNI			7			
	12/1	1/87	MIU	INU	.,.	COUNTRY CHEMING	J 2						

ALE	US_INCID	CTY_PRS	ALRPORT	LOCALE	REMARKS
HLE	_				(APINITAL)
	Y'ES		DTH	DETROIT, MICHIGAN	COMP STATORS BENT
RQ1	YES		BKL	CLEVELAND, OH	COMP DIFFICIES DENI
VE.			LEEDS	LEEDS, ENGLAND	
05	YES				
CO	163		PSC	PRSCO, NA	1+25TG IMP DAM, 1BENT BLADE
0	YES		HSY	NEH ORLEANS, LA	COMP STATORS DAMAGED
NS.	A.F.P.		ENA	EVANSVILLE, INCIANA	
KO!	NO		LDK:	LINKOPING, SHEDEN	
	NB		LIN	MILANO, ITALY	5 F BLDS HERE BENT
AN			F'LCH	LONDON, UK	C. I BCD2. HENCE BCITT
DO	NO		SCL	SANTIAGO, CHILE	
TI	NO		J U _	THESSALONIKI, GREECE	A ACTO LOC DED DENT TO THE CITY
55	NO				1 1STG LPC BLD BENT TO FHD SIDE
\$5			CTE	THESSALONIKI, GREECE	1 15TG LPC BLD BENT TO FHD SIDE
ΙĮ	YES		SIE	SEA ISLE CITY, NJ	3 F BLOS TIP CURL
១៩	NO YES			OXFORD, ENGLAND	
ומו	YES		RDD	REDDING, CA	
	TNIII				ODOR, FAN STATOR DAMAGED
اے	YES		HSY	NEH ORLEANS, LA	
	YES		STL	ST. LOUIS, HO	
L	YES		TVC	TRAVERSE CITY AIRPORT HI	DENT THE OLD
IVE	YES		CHA		BENT IMP ELD
ISH	UEC			HAUSAU, HI	
IEN	YES		PHX	PHOENIX, ARIZONA	
: H	755		LAX	LOS ANGELES, CA	IMF CHMAGE
ня	ND		LANZHO	CHINA	
IAN	NO		BRYAN	PENANG, HFLAYSIA	12 EXIT GUIDE VANES DAMAGED
RA	YES		TOA	TORRANCE, CR	
	NO			VALPARAISO, CHILE NAVALBASE	1ST THE DONOGE
.PA	H0			LINDSAY, ONTARIO, CAMADA	FAN STATOR DAMAGE
POI	NO			DUMFRIES, SCOTLAND	75% THE HOUSE DENT WIND SE OURD MOOD
1FF	YES			Dom Naco, Decircing	75% IMP VANES BENT/CURLED OVER, ODOR
[YES		FLD	penench wo	2100
DFQ			FLU	BEDFORD, MA	ODOR
DE	YES			SHIDELY, SARATOGA, WY	
mil	Y'ES		GRR	GRAND RAPIDS, HI	
414	YES		GRR	GRAND RAPIDS, HI	
_Oi	YES	CHH-IAD	CHH	COLOMBUS, OHIO	2 F BLDS DAMAGED
[2]	YES	CNH-I AD	CHH	COLOMBUS, OHIO	
.Of	NO	POL-HOR	HORTA	AZORES, PORTUGAL	
IR	YES		ATL	ATLANTA, GA	
_AI	YES			VANDALIA, OHID	
чоζ	YES			HATERBURY, OXFORD, CONN	4.15
TE					4 LP COMP STATOR VANED DEATTACHED
NI (YES			MANION AIRPORT, ILL	PM EVT, 6 IMP BLDS BENT, 2 SEVERELY
ΞT.	YES			VICTORIA, LA	PH EVT, 16% TO LOSS ON POST GRO RUN
	YES			HIDDLETOHN, HD	FUEL NOZZLES AND COMBUSTOR CAN CLOGGED
1P	YES		HEH	HEMPHIS, TENN	FUEL NOZZLES REMOVED FOR CLEANING
[]	YES	YKH-PSC		PASCO, WASHINGTON	FOUND ON GED INSPEC
타	NO	PHK-PHK	PHK	AYRESHIRE, SCOTLAND	OIT STID ZINDI EU
PE ES	NO .		CUT	CHESTER, UK	5 FON DEDUCATED STO COMP DOM
-51	YES		-	ERIE, PA	5 FAN BLADES+1ST STG COMP DAM
ΙE	YES			MEMPHIS, TENN	740
HP	NO NO				IMP BLADES BENT
ΗI			500	SCHIPOL INT., AMSTERDAM	PROPELLOR DAMAGE
EE,	YES		FRG	QUEEN, NY	SLIGHT NICK ON A FAN BLADE
MT.	YES	LAX-CCR	CCP.	CONTRA COSTA, CONCORDI CA	
Li#	YES	LAX-CCR	CCR	CONTRA COSTA, CONCORD CA	
NT.	YES		SBA	SANTA BARBARA, CA	SEVERAL 1 STG IMP VANES BENT
יבנו	NO.			DUNSFOLD. ENGLAND	1STG IMP BENT OVER AT TIP (1""),T2 PROBE
E.	NO.		EDVE	BRAUNSHHEIG. FRG	
нц	NO.		BSL	BASLE, SHITZERLAND	4 FAN BLADES AND STATOR DAMAGED
75 64 54 7	YES	SNA-SJC		SAN JOSE, CR	POLITIC ALL DAMPS AND ALL DAMP
N L	YES	INFT ⇒JU	HPC		FOUND ON POSTFLIGHT INSPECTION
NA,	NO NO		HKC	KANAS CITY, HO	
KĄ			HLP	JAKARTA, INDONESIA	2 1 STG IMPELLER BLDS BENT
1	NO				FOUND ON GRD INSPEC
EI.	NO.		HRR:AFE	AFRICA	
od	NO			HOODFORD, ENGLAND	HOMENTARY 20% TO LOSS, IMP DAMAGE
~*					

EDATE	EVT#	ETIME	SIGH_EVT	AIRCEMET	ENGINE	DRSH	ENG_POS	DHG_CODE :	SEUPPT TV	POM II
12/13/87			MULT ENG-BIRDS	JETSTAR	TFE731	3	2 LEFT INBOARD			
12/13/87			MULT ENG-BIRDS	JETSTAR	TFE731	3	4 RIGHT OUTBOARD	A,D,K		NONE
12/13/87			MULT ENG-BIRDS	JETSTAR	TFE731	3	3 RIGHT INBOARD	A,D		NONE
	5.2					3R		A,D		NONE
12/16/87	96			BAE125 DO 228	TFE731	5 5	2 RIGHT	H,D		NONE
12/17/87	71		MULT ENG-BIRDS		TPE331		1 LEFT	A,K	3	
12/17/87	71		MULT ENG-BIRDS	00 228	TPE331	5	2 RIGHT	H,K	3	
12/30/87	99	16:00:00		LEAR 35A	TFE731	2	2 RIGHT	A,D,K		YES
01/07/88	162		MULT ENG-BIRDS	LEAR 35	TFE731	2	2 FIGHT	H,D,G	2	YES
01/07/88	162		HULT ENG-BIRDS	LEAR 35	TFE731	2	1 LEFT	A.D.G		YES
01/13/88			INV POH LOSS	BAE146	ALF502	R5	4 RIGHT OUTBOARD	H.C.E.K		COMPFI
01/15/88	_	14:00:00		CITATION 3	TFE731	38	2 RIGHT	A.C.H		NONE
01/16/88		11:40:00		BRE146	ALF502	R5	3 RIGHT INBOARD	A*C		NONE
01/22/88	77	7:00:00		COMH 681	TPE331	438L	2 RIGHT			HONE
02/03/88		18:40:00		BRE 146	ALF502	P.5	1 LEFT OUTBOARD		à	NONE
02/11/88		22:22:00		BRE 125-700	TFE731	3R	2 RIGHT		9	HONE
02/15/88		12:30:00		BRE 146	ALF502	F:5	1 LEFT OUTBOARD		9	NONE `
02/16/88	76	8:50:00		00 228	TPE331	5	2 RIGHT	A,K	3	YES
02/18/88	62		HULT ENG-BIRDS	BAE146	ALF502	R:5	3 RIGHT INBOARD		9	NONE
02/18/88	62		HULT ENG-BIRDS	BAE 146	ALF502	F:5	1 LEFT OUTBOARD		9	NONE
02/22/88			MULT BIRDS	LEAR 35A	TFE731	2	2 RIGHT	A,E,D,K		YES
02/22/88		11:00:00		LEAR 35	TFE731	2	2 RIGHT			NONE
03/04/88			INV PON LOSS	MU 2	TPE331	10	1 LEFT	fi,K		SP00:
03/05/88	75	16:45:00		HETRO	TPE331	11	1 LEFT	H,K,P	2	
03/09/88	90	7:00:00		00 228	TPE331	5	2 RIGHT	H,K		NONE
03/10/88	72	9:45:00	NONE	BRE 146	ALF502	R3A	2 LEFT INBOARD	-		NONE
03/14/88	86	15:00:00	NONE	DG 228	TPE331	5	2 RIGHT			YES
03/22/88	76	20:40:00	NONE	BRE 125-700	TFE731	3R	2 RIGHT			NONE
03/22/88	23	10:15:00	NONE	LEAR C21A	TFE731	2	2 RIGHT			NONE
03/23/88	87	19:55:00	NONE	HETRO	TPE331	110	1 LEFT	A,K		NONE
03/25/88	73		NONE	BRE 146	ALF502	R5	1 LEFT OUTBOARD			NONE
03/29/88	74	21:00:00	NONE	BRE146	ALF502	R5	2 LEFT INBOARD			NONE
04/04/88	92	_	HULT BIRDS	FALCON 10	TFE731	2	2 RIGHT	A,G,I,K		FLAME
04/03/88		10:15:00		HESTHE ND	TFE731	3	2 RIGHT	A,D,G,K		
04/12/88	100	8:30:00		HESTH 1124	TFE731	3	2 RIGHT	R.D		HOMENT
04/18/88		17:00:00		CRSR 212	TPE331	5	2 RIGHT			YES
04/24/88		14:15:00		T478	JT 15D	5	1 LEFT	A,K		YES
04/25/88	21	11113100	NONE	BRE146	ALF502	R5	2 LEFT INBOARD		_	NONE
04/27/88		22:00:00		BRE146				۵.		NONE
05/01/88	89	22.00.00	NONE		ALF502	R5	4 RIGHT OUTBOARD	A,L		NONE
05/02/88	94	8:50: 0 0		BRE146	ALF502	R3A	2 LEFT INBOARD			NONE
				CESSNA 550	JT 150	4		^	9	_
05/03/88		13:42:00 15:30:00		HETRO .	TPE331	110	1 LEFT	A,K		HONE
05/04/88				BAE 125	TFE731	3R	2 RIGHT	A,L		NONE
05/05/88		10:35:00		CESSNA 552	JT 15D	5	1 LEFT			NONE
05/10/88		16:00:00		BRE146	ALF502	R5	1 LEFT OUTBOARD		ė	HONE
05/20/88		13:00:00		LEAR 35	TFE731	2	1 LEFT	A,C,P	3	
05/27/88		20:40:00	_	COMH 980	TPE331	25	2 RIGHT		ė	
05/30/88		21:00:00		BRE146	ALF502	R5	3 RIGHT INBOARD		9	NONE
06/04/88		19:30:00		HESTHE ND	TFE731	3	1 LEFT		9	NONE
06/08/88		8:30:00		METRO III	TPE331	110	1 LEFT		લ	NONE
06/11/88	111		NONE	LERR 36	TFE731	2	1 LEFT	A,C	3	
06/20/88	95	9:40:00		BRE146	ALF502	R5	1 LEFT OUTBOARD			NONE
06/20/88	115	9:00:00		HETRO	TPE331	1 1 U	2 RIGHT			NONE
06/20/88		19:50:00	and the second s	CITAT 500	JT 15D	18	2 RIGHT	A,G		N1 CHH
06/27/88	112	3:00:00		LEAR 35	TFE731	2	1 LEFT	A,B,C,P		NONE
06/30/88	96		NONE	BRE146	ALF502	R5	1 LEFT OUTBOARD	- -		NONE
07/01/88	113	14:30:00	NONE	LEAR 35	TFE731	2	1 LEFT			NONE
07/05/88	120	8:00:00	NONE	JS 3101	TPE331	1006	1 LEFT			NONE
07/05/88	121	14:00:00	NONE	JS 3101	TPE331	10UF	1 LEFT			NONE
07/06/88		12:05:00		COMM 1000	TPE331	10	1 LEFT			SPUOL
07/11/86	104		NONE	BAE 146	RLF502	R5	2 LEFT INBOARD			
07/12/88			NONE	BRE 146	ALF502	R5	1 LEFT OUTBOARD			NONE
07/15/88			NONE	BRE 146	ALF502	R5	2 LEFT INBOARD			NONE
			· · =		TILL JUE	110	I AMDUME		ä	NONE

Severity Poh_Loss	DUSK DUSK DUSK	WEATHER OVERCAST OVERCAST OVERCAST
0.K 1 NONE NONE NONE NO TAKEOFF 50 160 VFR 3 2 NONE NONE NONE NO TAKEOFF 50 160 VFR 3 2 NONE NONE NONE NO TAKEOFF 50 160 VFR 3 2 NONE NONE NO APPROACH 1200 160 VFR 4 3 NONE RETARD NO LANDING 0 80 VFR	DUSK DUSK DUSK	OVERCAST OVERCAST
2 NONE HONE NONE NO TAKEDFF 50 160 VFR	DUSK DUSK	OVERCAST
2 NONE HONE NONE NO TAKEDFF 50 160 VFR	DUSK DUSK	OVERCAST
2 NONE NONE NO TAKEDEF 50 160 VFR	DUSK DUSK	
2 NONE NONE NOME NO APPROACH 1200 160 VFR 3 NOME RETARD NO LANDING 0 80 VFR	DUSK	
< 3 NONE RETARD NO LANDING O 80 VFR		
		CLERF:
C 3 NOME RELIGIO IN THIRTING (180 ALK	LIGHT	CLEAR
	LIGHT	CLEAR:
3,K 1 YES HONE NONE NO CLIMB VFR	LIGHT	CLEAR
0.6 2 YES SOME NO TAKEOFF 0 130 IFR		OVERCAST
0.6 2 YES SOME NO TAKEOFF 0 130 IFR		OVERCAST
C.E.K 1 COMPRESSOR IDLE INVOLUNTATAKEOFF 800 VFR	LIGHT	CLEAR
2.H 3 NONE SOME NOME NO TAKEOFF 110 VFR 3 NONE NO UNKNOWN	LIGHT	CLEAR
	Dark	CLERR
9 NONE HONE NONE NO TAKEOFF 40 100 VFR	DAMM	SCATTERED
9 NONE 1.2 IOLE NO LANDING 115 VFR	DUSK	CLEAR
9 NONE HONE NONE NO TAKEOFF O 120 IFR	DARK	F06
9 NONE .6 NO TAKEOFF 120 IFR	LIGHT	CLEAR
< 3 YES HONE CUTOFF VOLUNTRRYTAKEOFF 0 100 VFR	LIGHT	CLERR
9 NONE .3 NO LANDING 115 VFR	LIGHT	CLEAR
9 NONE .3 NO LANDING 115 VFR	LIGHT	CLEAR
B.D.K 1 YES HIGH NOME NO LANDING 20 120 UFR	DARK	CLEAR
9 NONE NONE NO APPROACH 400 140 VFR	LIGHT	CLERR:
< 3 SPOOL DOWN MIGH CUT OFF INVOLUNTRAPPROACH 100	DARK	DRY
CUTOFF VOLUNTARYAPPROACH 1000 160	DUSK	OVERCAST
S NONE IDLE NO TREOFF 0 70 VFR	LIGHT	
9 NONE .2 HONE NO LANDING 0 80 VFR	LIGHT	CLERR:
4 YES YES NO LANDING 0 70 VFR	LIGHT	CLERR:
9 NONE HONE NO APPROACH 2000 130 IFR	DARK	SCAFTERED
9 NONE NONE RETARD NO TAKEOFF 0 95	LIGHT	SHOW
< 4 NONE NONE NONE NO UNKNOWN 600 130		SCATTERED
***************************************	LIGHT	SCAFTERED
	DODY	CLERR
	DARK	
	LIGHT	SCATTERED
· · · · · · · · · · · · · · · · · · ·	LIGHT	CLEAR
0 2 YES NONE NO CLIMB 3000 170 VFR	LIGHT	CLEAR
< 3 YES HIGH NO LANDING O 80 VFR	LIGHT	CLEAR
9 NONE HOME NOME NO APPROACH 2300 180 IFR	LIGHT	CLERR:
9 NONE NO UNKNOWN		
3 NONE NO UNKNOWN	DARK	CLERR
a none NO nukn om		
9 NO APPROACH 170 IFR	LIGHT	RAIN/SNOW
3 NONE NONE NONE NO TAKEOFF O 120 VFR	LIGHT	CLERR
S NONE HONE NONE LANDING 10 122 VFR	LIGHT	OVERCAST
9 NONE NOME NO UNKNOWN VFR	LIGHT	CLERR
9 NONE NO UNKNOWN VFR		
P 3 SOME CUTOFF VIBES TAKEOFF O VFR	LIGHT	CLERF:
9 . NONE NO UNKNOMN 2000-130 IFR	DARK	SNOW
. 9 NONE NO UNKNOWN	DARK	
I 9 NONE HONE NONE TAKEOFF O 120	LIGHT	CLERR
9 NONE HONE NO RPPRORCH 900 180 IFR	LIGHT	SCATTERED
1. 3 NONE NO UNKNOWN		20111121122
9 NONE 0-2 NO LANDING VER	LIGHT	CLEAR:
9 NONE NONE NO RPPROMCH 50 100 UFR	LIGHT	SCAFTERED
2 NI CHANGE NONE IGLE NO CLIMB 1300 148 1FR	DRY	ELEAR:
18.C.P 3 NONE NONE NONE NO APPROACH 100 125 UFR	DARK	
9 NONE NO UNKNOWN	LITTE	CLEAR
I 9 NONE NONE NONE NO LANDING 10 VFR	LICUT	CI FOR
	LIGHT	CLERR
1 A MA 400	LIGHT	OVERCAST
AND THE PARTY OF T	LIGHT	H8224
	LIGHT	CLEAR
Y 9 NONE NO UNKNOWN		
9 NONE NO UNKNOHN		
3 HONE NO NIKHOMM		

EDATE	EVTS . CREH_AC	CREH_AL	BIRO_SEE	BIRD_NAM	BIRD_SPE #_BIRDS	HT_02_1	US_INCID	CTY_PRS	AIRP
12/13/87	ATB	NO	FLOCK	COMMON LAPHING	5N1 ×	7.7	NO		
12/13/87	ATB	NO	FLOCK	COMMON LAPHING	5N1 ×	7.7	NO		
12/13/87	ATB	NO	FLOCK	COMMON LAPHING	5N1 ×				
12/16/87	NONE	ND	NO	200000	1		YES		
12/17/87	NOME	NO	FLOCK	GULL¥					can
12/17/87	NONE	NO	FLOCK	SEAGULLX					FOH
12/30/87				SCHOULLX					FDH
01/07/88	HONE	ND	MO	VOT DELIES	1		NO		
_	DIV	NO	YES	KRIKENE	¥				
01/07/88	VIO	NO .	YES	KAIKENES	X				
01/13/88	ATB	YES	SEVERAL	TURKEY VULTURE	1K1 1		YES	ORK-SNH	OAK
01/215/86	ATB	NO	ONE		1	ı	YES		SLN
01/16/ 88	NONE	NO			1		YES	LAX-SAN	
01/22/ 88	DIV	NO	SEVERAL	DOVEX	1		YES		JAX
02/03/88	NONE	NO	NO .	DOVEX	1		NO	HRE-BUQ	80 0
02/11/88	NONE	NO	NO		1		YES		•
02/15/88	NONE	NO	ONE	SHALLOH*	1		NO	KAB-NKH	KAB
02/16/88	ATB		FLOCK	CRONX	1		NO		1500
02/18/88	NONE	NO	FLOCK	HOUSE HARTIN	18269			HRE-HSV	HSV
02/10/88	NONE	NO	FLOCK	HOUSE HARTIN	18269 2			HRE-HSV	
02/22/88	NONE	NO	SEVERAL	SNOW GOOSE	2J26 2			AKE-NOV	HSV
02/22/88	NONE	NO	NO	SPARRONX	2020 2		YES		HOU
03/04/88									FHA
03/05/88	HONE	NO	NO	LAPHI NG	5N1 1				LBG
03/03/88	NONE	NO .	NO		1		YES		POX
	ATO	NO .	NO	50000000	1		YES		I SP
03/10/98	NONE	YES	SEVERAL	SPARRONX	1		YES	DEN-ASE	ASE
03/14/88	NONE	NO	SEVERAL	HOOD PIGEON	2 P9 1				
03/22/88	NONE	NO	ONE	RING BILLED GULL	14N12 1	16.0	NO		CYYZ
03/22/88		NO	ONE	GRAY PARTRIDGE	4L85 1	14.0	NO		
03/2 3/88 .		YES	YES	AMERICAN HIGEON	2J 71 1				HON
03/25/88	NONE		NO	SPARRONX	1		NO	BEJ-LAN	11011
11375378A	NONE	YES	YES	· ·	3		NO	DEC CIII	
U1/01/88		NO	NO	CANADA GOOSE	2J30 2				5444
04/03/88		NO	THO	IMMATURE COMMON LOOM	1E3 :				PHK
04/12/88		NO	SEVERAL	GULL#					
04/18/88		NO	YES	QUELTENEX	1				
04/24/88				MULLIERER					
04/25/88		NO	MO .		1		YES		PUR
				00.000			YES		
04/27/88	******	NO		COMMON GRLLINULES	7H112	10.7	YES		IRD
05/01/88					1	ļ	YES		
05/02/88		NO	SEVERAL	DUCK	1	ļ	NO	YXD-YHH	YHH
05/03/88		NO	YES		:	l	YES		SBP
05/04/88		YES	FLOCK	SPOTTED DOVE	2 P65	5.5			SSL
05/05/88	NONE	NO	NO	_	1		YES		J.31.
05/10/88							YES	SHF-SHR	
05/20/98		NO	ONE	COMMON SHIFT	1055	_	NO.	2017 "2010	~
05/27/88		NO	YES	SEAGULL	10.33				STR
05/30/08		NO	1 6	J. 100LLX	•		NO UEC		CYY
06/04/88		NO	YES	YILL DEED	ENDO	-	YES	SFO-SNA	
06/08/08				KILLDEER	5N33		YES		
06/11/88		NO	NO			t	YES		BNA
	, ,,,,,,,	NO	NO ·	NEH HORLD FRUIT BAT	SEE REMAR	l 0.5	NO		
06/20/88	11011	YES	SEVERAL	BLACK CROHNED PLOVERSE	:	l 4 0.0	NO	FTV-HRE	HRE
06/20/08		NO	H0		:	1	NO		HH>
06/20/89		NO	NO _			1	NO		,,
06/27/88	HONE	NO	ONE	BARN OHL	152	1 11 3	YES		
06/30/88		NO	NO				NO		
07/01/89		NO	YES	AMERICAN KESTREL	5K26				
07/05/86	******	NO	FLOCK	STARLINGE	JACO		NO.		CAI
07/05/88		NO	DNE				YES		DAY
07/05/88			SEVERAL	STARLINGX			YES		DRY
		NO NO	PEAFIGIF	GULLX		1 34.0			HUC
07/11/86		NO		KILLDEER	5N33	3.0	YES		
				SHIFT	11157	1 2.0	NO		
07/12/ 6 07/15/8		NO		BARN SHALLOH	1U52 18237		YES		

;

LC	02_1	US_INCI	D CTY_PRS	AIRPORT	LOCALE	REMARKS
	7.	* NO				
CC	7.	ND			COVENTRY, ENGLAND	BYPASS+CORE INLET STATORS, LPC BLDS BENT
Ct	7.	NO NO			COVENTRY, ENGLAND COVENTRY, ENGLAND	GUR T VEHICLE BIRD CONTROL
C(R)		YES			RICHHOND, VA-BYRD FIELD	GUN + VEHICLE BIRD CONTROL
FF	0.1	NO.		FBH	FRIEDRICHSHAFEN, GERHANY	FOUR FAN ELADES DAMAGED, 1 AT ROOT 1 STG IMP BLDS BENT
FF	n a	NO.		FDH	FRIEDRICHSHAFEN, GERHANY	IMP SLIGHTLY DAMAGED
61	128.0	NO			CRICIUMA, SOUTHERN BRAZIL	SIX F BLDS TIPS BENT, LPC DAMPGE
US	128.0) NO			USHUAIA, ARGENTINA	E FHN BLADES BENT AND BROKEN
U'		YES	One che		USHUAIA, ARGENTINA	16 FAR BLADES BENT AND BROKEN
51		YES	ORK-SNH	ORK	SAN FRANSICO, OAK., CA	ALL COMP STAGES DAMAGED. ENG FLAMED OUT
5		YES	LAX-SAN	SLN	SALINA, KS CA	3 FAN BLADES BENT
.31		YES		JRX	JACKSONVILLE, FL	FOUND ON GRD INSPEC., 2 FAN BLADES BENT
- 56 <u>(</u> 2)		NO	HRE-BUQ	BUQ	BULARHAYO, ZIMBABHE	HINDE CODE BOHOCE DEHOVIER THE CERTIFICE
7		YES			TRHPA, FL	MINOR CORE DAMAGE REMAINED IN SERVICE
H		NO.	KUB-MKN	KAB	MATABELELAND, AFRICA	BIRD HENT THROUGH BYPASS
B	0.6	NO NO			BAGDORA, BENGAL, INDIA	TO HOMENTARILY DROPPED BELOW 60%
H		NO NO	HRE-HSV	HSV	HASVINGO, ZIMBABHE	BIRD HENT THROUGH BYPASS
H	88.0	YES	HRE-MSV	HSV	MASVINGO, ZIMBABNE	ONE BIRD INTO CORE, ONE THROUGH BYPASS
Н		YES		HOU	HOUSTON, TEX	STGS 1 THRU 4, LPC+HPC BLDS NICKED
5	7.7	NO		FKA LBG	SIERRA VISTA, AZ	576 4 Aug
P		YES		PDX	PARIS, FRANCE PORTLAND, OR	STG 1 AND 2 IMP DAMAGE
R		YES		ISP	RONKOKOHA. NY	15TG IMP BENT+1 BROKEN BLD,25TG VANE DAM CHG IN ENG NOISE, 2 BENT IMP BLOS
F		YES	DEN-ASE	ASE	ASPEN, COL	CHO TH ENG HOLSE, 2 BENT THE ELUS
<u>,</u>	18.0				SUFFOLK, ENGLAND	IPSWICH RIRPORT, RPH DROPPED TO 40 2
τ	16.0 14.0			CYYZ	TORONTO, CANADA	The second secon
R		YES			RAHSTEIN AIR BASE, GERHANY	
۲		NO	BEJ-LAN	HON	HURON, SD	
۲		NO	DES-EHN		HOHHOT, CHINA	
1	128.0			PHK	ISLAMABAD, PAKISTAN HHEELING.IL	
	102.0			I BD	CLEBURNE. TX	NACELLE DAM, \$5 BEARING OVERLOAD
į.	12.0				HERTLE BEACH, SC	N2 INCREASE, N2+TEMP DECREASE MOMENTARILY
F	96.0				RANCAQUA, SANTIAGO, CHILE	EGT UP 20 DEG C, SEVERAL BENT F BLADES 2-1STG INP BLDS BENT, 1 APPROX 30 DEG
F		YES		PUR	PUEBLO, COLORADO	2 1510 1m 8205 82M1, 1 HFFROM 30 BEG
ŧ	10.7	YES			CR	FOUND DURING GROUND INSPECTION
F	10.7	YES		IAD	HASHINGTON, DC-DULLES	FOUND ON GRD INSPEC, HULT AC STRIKES
1		NO	YXD-YMM	· · · · · · · · · · · · · · · · · · ·	COLORADO SPRINGS, COL	FOUND DURING GROUND INSPECTION
i		YES		YHH SBP	FORT MCMURRRY, CANADA SAN LUIS OBISPO, CA	NO ENGINE INGESTION OCCURED, GEAR IMPACT
,	5.5	NQ		S SL	SINGAPORE	SLIGHT 1STG+2 IMP DAM, DEBRIS IN F NOZZLE
1		YES		J 32	PENSACOLA, FL	FAN DUCT DRHAGE.
		YES	SHF-SNR	•	CR	FRIME BURING CORNING INCRECTION
	1.5			STR	STUTTGART, GERHANY	FOUND DURING GROUND INSPECTION BENT F BLD HAD 1.5" CRACK, VLNTRY IFSO
		NO UPC		CYYZ	TORONTO, CANADA	DENT I BED HIND 113 CRICK, VEHICL 1130
	3.0	YES YES	SFCI-SNR		CR	FOUND DURING GROUND INSPECTION
		YES			DENVER, CO	DIFFERENT ENGINE SOUND AFTER INGESTION
	0.5			BNA	NRSNVILLE, TN	• .
	40.0		FTV-HRE	HRE	BRAZIL HARARE, ZIMBABHE	SPECIES (STENODERMATINAE) NOT IN CODES
		NO		HKX	HALHOE, SHEDEN	
		NO			LINATE, HILAN	
	11.3				PALH SPRINGS, CR	ENGINE NOISE, ITT 20-500EG.C ABOVE NORM
	<i>a</i> ^	NO NO			LUTON, SCOTLAND	ABRADABLE BEHIND FAN DANAGED BY IMPACT FOUND DURING GROUND INSPECTION
	4.0 8.0			CYYC	CALGARY, ALBERTA, CANADA	- Anim Moutile avenue Tilbico. Tel.
	8.0			DAY	VANDALIA. OH	CREW TOOK EVASIVE ACTION
	34.0			DRY	VANDALIA, OH	
	3.0			HUC	MUNICH, GERHANY	•
	2.0				APPLETON, HISC	FOUND DURING GROUND INSPECTION
	9.0	YES		FNA	GUERNSEY CHANNEL ISLANDS BAERFIELD, FT HAYNE, IND	
					TMU	FOUND DURING GROUND INSPECTION

EDATE	EVT\$		ETIME	SIGN	_EVT	AIRCRAFT	ENGINE	DASH	ENG	POS	DMG_CODE SEVERI	T Y'	POH_LOSS	MAX_VI
07/16/88		107		NONE		BAE 146	ALF502	R5	3 RI	GHT INBOARD		4	NONE	
07/18/88			10:00:00		ENG	BRE 146	RLF502	R5		GHT INBOHRD	Ĥ.K		NONE	
07718788			10:00:00			BRE 146	ALF502	R5	4 RI	GHT DUTBORRD			NONE	
07/19/88			15:00:00			FALCON 10	TFE731	2	2 RI	GHT			HONE	HONE
97721788		1 18	20:40:00	HONE		LEAR 35	TFF731	2	2 RI	GHT			NONE	NONE
07/21/88		128	21:15:00	HULT	ENG	HU-2	TPE331	6R	1 LE	FT		9	NONE	NONE
07/21/88		128	21:15:00	HULT	ENG	HU-2	TPE331	€A	2 R I	GHT		9	HONE	NONE
07/25/88		109		NONE		BRE 146	ALF502	R5	3 RI	GHT INBOARD		9	NONE	
07729788		124			BIRDS	LEAR 35A	TFE731	2	1 LE	FT	Ĥ,D	2	NONE	NONE
08/04/88			15:00:00			JS 31	TPE331	12UAR	2 RI	GHT	A.K.P	2	YES	NONE
087 60 730		143	15:30:00		POH LOSS	DO 228	TPE331	5	2 RI	GHT	H,K	3	SPOOL DOWN	HOHE
087.037.88		197		HONE		BAE 146	ALF502			GHT OUTBOARD		9		
08/16/88		125	7:53:00		BIRDS	LEAR 35	TFE731	2	2 RI		A,C,K	1		NONE
08/22/88		199	•	HONE		JS 3103	TPE331	100	1 LE			9		
08/23/88		202		NONE		JS 3101	TPE331	100	1 LE			9		
06/25/88			17:15:00			HETRO II	TPE331	1008	1 LE		fl,K	3		SOME
08/31/88			12:00:00		~u0	LEAR 36A	TFE731	2	1 LE		A.B		HONE	NONE
09707788			19:35:00			BRE 146	HLF502	R5		GHT INBOARD			NONE	
09707788			19:35:00			BAE 146	ALF502	R.5		FT OUTBOARD	o e e		NONE	*******
09/13/88			17:00:00		BIKDZ	CITATION 3	TFE731	38R	2 RI		A.C.K		YES	NURE
09/15/88		134		NONE		BRE 146	ALF502	RS DE		FT INGORED FT OUTGORD			HONE	
09/15/488		135	13.30.00	HONE		BRE 146	ALF502	R5			0.1		NONE	NIC:AU
09/22/88		137	13:36:00	NONE		JS 3101 CL600	TPE331 ALF502	10UG L-2C	1 LE		A,L		NONE	MEINE.
09/23/88		144	9:48:00			JS 3101	TPE331	1006	1 LE				NO	. A1. 1445
09/24/88		131	9:42:00			JS 3101	TPE331	1000 1000	1 LE				10% TORQUE	EMUME MUME
09/27/88		132	8:30:00			CESSNA 500	JT 150	18 18	1 LE				NONE NONE	410342
09/29/88		145	5:30:00			SA26	TPE331	1	2 RI				SNALL	HIGH
09/30/88		155	9:30:00			METRO III) PE331	11	1 LE		Ĥ,K		HONE	NOME
10/06/88		201	3.00.00	HONE		JS 3101	TPE331	100	1 LE		1140	ġ	HUME	HOME
10/08/88		141	8:20:00		RIRAS	FALCON 50	TFE731	3	2 CE		Ĥ,Ū,Р	2	NONE	SOME
10/11/88			23:00:00		D211132	BRE 125-700	TFE731	3	2 RI		11,00,1	4		HOHE
10715788			10:30:00			FALCON 10	TFE731	ž	1 LE		Ĥ,D	2	HOHE	HUHE
10/19/88		203		NONE		JS 3101	TPE331	100	1 LE		,.	9	110756	112/12
10/20/88		157		NONE		COMM 6900	TPE331	5	1 LE		A,K	3		
10/20/88			14:00:00			BRE 125	TFE731	3	2 RI		H.B.K	1		SOMS
10/22/88		138	2:00:00			BRE 146	ALF502	R5		FT OUTBOARD	A,C		NONE	302
10/22/88		139	8:30:00	NONE		BRE 146	ALF502	R5		FT INBOARD	A,C		N1 CHANGE	
10/22/88		156	7:00:00	HULT	BIRDS	JS 3101	TPE331	1006	2 RI		H.K		YES	SILIME
10/26/88	:	136	12:20:00	NONE		5211	JT 15D	40	2 CE				NONE	FOINE
10/26/88		168		NONE		FALCON 50	TFE731	3	3 RI		A,K,P		NONE	NONE
10/27/88		146	19:20:00	NONE		BRE 146	ALF502	R3A	1 LE	FT OUTBOARD			NONE	NONE
10/29/88		140	13:28:00	NONE		BRE 146	ALF502	R5	1 LE	FT GUTBOARD			NONE	
10/29/88		161	23:00:00	HONE		HESTH 1124	TFE731	3	2 RI		A,D	2		NONE
11/03/88			7:00:00	NONE		LEAR 35	TFE731	2	2 RI	GHT	A.C	3		HONE
11/03/88		158		NONE		METRO	TPE331	1 1U	2 RI		A,K	-	NONE	NONE
11/09/80			10:15:00			FALCON 50	TFE731	3	1 LE				NONE	MONE
11/10/88		169		NONE		JETSTAR	TFE731	3	3 RI	GHT INBOARD	Ĥ	4		HOHE
11713788		166		NONE		LEAR 35	TFE731	2	1 LE			9	NONE	NONE
11/21/88		167			BIROS	HESTH 1124	TFE731	3	1 LE	FT	A,D		NONE	NONE
11/21/88			15:00:00			LEAR 35	TFE?31	2	2 RI	GHT	A		NONE	NONE
11/28/98		147		NONE		BRE 146-QT	A! F502	R5		FT OUTBOARD		9	NONE	
11/28/88		159	8:00:00			JS 31	TPE331	10 UG	2 RI		A	4	NONE	SOME
11/29/88		148		NONE		BRE 146	RLF502	R5		GHT INBOARD		9	NONE	
12/01/88			14:00:00			BRE146	ALF502	R5			H,C	3	NONE	HOHE
12/02/88		150			BIROS	BAE146	ALF502	R3A		GHT INBOURD		9	NONE	NONE
12.16788		151		NONE	•	BAF 146	ALF502	R5		GHT OUTBOKED		9	NONE	
12/18/88			17:30:00			JS 3101	TPE331	10 UG	1 LE		H,K	3	YES	NONE
12/21/88			13:00:00			T- 4 7	JT 15D	5	2 RI		A,D,K,P	1		HI GH
12/22/88		152		NONE		BRE146	ALF502	R5		GHT INBOARD	A,C,P	3	NONE	
10/02/88			6:00:00			DIAHOND 1A	JT 15D	40	1 LE		A,G	2		SOME
10/28/48	;	173	16:00:00	NUNE		CESSNA 551	JT 150	4	2 RI	GHT	A,G,P	2	NONE	

IDE SEVERI	TY POH_LOSS	MAK_VIBE	THROTTLE	IFSD	POF	ALTITUDE S	PEED	FL_RULES	LT_CONDS	HEATHER
00 2012	··· -			NO	UNKNOHN					
	9 NONE					:	210	IFR	DARK	DRIZZLE
	1 NONE			NO	LANDING LANDING		210	IFR	DARK	ORIZZLE
	9 NONE			HO			50		LIGHT	SCATTERED
	9 NONE	NONE	NUNE	NO	LANDING	980		VFR	LIGHT	CLERR
	9 NONE	NUNE	HOME	HO	UNKNOHN	300		IFR	DARK	SCAFTERED
	9 NONE	NONE	HOME	HD	TAKEOFF			IFR	DARK	SCATTERED
	9 NONE	NONE	NÜME	NO	TAKEOFF				LIGHT	CLERR
	4 NONE			HO	UNKNOHN			IFR	LIGHT	CLEAR:
	2 NONE	NONE	HONE	ND	LANDING	200	120	VER	LIGHT	BROKEN
	2 YES	NONE	HONE	NO	APPROACH	2 0 0		VFR	LIGHT	CLERF:
	3 SPOOL DOWN	HONE	CUTOFF	INVOLUNT	FAAPPROACH	200	103	71.15		
	9				UNKNOHN		120	IFR	LIGHT	SCAFTERED
	í	NONE	RETARD	NO	TAKEOFF	U	120	11.0	E. 2	50 1225
	9				APPF:DACH					
					UNKHOHN		~ =	VFR	LIGHT	CLEAR:
	9	SOME		NO	TAKEOFF	<u>o</u>	95	VFR	LIGHT	CLEAR
	3 3 NONE	HONE	NONE	NO	TAKEOFF	3	120	VER	DARK	CLERF.
		110112		NO	TAKEOFF					
	4 HONE			ND	TAKEOFF				DARK	CI COD
	9 NONE	NONE		ND	TAKEOFF	50	130		LIGHT	CLEAR
	1 YES	Herric		NO	IJNKNOHN					
	9 NONE			NO	UNKNOHN					ou for.
	4 NONE	HONE		NO	APPROACH	10 0	125	VFR	LIGHT	CLEAR
	HONE	MUME		NO	UNKNOHN			_		
	9 NO	· MONE		NO	TAKEOFF	0	101	VFR	FIGHT	CLERF
	4 10% TORQUE	LAUNE		NO	APPROACH	200	125	VFR	LIGHT	CLERF
	4 MONE	NONE	NOME	NO	UNKHOHN			VFR	LIGHT	CLEAR
	9 NONE		HONE CUTOFF	OTHER	TAKEOFF		97	VFR	EIARK	CLEAR
	9 SMALL	HI GH		NO	HPPROACH			VFR	LIGHT	SCAFTERED
	3 HONE	NONE	HONE	110	UNKNOHN					
	ý		017055	OTHER	CLIMB	1150	170	IFR	LIGHT	CLERR
	2 NONE	SOME	CUTOFF		TAKEOFF		150	UFR	EIARK	OVERCAST
	9 NONE	HONE	HONE	NO	TAKEOFF		100	VFR	LIGHT	CLEAR
	2 HONE	HONE	RETARD	NO		_				
	9				UNKNOHN					
	3		_	NO	APPROACH	700	160	UFR	LIGHT	CLEAR:
	1	SOME	RETARD		CLIMB	,,,	. 100		EIRRK	OVERCAST
	4 NONE		HONE	+.0	UNKHOHN		180	IFR	LIGHT	F06
	3 N1 CHANGE		NONE	NO	TAKEOFF	Si	120	VFR	EIRHN	SCAFTERED
	4 YES	SOME	CUTOFF		ARYTHKEDI F	31	3 120	VFR	LIGHT	CLEAR
	9 NONE	NONE	NONE	NO	UNKHOHN			71.5	LIGHT	CLERR
	1 NOME	HONE		NO	IKAT		0 0	VFR	DARK	CLEAR
	9 NONE	HONE		NO	TRXI		0 0	VFK	LIGHT	CLERR
	9 NONE	_		NO	APPROACH		3 110	VFR	DARK	OVERCHST
		NONE	IDLE	NO	TAKEOFF		120		EIRHN	SCAFTERED
	2	NONE	IDLE	NO	LANDING	3	0 125	VFR	Cilimia	14/11/11/11/20
	3	NONE		NO	UNKNOHN		_		SHICK	CLEOR
	NONE	HONE	HONE	NO	APPROACH	110	0 180	1 FR	DUSK	CLEAR
	a HONE	NONE	113.12	NO	UNKNOHN				F.O.L	CI EDG
	4	NONE	HONE	NO	APPROACH			VFR	DAY	CLERR
	9 NONE	NONE	NONE	NO	TAKEOFF		0 140	VFR	LIGHT	SCAFTERED
	2 NONE		NONE	NO	LANDING	10	0 150	UFR	LIGHT	CLERR
	4 NONE	HONE		ND	UNKNOUN					
	9 NONE	cour	HONE	NO	HEPROACH	30	O 130	VFR	LIGHT	CLERR
	4 NONE	SOME	HONE	NO	UNKNOHN	_				
	3 NONE		HONE		APPROACH			IFR	LIGHT	CLERF:
	3 NUNE	HONE	NONE	NO	LANDING		0 110	IFR	DARK	DRIZZLE
	9 NONE	NONE	RETARD	ND						
	9 NONE		HONE	NO	TONEDEE	26	0 120		DUSK	OVERCAST
	3 YES	NONE	NONE	NO	TAKEOFF		30 320	IFR	LIGHT	OVERCHST
: , P	1	HI 6H	IOLE	NO	CRUI SE	130	,, ,,,		LIGHT	
	3 NONE			NO	UNKNOHN		0 120	UFR	DARK	SCAFTERED
	2	SOME	RETARD	ND	TAKEDFF		0 100	IFR	LIGHT	
,	2 NONE		IDLE	NO	TAKEOFF		0 100	41.77		
	- · · · ·									

. coare Cuf	CDEN OF	CREH AL	DIPN SEF	BIRD_NAM	BIRD_SPE	35	HT_02_1	US_INCID	CTY PRS	AIRPORT
EDATE EVI	CREH_AC	-	PIKD_2CC		5H33	1		YES	-	FNS
07716788	HONE	NO		KILLDEER	5N33	ī		YES	CRH-ORD	ORD
07718788		YES		KILLDEER		î		YES	CRH-ORD	ORD
07718788		YES		LESSER YELLOHLEGS	61120	1	8.0		CKM-UKD	LBG
07/19/88	NONE	NO	NO	EURASIAN KESTREL	5K27		0.0	YES		
07/21/88	NONE	NO	YES			1	14.0			PTK
07/21/08	NONE	NO	NO	GRAY FACED BUZZARD*			14.0			
07/21/89	NONE	NO	NO	GRAY FACED BUZZARD*			14.0			
07/25/89	NUNE	NO		AMERICAN KESTREL	5K26	1	3.5	YES	ORD-FHA	
07/29/88	HONE	NO	NO	GULL¥		*		NO		
	NONE	YES	FLOCK	HOOD PIGEON*		1	24.0	NO		
08/04/88		NO	ONE	SEAGULL¥		1	24.0	NO		GHT
08/209/88	NONE	NU	OIL	20110000		1		YES		
00/09/88				GULL≆		풎		NO.		LDK
08/16/88	NONE			SOLEA		1		무 ES		PHL
08/22/88						1		YES		
08/23/88						1		YES		TUP
08725786	ATO	NO	ONE	C111 1 ~		î		NO		NCE
08/31/98	ATB	NO .	ONE	GULLX	17774	1	1 6	YES		ROA
09/07/88	ATB	YES	SEVERAL	HORNED LARK	17274					
09/07/88	ATB	YES	SEVERAL	HORNED LARK	17274	1		YES		ROA
Q9/13/88	NONE	NO	seve ra l	ROCK DOVE	2P1	2	14.0	YES		BUR
09/15/88	NONE	NO	NO			1		YES		
09/15/88			NO			1		YES		FHA
04722788	NONE	NO	NO	GULL≭		1		YES		YKM
	NONE	NO	NO			1		YES		
09/22/88			YES	HORNED LARK	17274	1	1.5	YES		TUP
09/23/88	NONE	NO NO	ONE	SPARRON OR STARLING*	2.2.	1		YES		PDX
09/24/88	HONE	NO		SCHROOM OF STURETING		1		YES	CCR-SHF	CCR
09/27/88	NONE	MO	NO	OU =		î		YES	CCA JIII	DEN
03/53/88		NO	ONE	OHLX		î		YES		SBA
09/30/88	HONE		ONE	GULL×						3 D FI
10/06/86						1		YES		Nor
10/03/88	NONE	NO	ONE	CORMORAN×		4		NO		NCE
10/11/88	ATB	NO	NO	AMERICAN HOODCOCK	6N37	1		YES		FHL
10/15/88	ATO	NO	SEVERAL	Hamkx		1				EDLP
10/19/88						1		YES		
10/20/88			YES			1		NO		
10/20/88	ATB	NO	FLOCK	RING-BILLED GULL	14N12	1	16.0	YES		CCR
10/23/98	NONE	NO	NO	Will Division occu		1		NO	NUR-KOL	BNJ
	_		NO	HOURNING DOVE	2P105	1	4.0	YES	LAX-FAT	LAX
10/22/489	OTHER	NO.	SEVERAL	GULL¥	2. 100	2		YES		EDR
10/22/88	ATB	NO.		CULLX		1		NO		
10/26/88	NONE	NO.	NO	CONO. EUROPE	417202	ĵ		5 NO		TRN
10/26/88		NO	NO	SONG THRUSH	412282				CHC JOH	JAK
10/27/88	NONE	NO	NO			1		NO.	SNG-JAK	
10/29/88	NONE					1		YES	CRH-ROA	ROA
10753700	ATO	NO	NO	GULL*		1	="	NO		TLV
11/03/88	NONE	NO	NO	GULL*		1		YES		
11/09/88	NONE	NO	NO			1	-	YES		
11/09/88	NONE	NO	NO			1	="	YES		HPN
11/10/88	NONE	NO	ND	MEADOH LARK	64267	1	. 3.I	YES		
11/13/88	NONE	NO	YES	KILLDEER	5N33	1	3.	O YES		
11/21/88	ATB	NO	FLOCK	RING-BILLED GULL	14N12	3	17.	O YES		HNO
11/21/98	NONE	NO	YES	The second court		1		NO		SYD
11/20/88			NO			-	į.	NO		
	NONE	NO.		Cit i z		1		YES		PDX
11/28/88	NONE	NO	NO NO	GULL*	20105			O YES	IAD-HLB	
11/29/88	NONE		NO	MOURNING DOVE	2P105		_	YES		CEO
12/01/88	NONE	NO	NO				. 7		RNO-SFO	
12/02/88	NONE	NO	FLOCK	COMMON LAPHING	5N1			7 NO	FBU-NCL	NCL
12/16/88	NONE	NO	NO	LONGEARED OHL	2S 12O			O NO	-PHK	
12/10/88	ATB	NO	NO			:	l į	YES		DAY
12/21/88	DIV	NO	NO	LESSER SCAUP	2J125		1 16.	O YES		
12/22/88	~~ *						1	NO	EDI -ABR	
12/22/89	OTHER	NO	NO				1	YES	KY-LA	
	ALO	NO		GULL≭			1	NO	RON-C1 R	RON
127 28788	niu	170	~~ * F * * * * * * * * * * * * * * * * *					-		-

:DS	HT_02_1	US_[NCID	CTY_PRS	AIRPORT	LOCALE	REHARKS
1		YES		FHS	BAERFIELD, FT HAYNE, IND	FOUND DURING GROUND INSPECTION
1		YES	CRH-ORD		CHICAGO, ILL-OHARE	SEVERAL 1ST STG COMP BLDS BENT
1		YES	CRH-ORD	ORD	CHICAGO, ILL-OHARE	
1	8.0			LBG	LE BOURGET, FRANCE	
1		YES		PTK	PONTIAC, MI	
	14.0				HIYAKO, JAPAN	
	14.0				HIYRKO, JAPAN	TORQUE DROPPED 6% THEN RECOVERED
1	J.5	YES	CIREI-FHA		ILL-IND	FOUND DURING GROUND INSPECTION
*		NO			HILANO-LINATE, ITALY	
1	24.0				CAMBELL TOWN, UK	1 IMP BLD FRILED, 3 IMP BLDS BENT
1	24.0			GHT	HESTERLAND, GERMANY	2-15TG IMP BLDS BENT, DEBRIS IN F NOZZLE
1		YES			CHICAGO, IL	FOUND ON GRD INSPEC. DEBRIS ON INTAKE
*		NO		LDK	LINDKOPING, SHEDEN	
1		YES		PHL	PHILA, PA	
1		YES		T. 170	TURE! A NE	ENGINE REMOVED FOR INSPEC, BENT PROP TIP
1		YES		TUP	TUPELO, HS	1-15TG IMP BLD BENT
1	1 5	NO.		NCE	NICE, FRANCE	1 FAN BLADE LE CORNER SLIGHTLY BENT
1		YES		ROA	ROANOKE, VA	
2	14.0	YES		ROA	ROANOKE, VA	S DELIE III DI COMO A DIVINIONI COMO DELICA
1	17.0	YES		BUR	BURBANK, CA HASHINGTON, D.C.	3 BENT F BLADES, 6 DAMAGED CORE STATORS
1		YES		FHA	FT. HAYNE, IND	FOUND DURING GROUND INSPECTION
1		YES		YKM	YAKIMA, WA	FOUND DURING GROUND INSPECTION
1		YES		i Mi	TETERBORO, NJ	CONLING DAMAGE
î	1.5	YES		TUP	TUPELO, HISS	Ect pier
î	1.0	YES		PDX	FORTLAND, OR	EGT RISE
ī		YES	CCR-SMF	CCR	CONCORD, CA	FOUND DUDTNE COOIND INCOCCTION OF CHE
ī	64.0		CCF. JIII	DEN	DENVER, CO	FOUND DURING GROUND INSPECTION AT SHE PRECAUTIONARY SHUTDOWN, PROP SPIN DAMAGE
1		YES		SBA	SANTA BARBARA, CA	BENT IMP BLOS
1		YES		201.	DRYTON, OH	FOUND ON GRO INSPEC
4		NO	•	NCE	NICE, FRANCE	PRECAUTIONARY IFSD, FAN STATOR DAMAGE
1	6.0	YES		PHL	PHILA, PA	Recharge Trans Line Bulling
1	32.0			EDLP	PADERBORN, GERHANY	12 FAN BLADES BENT
1		YES			BALTIMORE, HD	FOUND ON GRD INSPEC
1		NO .			HOHENENS, AUSTRIA	TOTAL OF CAD TEST LE
1	16.0	YES		CCR	CONCORD, CA	8 F BLOS TIP CURL, COMP STATOR VANES FORM
1		NO	NUR-KOL	BNJ	BONN, WEST GERMANY	FOUND ON GRO, ONE DISTORTED FAN EXIT VANE
1	4.0	YES	LAX-FAT	LAX	LOS ANGELES, CA	N1 HUNTING APPROXIMATELY 2%
2		YES		BDR	BRIDGEPORT, CT	INTAKE COHLING AND PROP DAMAGED
1		NO			PAYA LEBAR, SINGAPORE	FOUND ON GRO INSPECTION, ENGINE REMOVED
1	2.5	NO.		TRN	TORINO, ITALY	AH EVENT, COMP STATOR VANES BENT
1		NO	SNG-JAK	JAK	JAKARTA, INDONESIA	SHALL BIRD
1		YES	CRH-ROA		ROANOKE, VA	
1		NO.		TLV	LOD, ISRAEL	
1	64.0	YES			SCHENECTROY, NY	
1		YES			SAN FRANCISCO, CA	1 BENT DIFFUSER: VANE
1		YES		HPN	HESTCHESTER, NY	
1		YES				
1		YES			HONTEREY, CA	CABIN ODOR
3		YES		HNO	BEDFORD, MA	
1		NO		SYD	SYDNEY, RUSTRRLIA	
-		NO		DD11	AYRESHIRE, SCOTLAND	FOUND DURING INSPECTION
1		YES	7.00 MI 6	PDX	FORTLAND, OR	
1		YES YES	IRD-HLB	cro	HASHINGTON, DC	FOUND DURING GROUND INSPECTION
<u>1</u>		NO YES	RNO-SFO		SAN FRANCISCO, CA NEWCRSTLE. ENGLAND	2 BENT FAN BLADES
1			FBU-NCL -FHK	NCL	PRESTHICK, SCOTLAND	COMP. BURTHS ASSIMIL THE TOTAL
1		YES	-r.mr	DAY	VANDALIA. OH	FOUND DURING GROUND INSPECTION
i		YES		Dill	ENGLAND AFB. LH	1-15TG IMP BLD BENT, COMB LINER CRACKED
1		NO	EDI -ABR		ABERDEEN, SUOTLAND	5 DIFFUSERS DAMAGED, NINOR IMPELLOR DMG
î		YES	KY-LA		OKENSBURO. LA	1 FRN BLADE + 4 EXIT GUIDE VANES BENT
i		NO	RON-C1A	RON	RONDON, COLOMBIA	3 FBLDS LE TIP CORNERS LIB APPROX. 1"X1"
•		•••	AUT CAN	11011	mondong occurrent	FALCONRY BIRD CONTROL

EDRITE	EVT#	ETIME	SIGH_EVT	AIRCRAFT	ENGINE	DASH	ENG_POS	DMG_CODE SEVERIT	Y POH_LOSS
01/02/89	177	14:00:00	NONE	ERE 125	TFE731	3	1 LEFT	ค,อ	2
01/08/89	170	16:00:00	NONE	LEAR 35A	TFE?31	2	1 LEFT	A,D,P	2 NONE
01/29/89	185		NONE	HETRO	TPE331	30	2 RIGHT	fi,K	3 NONE
01/30/89	176	12:00:00	NONE	HESTHIND	TFE731	3	2 RIGHT	A	4 NONE
02/07/89	175	11:44:00	NONE	BRE146	ALF502	R5	4 RIGHT DUTBOARD		9 NONE
02/21/89	179	16:10:00	NONE	BRE 125-800	TFE731	5R	2 RIGHT		9 NONE
02/22/89		16:30:00	NONE	CESSNA 550	JT 15D	4	UNK		9
02/28/89			HULT BIRDS	BRE146	ALF502	R5	2 LEFT INBORRD	A,K,L	1 COMPRESS
03/06/89	180		NOME	BRE 146	ALF502	R5	4 RIGHT OUTBOARD		9 NONE
03/07/89	181		NONE	BRE146	ALF502	R5	1 LEFT OUTBOARD		9 NONE
03/16/89			NONE	BRE 1-46	ALF502	R5			9
03/16/89	186		NOME	LEAR 35A	TFE731	2		A,D,K,P	i
03/17/89	183	15:15:00	NONE.	BRE 146	ALF502	R5	1 LEFT OUTBOARD		9 COMPRESS
03/20/89	187	12:30:00	NONE	LEAR 55	TFE731	ЭR	1 LEFT	A	4 NONE
03/21/89	184	13:00:00	NONE :	BRE 146	ALF502	R5	3 RIGHT INBOARD	A,K	1 NONE
03/24/89			NONE		TFE731				9
03/24/89	192		NONE.		TFE731				á
04/02/89	198	15:30:00	NONE	FAIRCHILD	TPE331	13		A	4
04/07/89	193	14:15:00	NONE	METRO	TPE331	1 1U	1 LEFT	A	4 YES
04/10/89	194		NONE	JS 3101	TPE331	. JUF	1 LEFT		3 NONE
04/12/89	189		NONE	BRE 146	FILF 502	R5	3 RIGHT INBOARD		9 NONE
04/13/89	195	18:30:00	NONE	COMM	TPE331	5	2 RIGHT	A	4 YES
04/26/89		-	NONE	BRE 146	ALF502	R5	1 LEFT OUTBOARD		S NONE
04/26/89	196			JS 3101	TPE331	1006	1 LEFT	A	4 YES

186_CODE	SEVERITY POH_LOSS	MRX_VIBE	THROTTLE	IFSD	P 0F	ALTITUDE	SPEED	FL_RULES	LT_CONDS	HEATHER
				NO	TAKEOFF	50		VFR		BROKEN
1.D	2	HOME	RETARD	NG	TRKEOFF	30	160	VFR	LIGHT	OVERCAST
1.D.P	2 NONE	NONE	KETTING	NO	LANDING					
i.K	3 NONE		RETARD	NG	TAKEOFF	50		VFR	LIGHT	CLERR:
4	4 NONE	NONE	KETTIND	NO	UNKHOHN				LIGHT	CLEAR
	9 NONE	NONE		NO.	APPROACH	800	250	VFR	LIGHT	CLEAR
	9 NONE	NONE		NO	TRKECFF				DAY	
	9			NO	APPROACH	900	170		DUSK	
1,K,L	1 COMPRESSOR			NO.	UNKHOHN					
	a NONE			NG	UNKHOHN					
	9 NOME			NO	LINKHOHN					
	9			YES	UNKHOHN					
3.D.K,P	1		IDLE	NO	APPROACH	50	110	VFR	LIGHT	CLEAR
	9 COMPRESSOR	MONE	TULL	NO	RPPROACH	1400	140	VFR	LIGHT	CLERR
3	4 NONE	NONE		HO	UNKHOHN			VFR	LIGHT	CLERR
i,K	1 NONE			110	UNKHOHN					•
	9				UNKNOHN					
	9				TAKEOFF			•		
4	4			NO	CLIMB	300		VFR	LIGHT	CLEAR
4	4 YES			NG	UNKHOHN			IFC	LIGHT	CLOUDY
	g none	HONE		NO	UNKHOHN				LIGHT	
	3 NONE			YES	LANGING	25	105	VFR	DUSK	CLEAR
-3	4 YES	HOHE		NO NO	UNKHOHN	40		IFR	LIGHT	CLERR
	9 NONE				TAKEOFF	100	120	IFR	LIGHT	OVERCAST
H	4 YES	NOME		NO.	I TINEUT F	100		2- 30		

01/02/89 DIV GULL* 1 NO 01/08/89 AFB NO ONE BUZZARD* 1 NO 01/29/89 HAGPIE* 1 40.0 NO 01/30/89 AFB NO YES ROCK DOVE 2P1 1 14.0 YES	GIG SHV ORD
01/29/89	SHV ORD
01/30/89 ATB NO YES ROCK DOVE 2P1 1 14.0 YES	ORD
	ORD
02/07/89 MOURNING DOVE 2P105 1 4.0 YES	
02/21/89 NONE NO YES GULL¥ 1 NO	- 44
02/ 22/89 NONE 1 YES TX-OK	TX
ng/28/89 YES FLOCK SNOW GOOSE 2J26 ¥ 88.0 YES SNA-Sh	F SHF
03/06/89 HOURNING DOVE 2P105 1 4.0 YES	FHA
03/07/83 1 No	
n3/16/89 1 NO	
00/16/89 NO NO 1 NO	
UJZ 17289 NONE NO YES GULL¥ 1 NO	
03/20/89 NONE NO REDTAIL HANKS 1 50.0 YES	POU
0 1/21/99 YES NO 1 NO	. 50
03/24/89 COMMON STARLING 21275 1 3.0 UNK	
03/24/89 HOUSE SPARROH 70Z12 1 1.0 YES	
04/02/89 GULL* 1 YES	LAX
04/07/89 ATB NO NO GULL# 1 YES	LRX
04/10/89 NONE NO NO MOURNING DOVE 29105 1 4.0 YES	LIIN
04/12/89 1 Mg	
11/17/19 NONE NO ONE RING-NECKED PHEASANT 4L161 1 40.0 YES	LHS
04/29/289 NONE NO MONGOLIAN PLOVER 5N45 1 2.0 NO	LHO
04/20/209 ATB NO YES STARLING 21275 1 3.0 YES	DAY
21213	חחד

-4	_02_1	US_INCID	CTY_PR5	AIRPORT	LOCALE	REMARKS
州森代斯语		NO			VICTORIA, CANADA	DEBRIS IN CORE AND BYPASS
₹.		NO		616	RIO DE JANEIRO	FRN EXIT GUIDE VANES BROKEN
10	40.0	NO			HOUNT GAMBIER, AUSTRALIA	1ST IMP BLADE BENT, 1ST DIFF VANES BENT
51F	14.0	YES		SHV	SHREVEPORT, LA	
.⊁ }•	4.0	YES		ORD	CHICAGO, IL-OHARE	FOUND ON GRD INSPEC
, E 4 e		NO			CHESTER, ENGLAND	
1		YES	TX-DK	TX	HONROE, TX	
T.	98.0		SNA-SHF	SHF	SRCRAHENTO, CR	DAMAGE TO 1 STG COMP BLADES
•	4.0	YES		FNA	FORT HAYNE, IN	FOUND ON GRO INSPEC
:1		NO				
15.		NO			BUDRPEST, HUNGARY	
18		NO			BRAZIL	FAN BYPASS STATORS EXITED FAN EXHAUST
计技术		NO			OXFORDSHIRE, ENGLAND	ENG REMOVED TO CLEAN OUT BIRD DEBRIS
f	50 .0			POU	HAPPINGER, FL	
		NO			CARATHA, AUSTRALIA	FOUND ON GRO, 8,1ST STG COMP BLOS BENT
		UNK				
r	1.0	YES				
E E		YES		LAX	LOS ANGELES, CA	ENGINE CHANGE, AIRCRAFT SA227AC
15-	4.0	YES		LAX	LOS ANGELES, CA	•
Έ	4.0	YES			DAYTON, OH	
É		NO			BENSON, ENGLAND	FOUND ON GRD INSPEC
m m m m		YES		LHS	LEHISTON, ID	
15		NO		5011	BEIJING-LANZHOU, CHINA	
1	J.U	YES		DAY	VANDALIA, OH	

APPENDIX C STATISTICAL METHODS USED

Statistical analyses are based on an underlying probabilistic model of the process that gave rise to the data. For example, to provide the basis for comparing the weights of ingested birds in the United States and overseas, it is necessary to hypothesize an underlying random distribution of bird weights. That is, the analyst hypothesizes that there is a population of birds, that these birds have different weights, and that the ingestion process "picked" birds from this population in such a way that all birds had equal chances of being selected (this is really the meaning of "random").

Statistical analyses are somewhat more sophisticated than descriptive data analyses, and more care is required to ensure that the methods are appropriate for the data. Statistical analysis is basically formalized inductive reasoning. Hypotheses about bird ingestion hazards are evaluated for consistency with the data that have been collected. Statistical analysis provides the rules for quantifying the level of consistency between the data and a given hypothesis, and thereby forms the basis for objective and unbiased decisions. The process is known formally as statistical hypothesis testing, and a brief outline of the procedure is presented here.

The basis of a statistical hypothesis test is the hypothesis, which is a formal statement about a relationship in the data. If the data are found to be inconsistent with the hypothesis, then the hypothesis is rejected. Conversely, if the data are consistent with the hypothesis, the hypothesis cannot be rejected and is then tentatively accepted. (Note that a tentatively accepted hypothesis may have to be rejected on the basis of later data; hence, failure to reject is not the same as proof of validity. By contrast, a hypothesis that is rejected is unlikely to be "accepted" on the basis of later data.)

For instance, in comparing the weight distributions of United States ingestions versus foreign ingestions, one hypothesis is that there is no difference in the sizes of the birds ingested in the two regions. However, because of randomness in the ingestion process, it would be very surprising if the data on bird weights were identical for the two regions. The purpose of the statistical analysis, then, is to determine whether the data are consistent with the hypothesis, despite the occurrence of random variation.

The rules for deciding whether to accept or reject the hypothesis are based on the possible errors that could be made. A type I error refers to the situation in which the hypothesis is true but we reject it. A type II error occurs when the hypothesis is false but we fail to reject it (we accept it).

The goal of the statistician is to minimize the likelihood of both types of errors. Unfortunately the likelihood of a type I error is reciprocally linked to the likelihood of a type II error, so that lowering the likelihood of either type of error raises the likelihood of the other type error.

Since only one of the errors can be fully controlled, it has become standard practice to control the likelihood of a type I error and accept whatever probability of a type II error results. The likelihood of a type I error is called the "significance level" of the test. The test hypothesis is chosen so that it should be accepted unless there is strong evidence that it is not true.

If the data appear to present strong evidence that the hypothesis is false, then the hypothesis is rejected. With likelihood equal to the significance level, this rejection is a mistake caused by randomness in the data.

For instance, if we hypothesize that there is no difference in the weight distributions of birds ingested in the United States and overseas, we would then select a statistical test which has a low significance level (such as 1 percent). That is, the probability of falsely rejecting the hypothesis is controlled to be 1 percent. If the test showed the data to be inconsistent with the hypothesis, then we would consider ourselves safe in rejecting the hypothesis.

Another aspect of evaluating the efficiency of a statistical test is its ability to detect when the test hypothesis is false. This ability is called the power of the test and is defined to be the probability of rejecting the test hypothesis when it is false and should be rejected. Generally there are many alternatives to the test hypothesis. For instance, one alternative to the hypothesis of equality of bird weight distributions inside and outside the United States is that birds outside the United States are heavier than those inside. Yet another alternative hypothesis is that birds outside the United States are lighter than those inside the United States. A test which was very powerful under the first hypothesis might be very weak under the second hypothesis. The power of a test is therefore a function of the specific alternative hypothesis being considered.

A variation on the statistical hypothesis test is the calculation of a confidence interval for a parameter such as the overall probability of ingestion (POI). The POI is computed by dividing the number of ingestion events by the number of opportunities for an ingestion event. However, because of randomness, the actual number of ingestions might be more or fewer than the number associated with the "true" POI. Since we have made no specific hypothesis about the POI, we use a confidence interval to describe the range of probabilities which is consistent with the data. The confidence level associated with a confidence interval is the likelihood that the true value of the parameter (in this case the POI) is contained within the interval. The confidence level thus amounts to one minus the significance level of a hypothesis test.

In determining whether the data are consistent with a particular hypothesis, we must sometimes account for "degrees of freedom." Suppose that a population can be described by two parameters. For illustrative purposes we can use the mean and qstandard deviation. Note in particular that the mean is used to compute the standard deviation. Suppose we have a hypothesis that a certain population has specific values for the two parameters. We could test the hypothesis by collecting a sample of, say, 10 items from the population. We would compute the sample mean and use a statistical test to compare this with the hypothesized mean. In addition, we would compute a standard deviation from the sample data, using the hypothesized mean rather than the sample mean in the computation. We would then use a statistical test to compare the computed standard deviation with the hypothesized standard deviation. In both cases, we would reject the hypothesis if the statistical test showed there was "too much" difference between the computed and hypothesized values. In computing the two "statistics," we would have used the 10 independent sample values. The tests would then be said to have 10 degrees of freedom.

Suppose, alternatively, that we have no hypothesis about the mean, but we wish to estimate the standard deviation. We could again collect a sample of 10 items. We would compute the mean from the sample, and use this computed mean in the computation of the standard deviation. In statistical parlance, we have "used up one degree of freedom" by so doing. The standard deviation no longer involves 10 independent items. Once the sample mean is fixed, then only 9 items can be picked independently. The value for the 10th is already determined by the first 9, since it must be such as to produce the fixed mean.

A similar situation arises in chi-square tests. For instance, suppose an overall rate is to be compared with a rate in each of several categories. An instance of this is computing an overall ingestion rate per operation and comparing this with individual engine ingestion rates. Computing the overall rate uses up one degree of freedom, reducing the degrees of freedom available to determine the power of the test in distinguishing genuine differences among the categories.

In general, then, when an estimate of one parameter involves another parameter, which itself must be estimated from the sample, we lose degrees of freedom. The consequence is that the statistical test is less effective. For a given likelihood of a type I error, there is a higher likelihood of a type II error (the test has lower power) than would be the case if more degrees of freedom were available. In all cases in the report where this issue is relevant, the number of degrees of freedom of the statistical test is stated.

In the report, the term "Bernoulli trial" is used. This refers to a situation (trial) in which only two outcomes are possible: heads/tails, success/failure, damage/no damage, etc.